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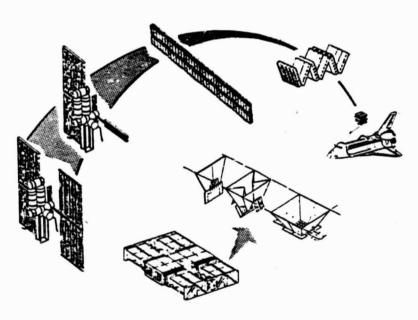
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LOW CONCENTRATION RATIO SOLAR ARRAY FOR-LOW EARTH ORBIT MULTI-100 kW APPLICATION

VOLUME 2—DRAWINGS

FINAL REPORT





Prepared for:

National Aeronautics and Space Administration George C. Marshall Space Flight Center Marshall Space Flight Center, AL 35812

Contract NAS8-34214

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Shuttle Integration & Satellite Systems Division



SSD83-0075-2

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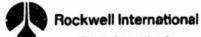
VOLUME 2 - DRAWINGS

JULY 1983

PREPARED FOR:

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION GEORGE C. MARSHALL SPACE FLIGHT CENTER, AL 35812

CONTRACT NAS8-34214



Shuttle Integration & Satellite Systems Division 12214 Lakewood Boulevard Downey, California 90241

TECHNICAL REPORT INDEX/ABSTRACT

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DESCRIPTIVE TERMS

Concentrators, Solar Arrays, Concentrator Arrays, Concentrating Silicon Solar Cells, Concentrating Gallium Arsenide Cells, Low Earth Orbit Multi-100 kW Solar Array Design

ABSTRACT

This report describes a preliminary design effort directed toward a low-concentration-ratio photovoltaic array system based on 1984 technology and capable of delivering multi-hundred kilowatts (300 kW to 1000 kW range) in low earth orbit. The array system consists of two or more array modules each capable of delivering between 113 kW to 175 kW using silicon solar cells or gallium arsenide solar cells, respectively.

The array module deployed area is 1320 square meters and consists of 4356 pyramidal concentrator elements. The module, when stowed in the Space Shuttle's payload bay, has a stowage volume of a cube with 3.24 meters on a side. The concentrator elements are sized for a geometric concentration ratio (GCR) of six with an aperture area of 0.5 meters x 0.5 meters.

Volume 1 discusses the structural analysis and design trades leading to the baseline design. It describes the configuration, as well as optical, thermal and electrical performance analyses that support the design and overall performance estimates for the array. Experimental results are also presented for a concentrator element using both silicon and gallium arsenide solar panels. They confirm the preliminary design analysis and performance estimates. Recommendations are provided for future development effort for low earth orbit application. Volume 2 provides drawings for the preliminary design configuration and for the test hardware that was fabricated for design evaluation and test.

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FOREWORD

This report describes the effort performed for the preliminary design of low-cost concentrator multi-hundred kilowatt solar arrays. The Volume 1 report summarizes activities performed between June 18, 1981 and July 1983, as required by Contract NASS-34214 Statement of Work. Volume 2 contains drawings prepared describing the preliminary design configuration, test hardware and manufacturing flow concept. The report was prepared by the Shuttle Integration and Satellite Systems Division of Rockwell International Corporation for the NASA George C. Marshall Space Flight Center (MSFC), Huntsville, Alabama. The NASA technical Contractor Officer Representative for the activity is Mr. W. L. Crabtree. The contents of this document are not necessarily endorsed by the NASA-MSFC.

Mr. S. J. Nalbandian is the project supervisor. Dr. E. P. French is the assistant project supervisor. Principal contributors to the project were:

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H. C. Ayers	Reflector Design
Z. Backovsky	Testing and Thermal Analysis
R. A. Bellgardt	Electrical Test Equipment
M. S. Biss	Overall Preliminary Design
J. L. Edwards	Structural Analysis
J. D. Eliot	Mechanical Test Equipment
Dr. E. P. French	Optical and Thermal Analysis
G. C. Frey	Materials
R. V. Frost	Reflector Panel Fabrication
H. S. Greenberg	Initial Structural Design and Analysis
K. M. Hicks	Manufacturing Planning
Dr. L. Hsu	Solar Cell Technology
R. L. Long	Materials
M. W. Mills	Electrical Testing and Analysis
Dr. T. S. Nishimoto	Structural Analysis
F. A. rerry	Structural Analysis
A. M. Pope	Development Plans
D. A. Reed	Initial Preliminary Design
A. A. Sileski	Test Planning

Test Hardware

L. Vega

Shuttle Integration & Satellite Systems Division



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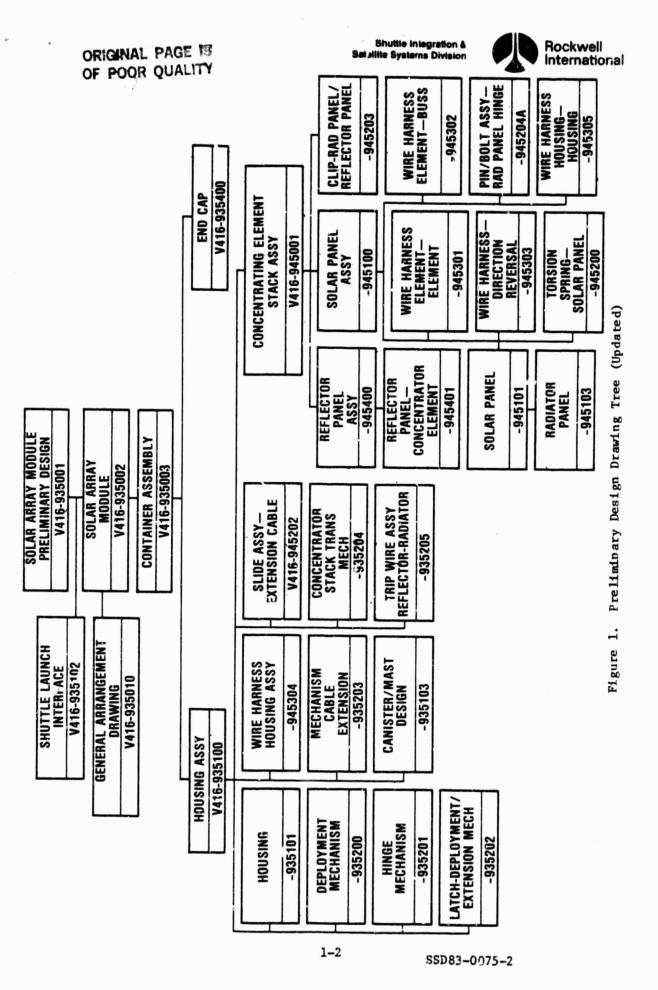
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1.0 PRELIMINARY DESIGN DRAWINGS

1.1 DISCUSSION

The solar array preliminary design developed under this program is described in a set of 30 level-one drawings. These drawings and their relationships are depicted in the drawing tree shown in Figure 1. Together with their associated callouts and specifications, the drawings provide a physical description of two variants of a 1320 m² solar array module (deployed area), one fitted with silicon solar cells and the other with gallium arsenide solar cells. The requirements, trades and analytical studies leading to the design are fully described in Volume 1 of this report.

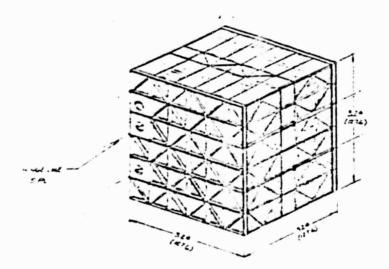




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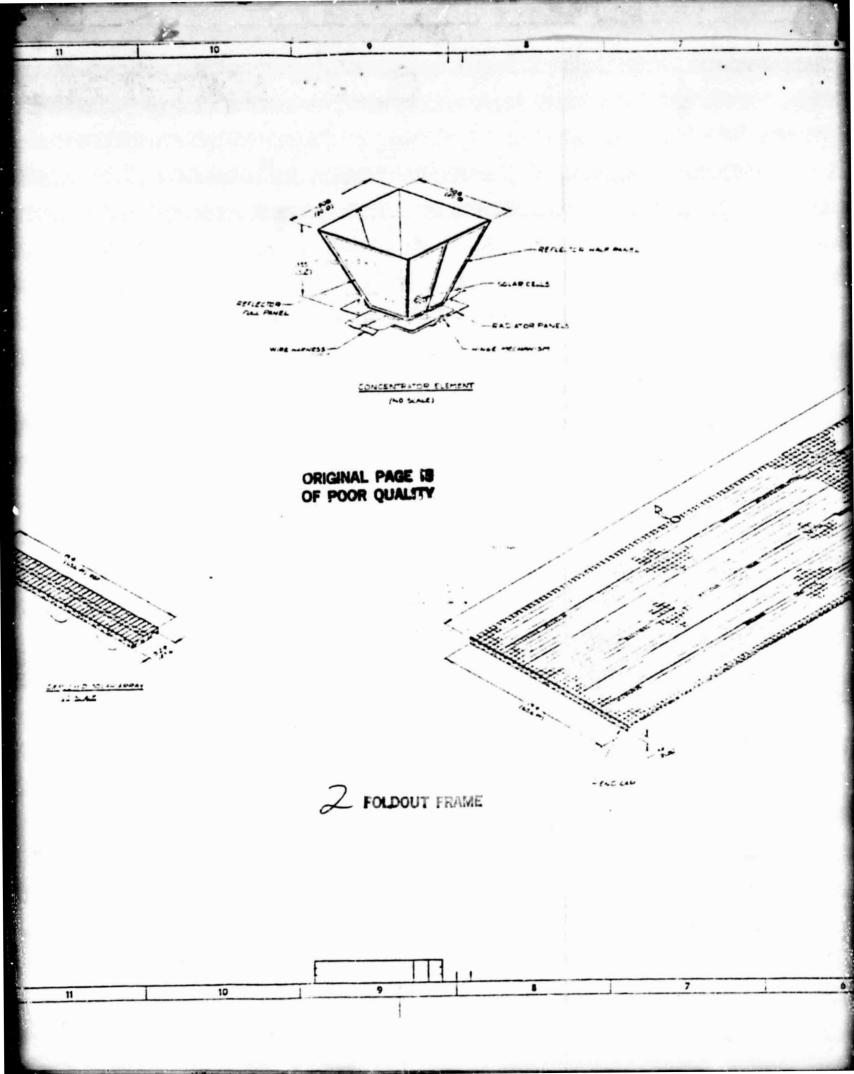
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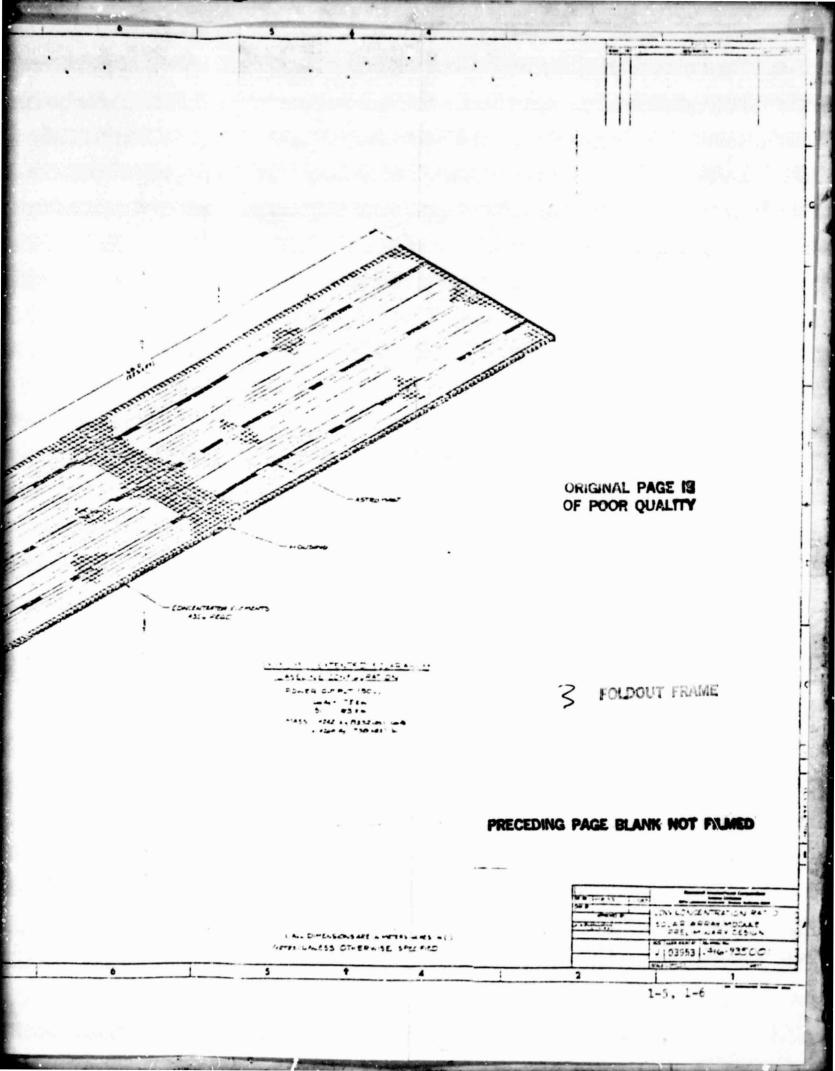


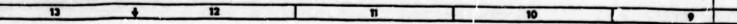
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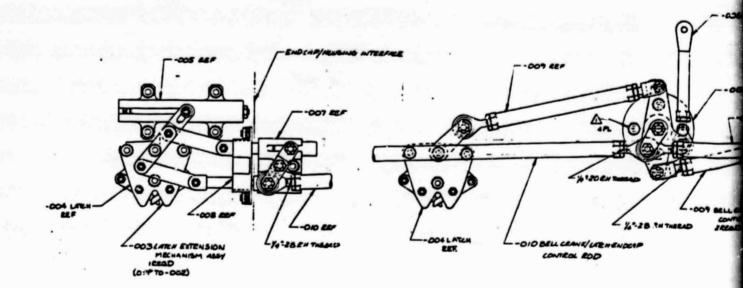
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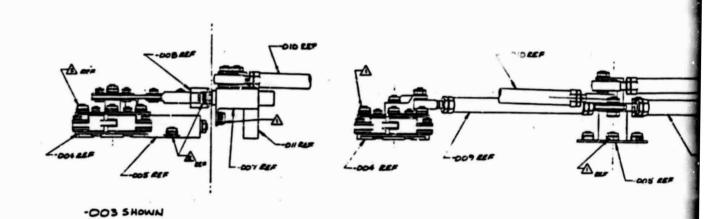


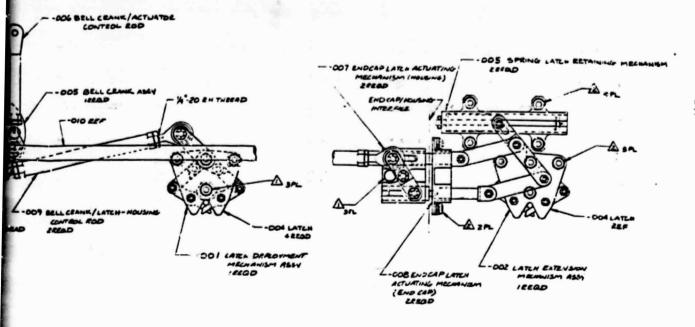
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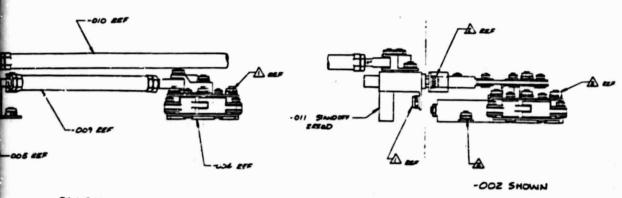
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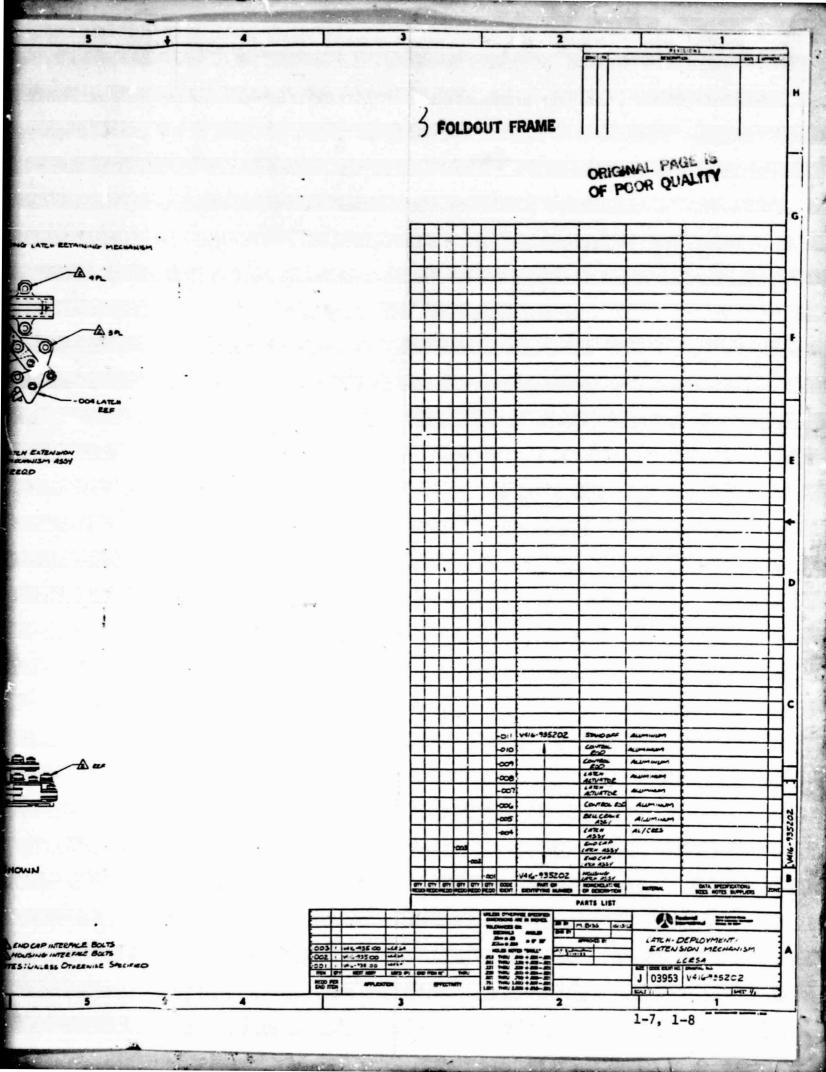
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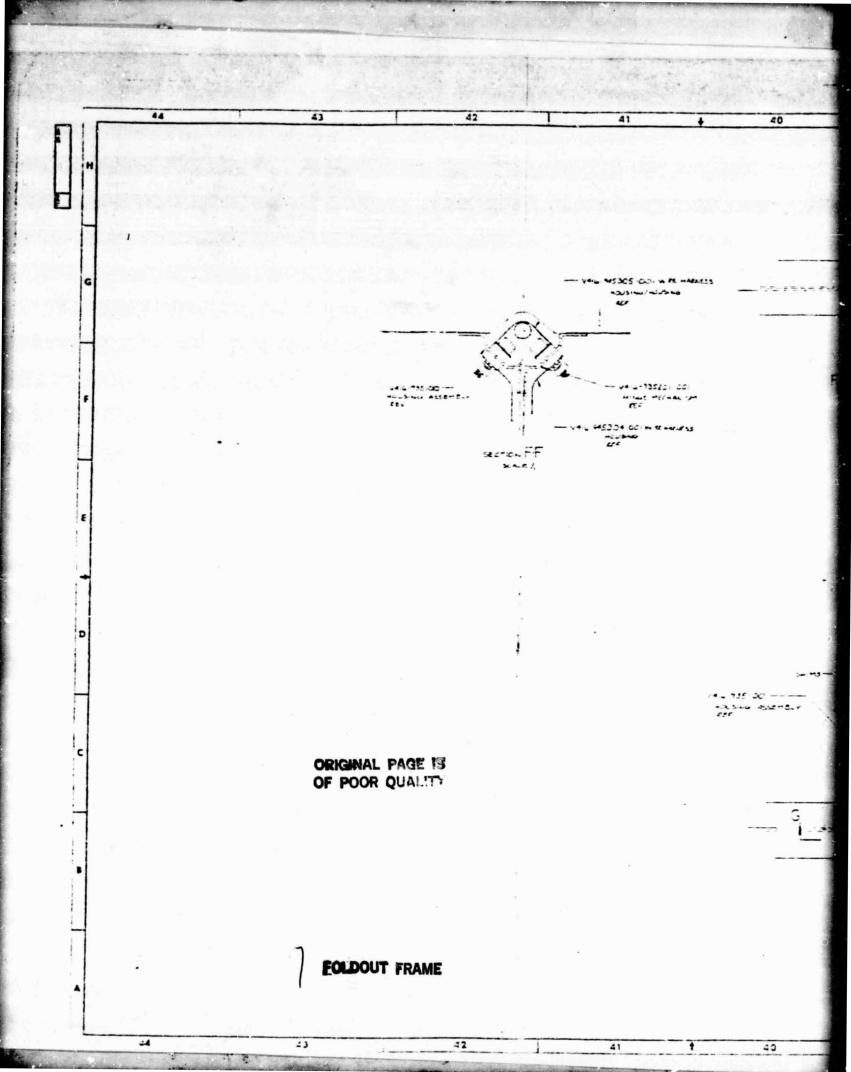
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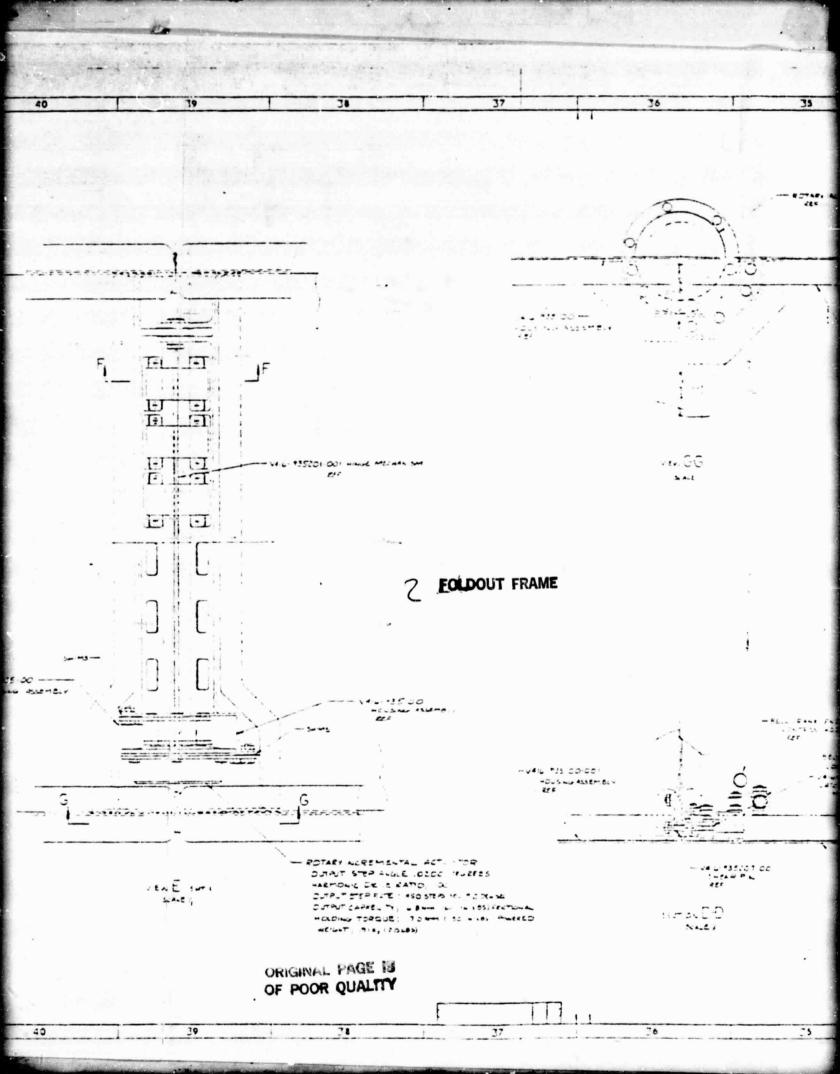
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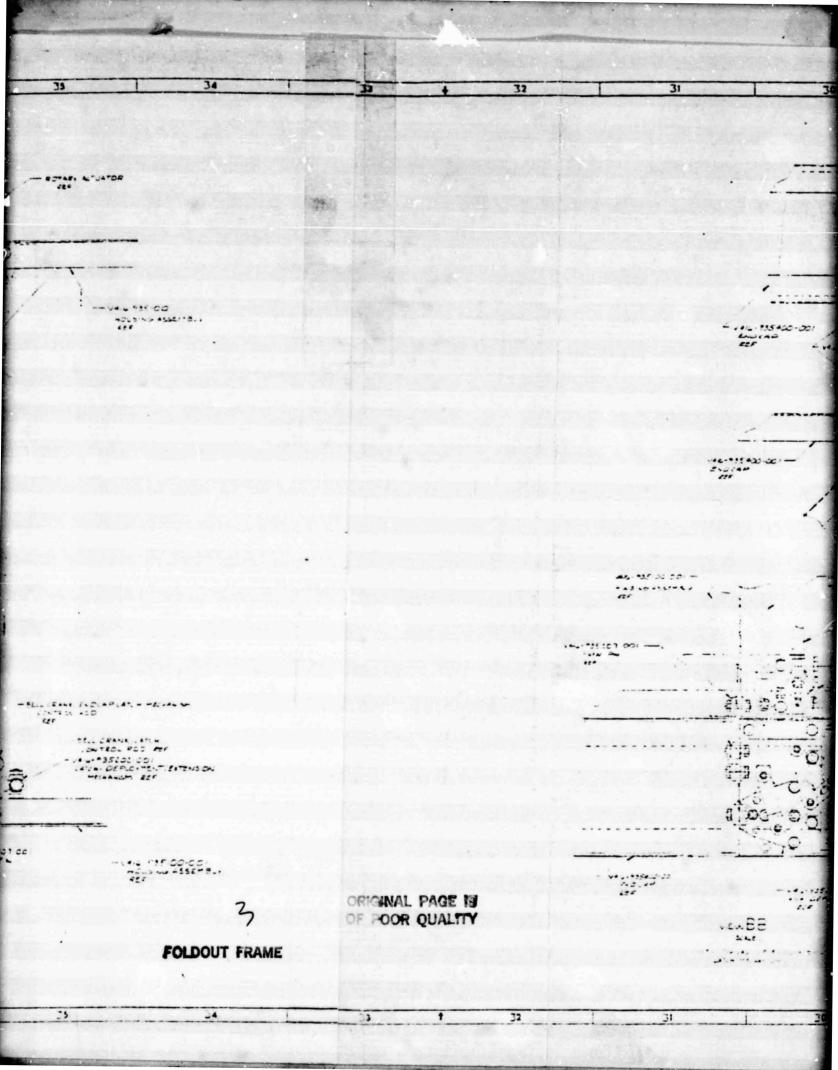
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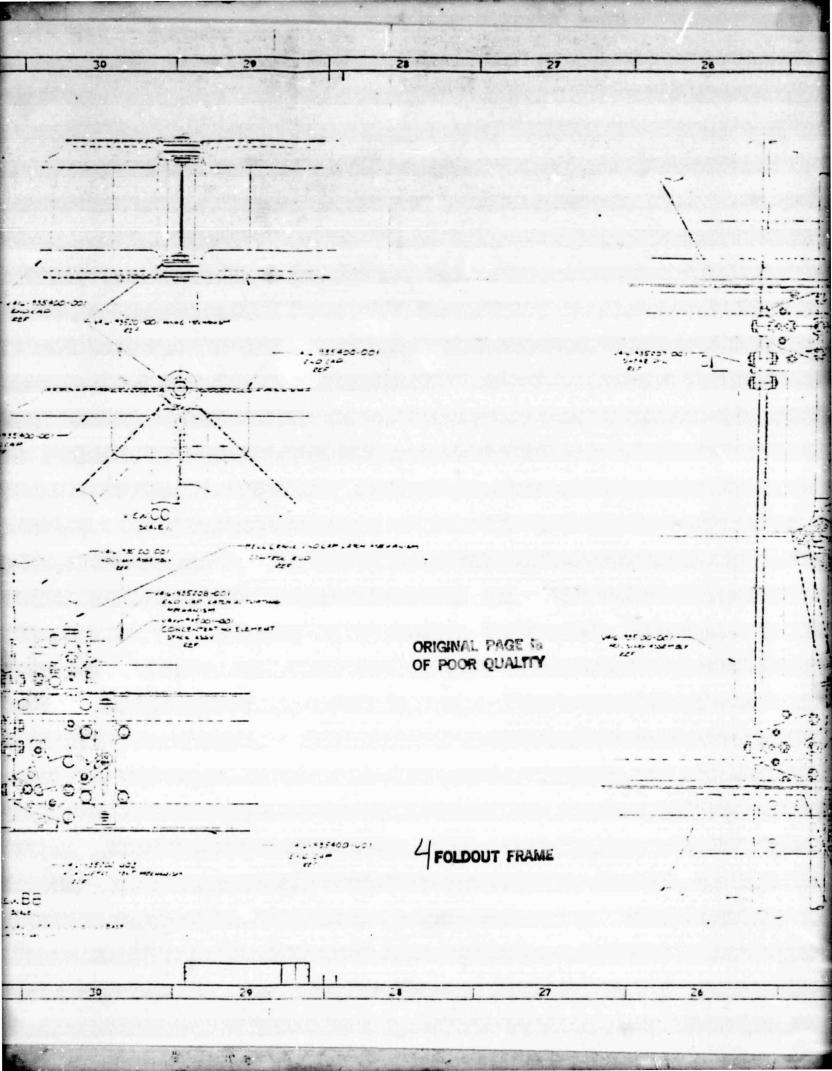
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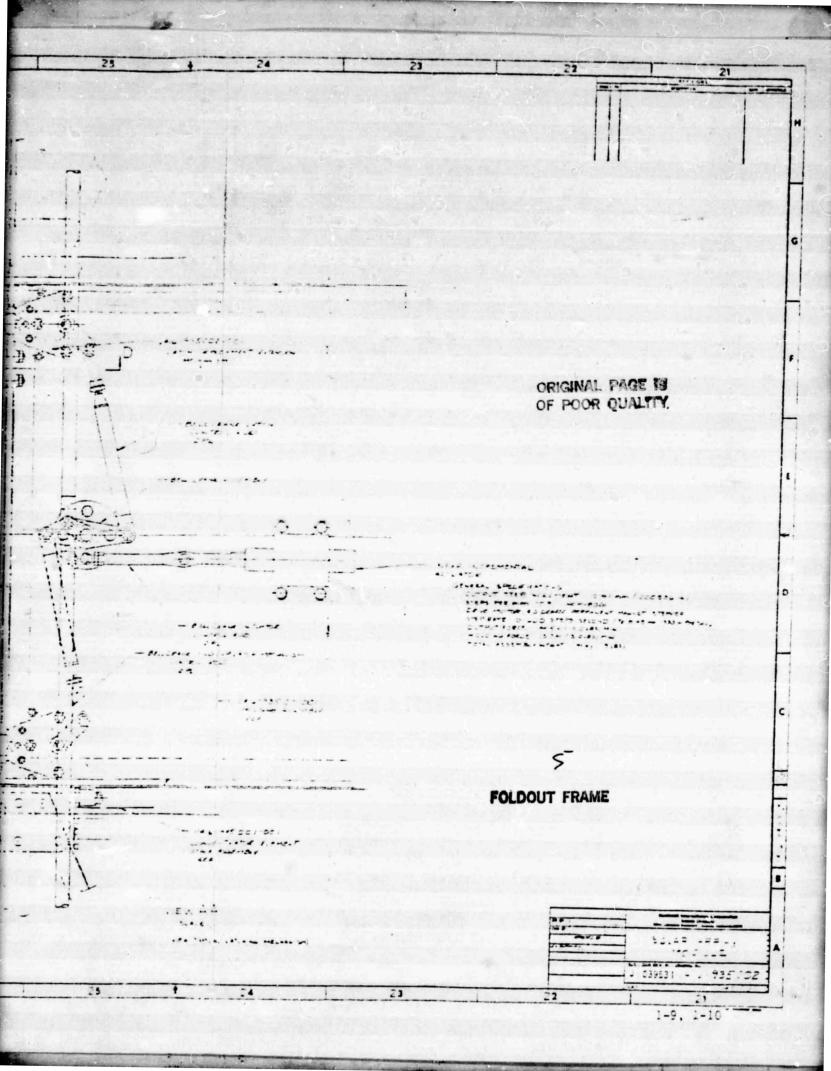


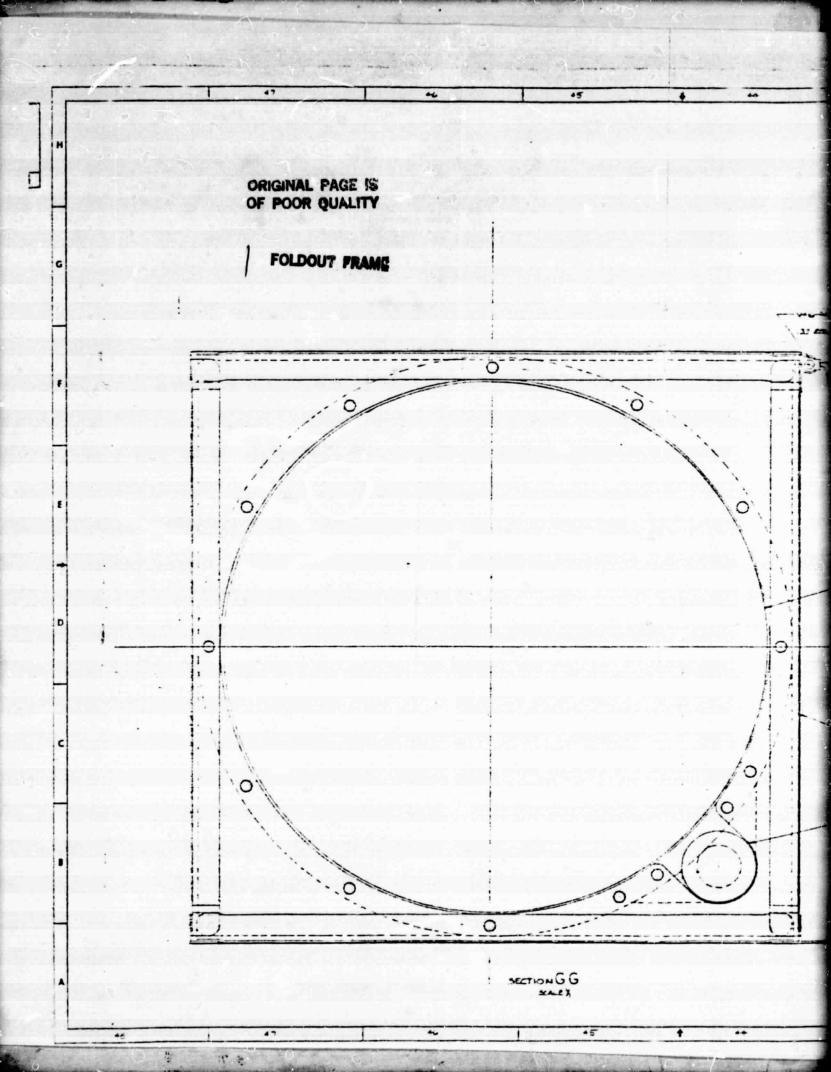




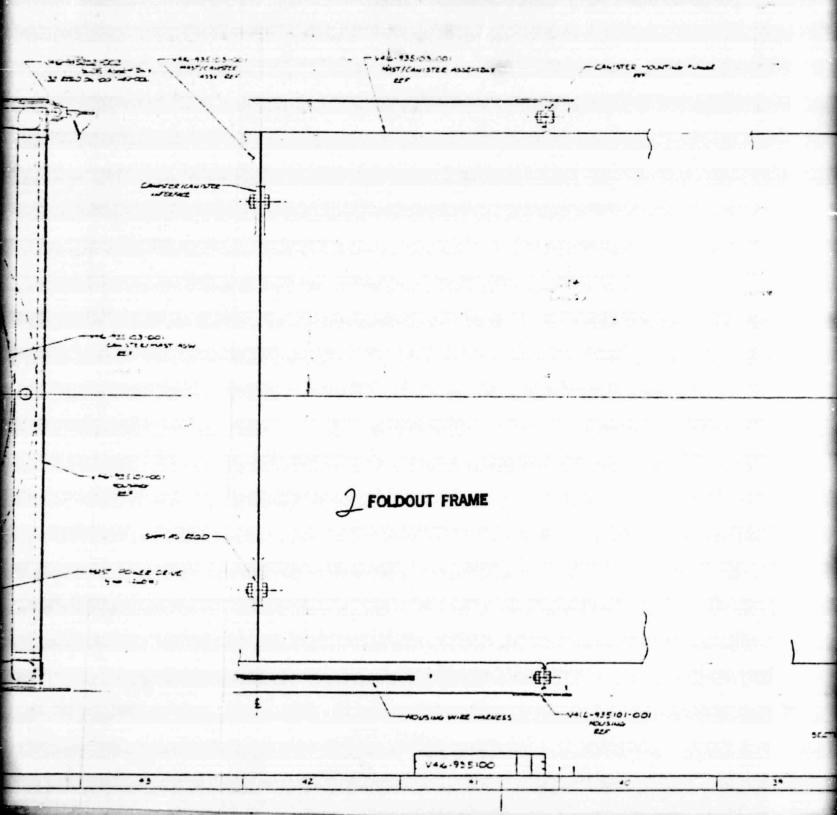


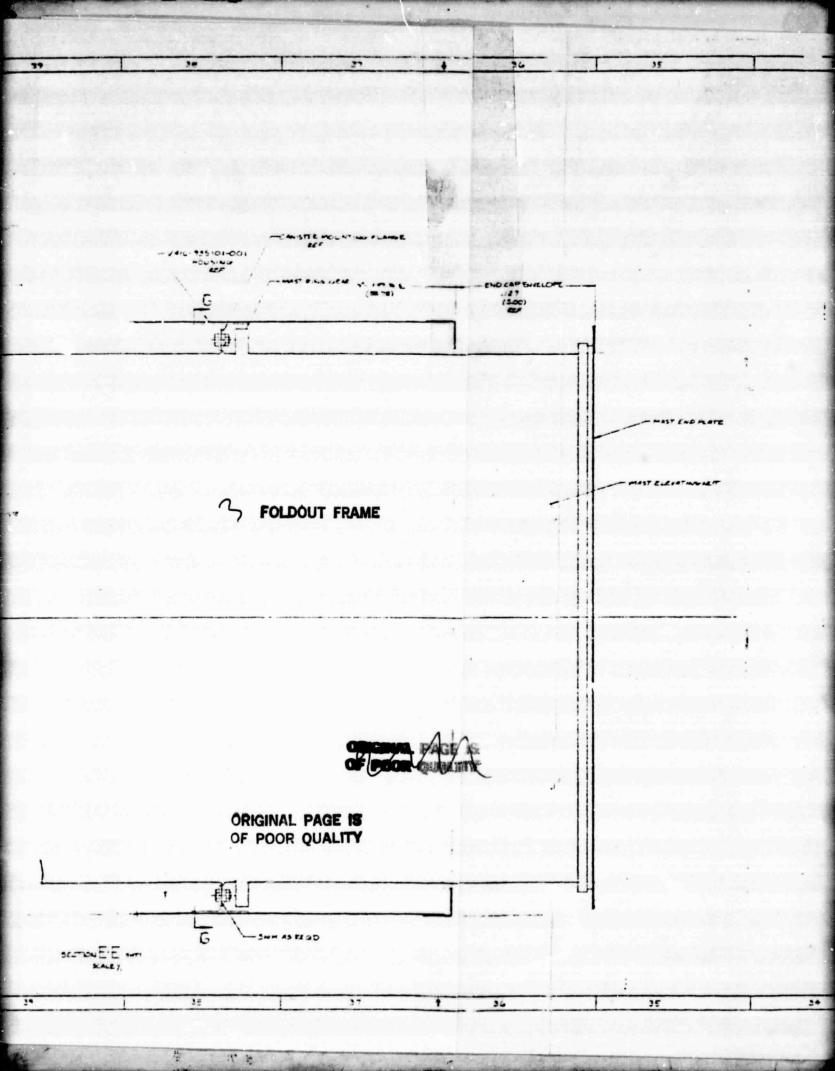


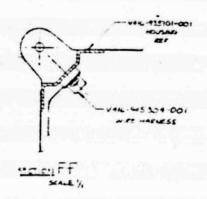




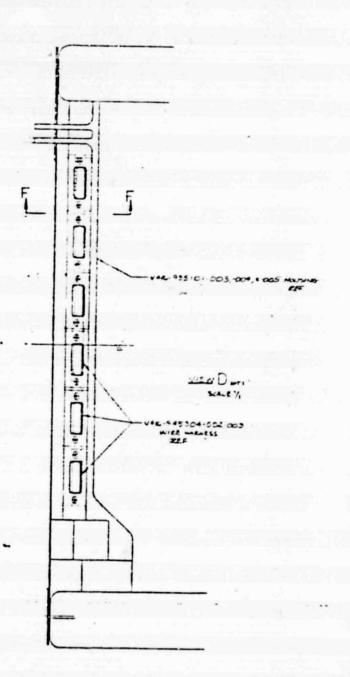
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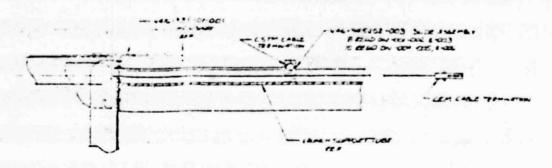


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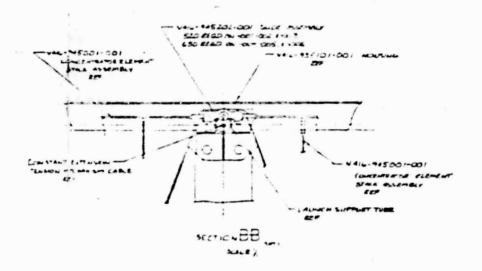
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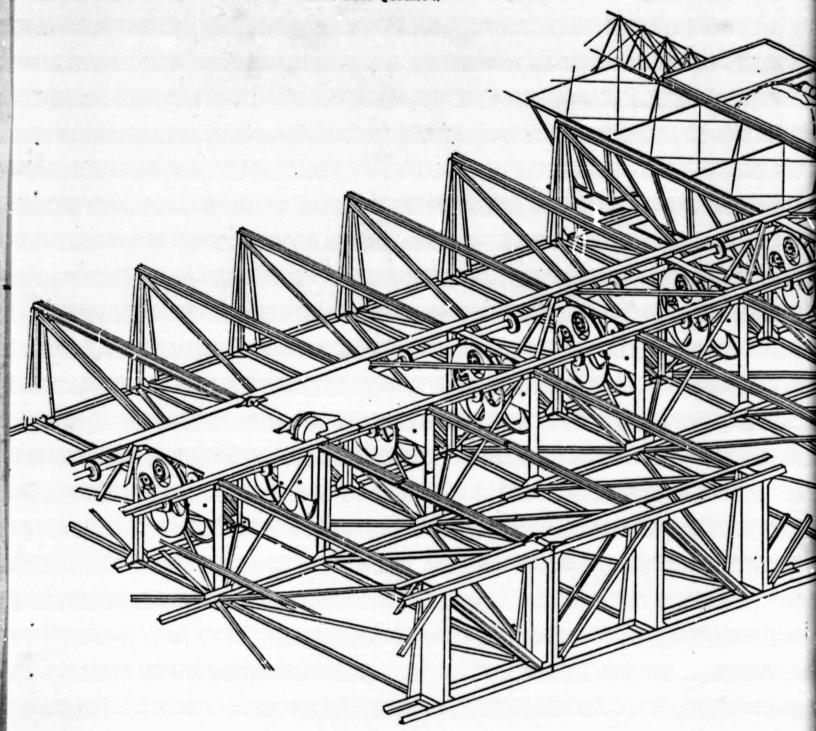
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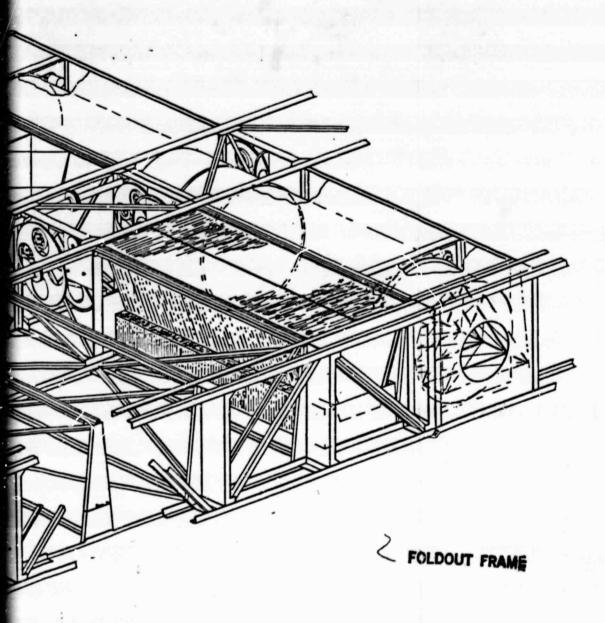


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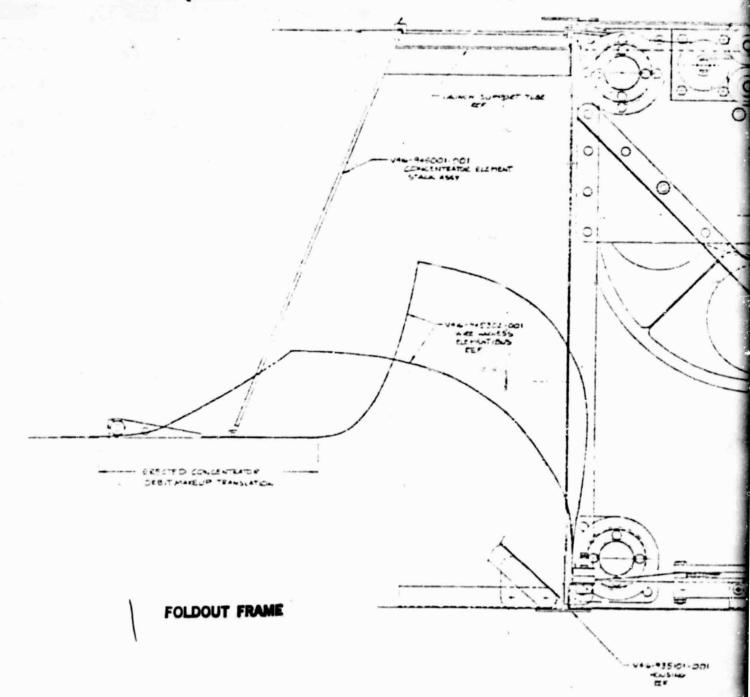


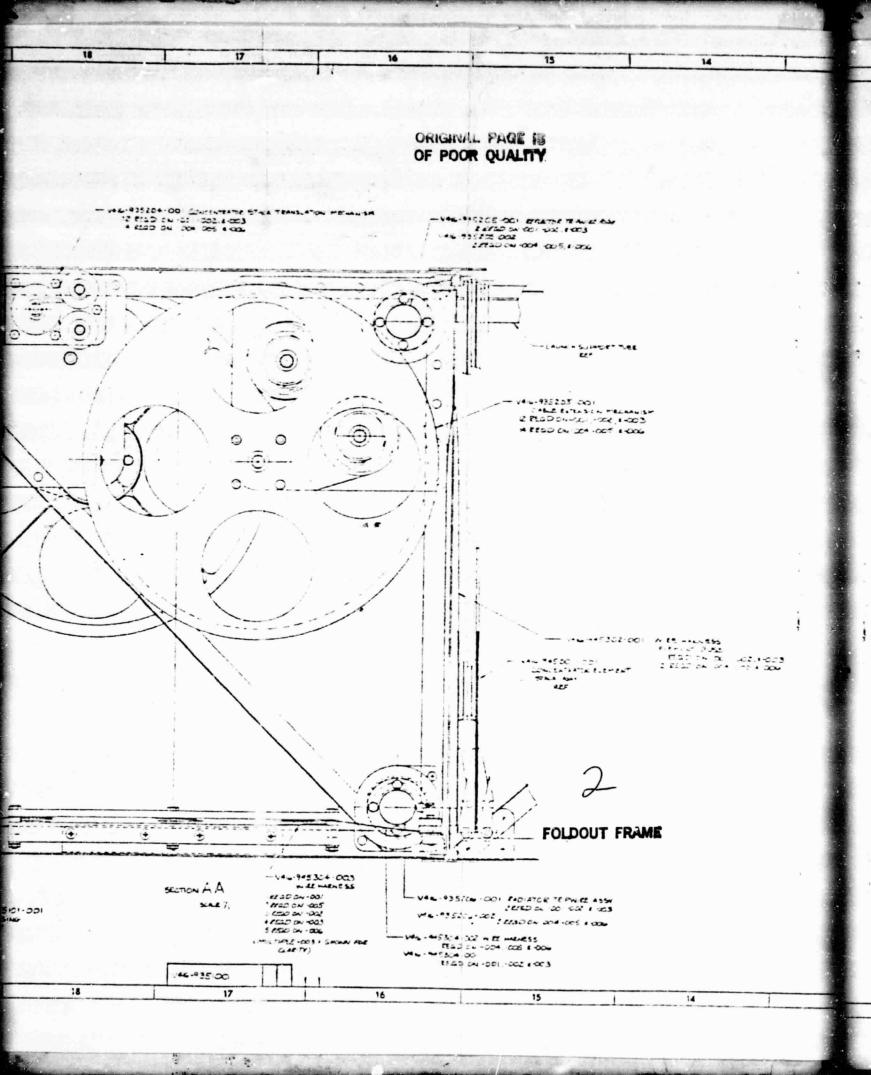


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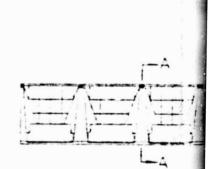
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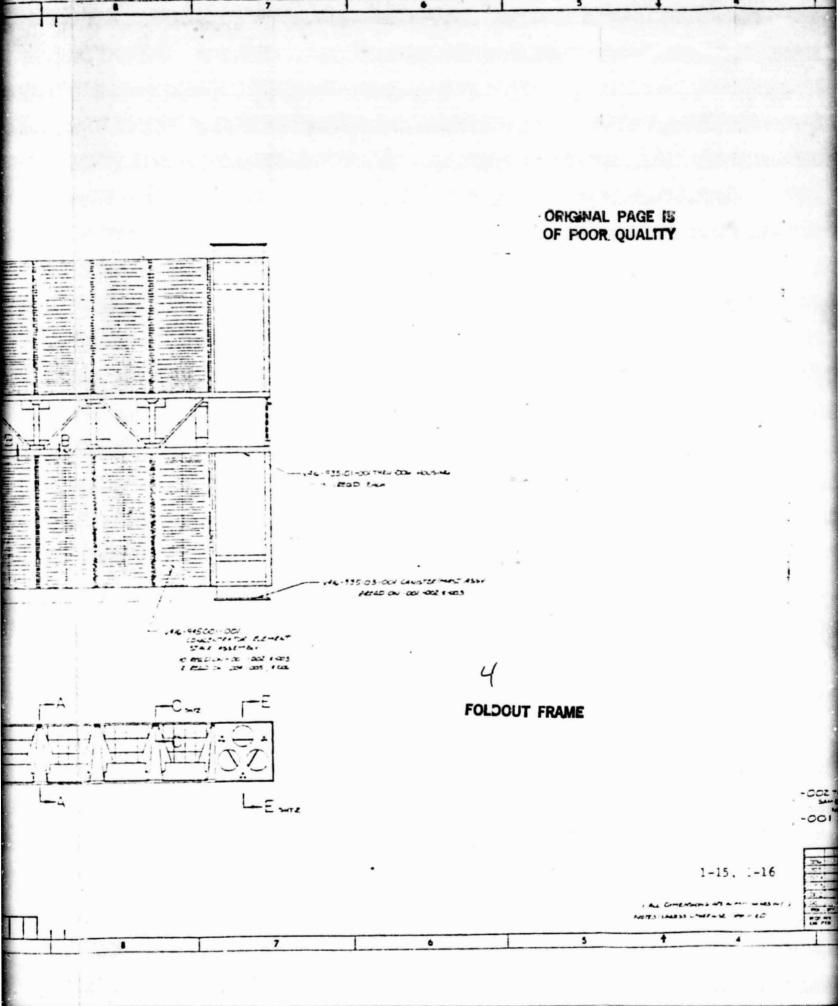
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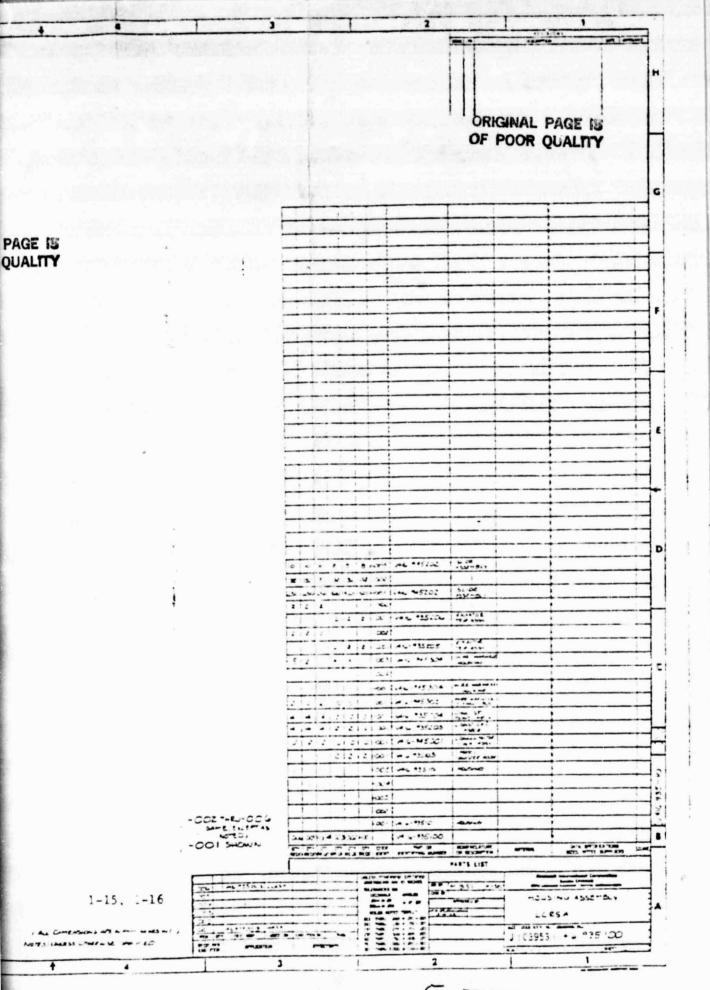


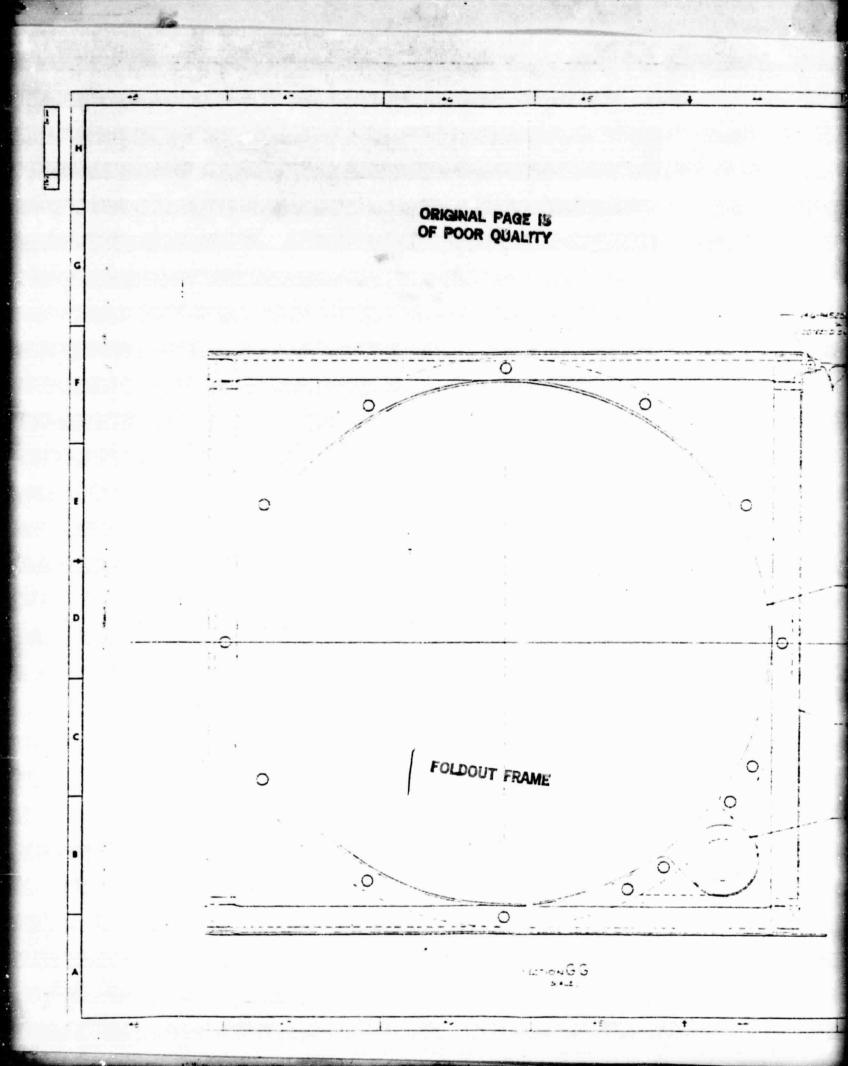
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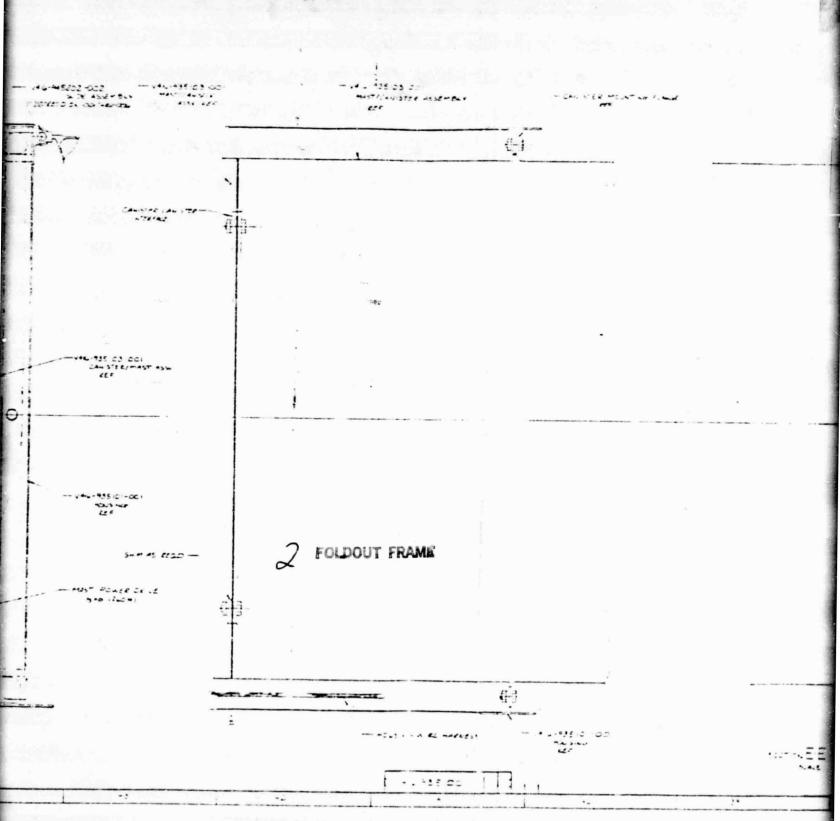
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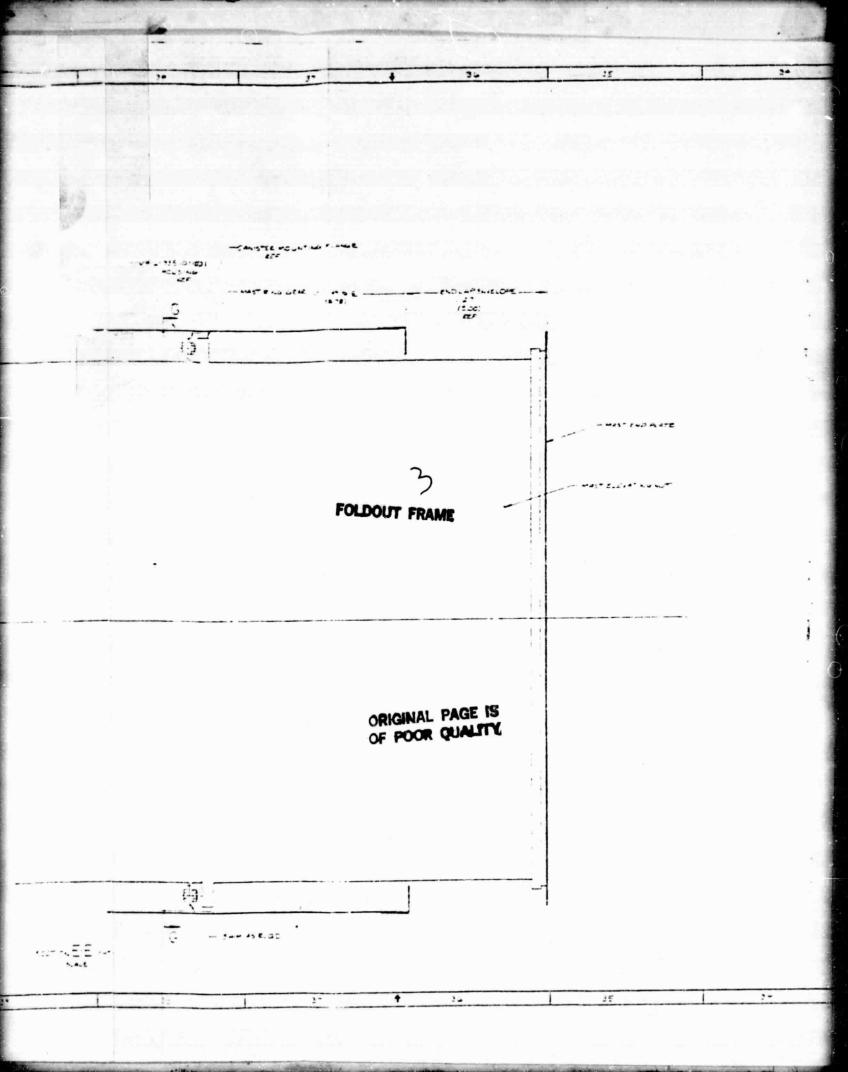
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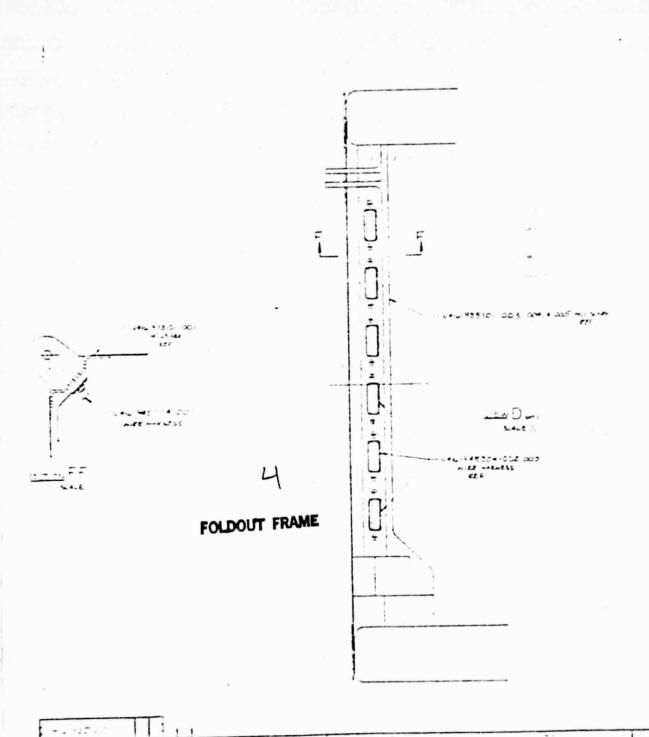


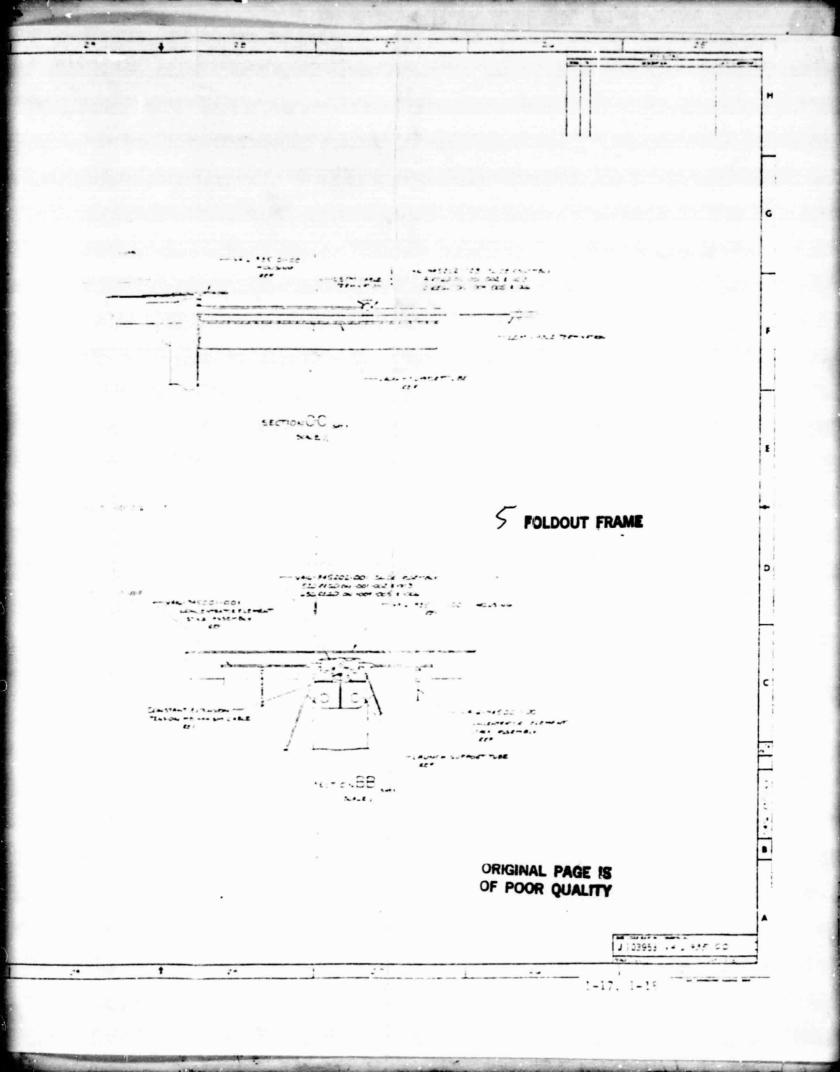








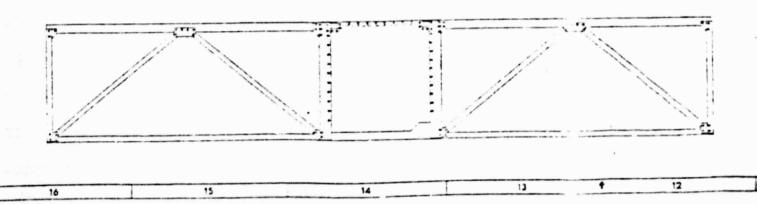


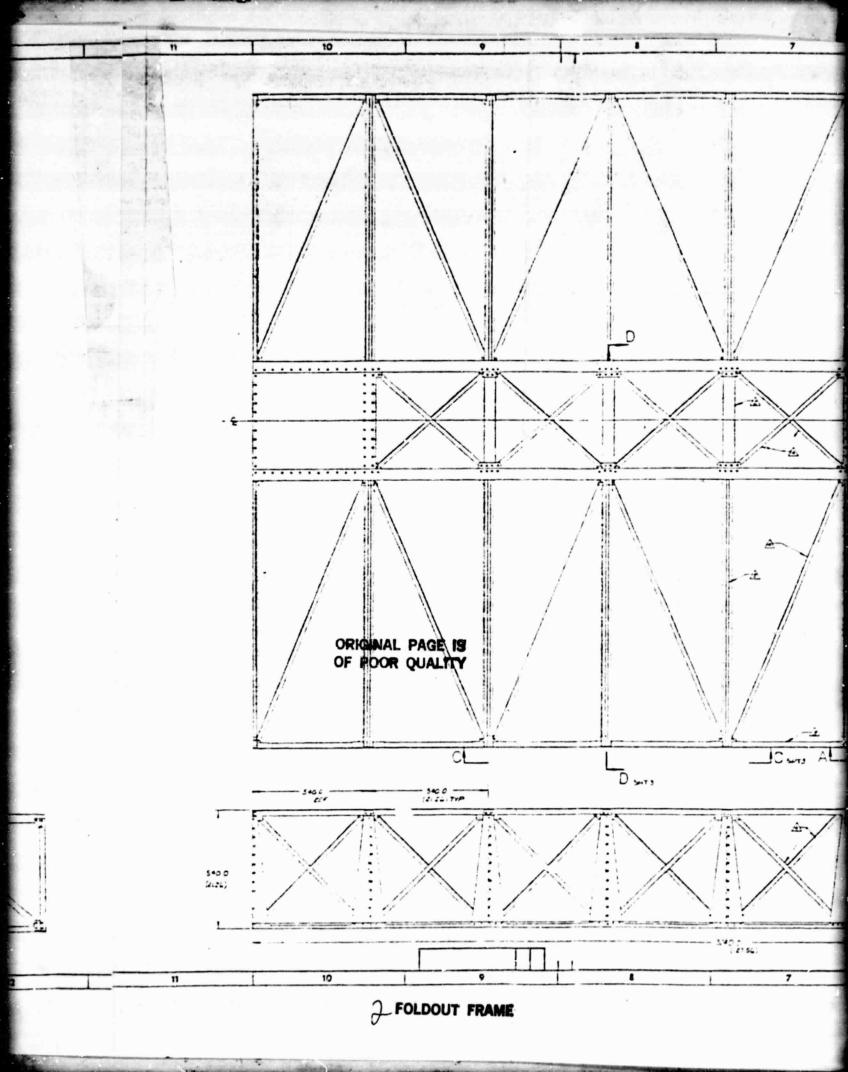


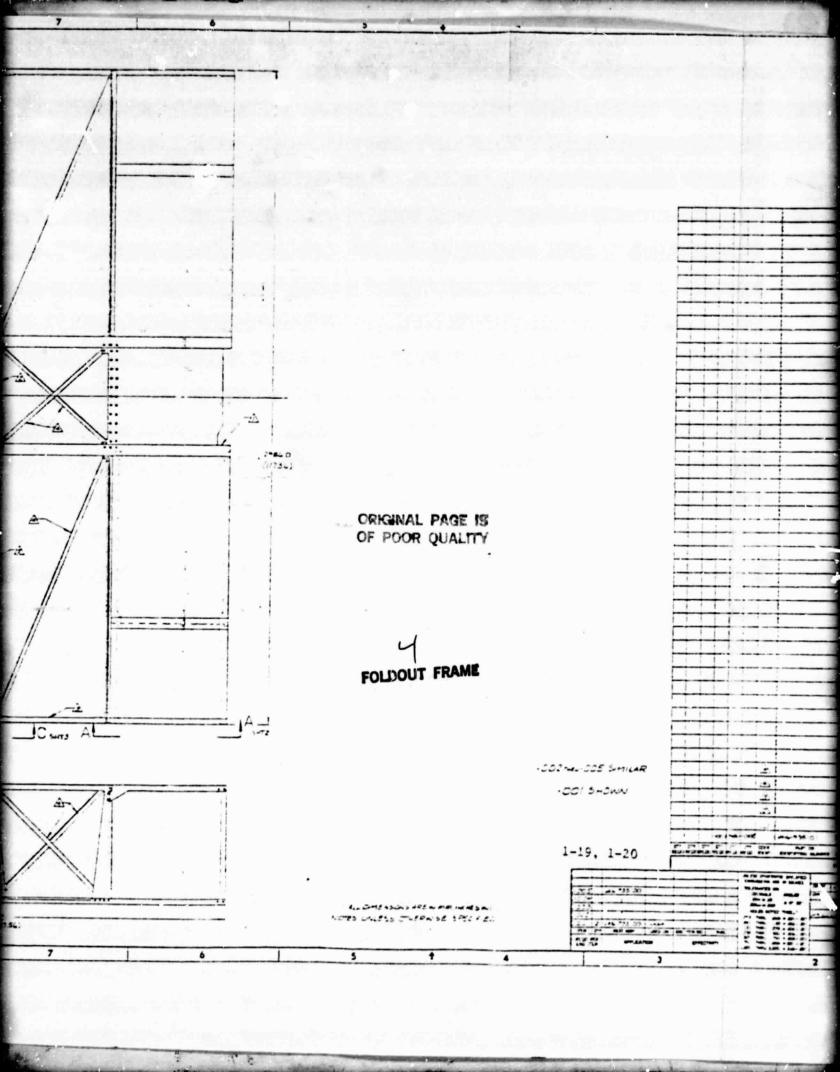
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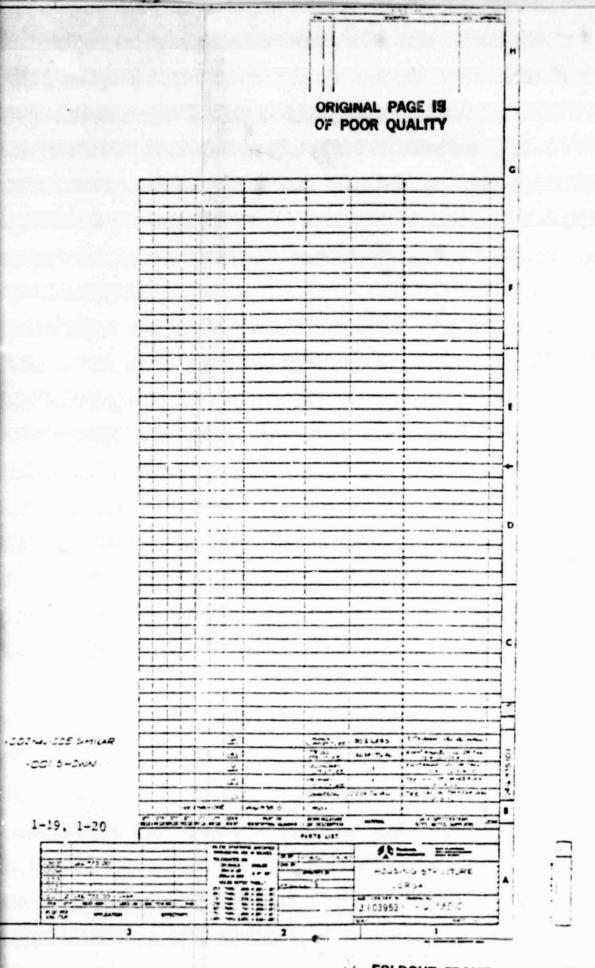
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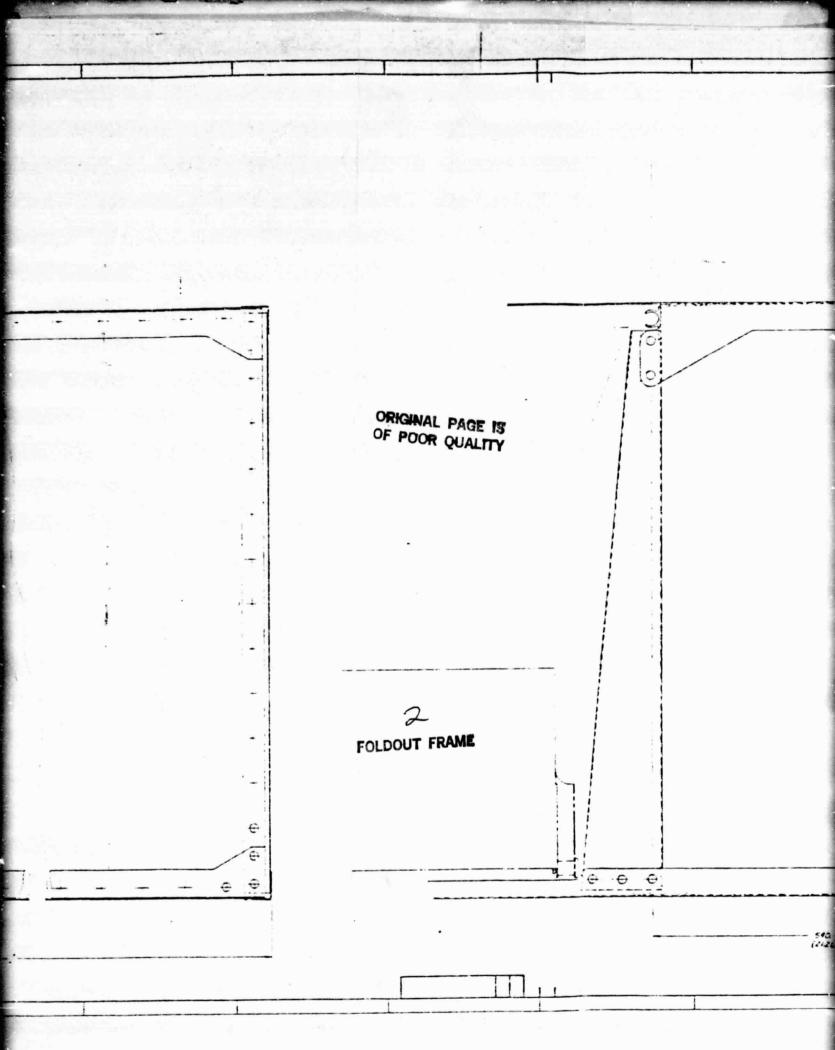




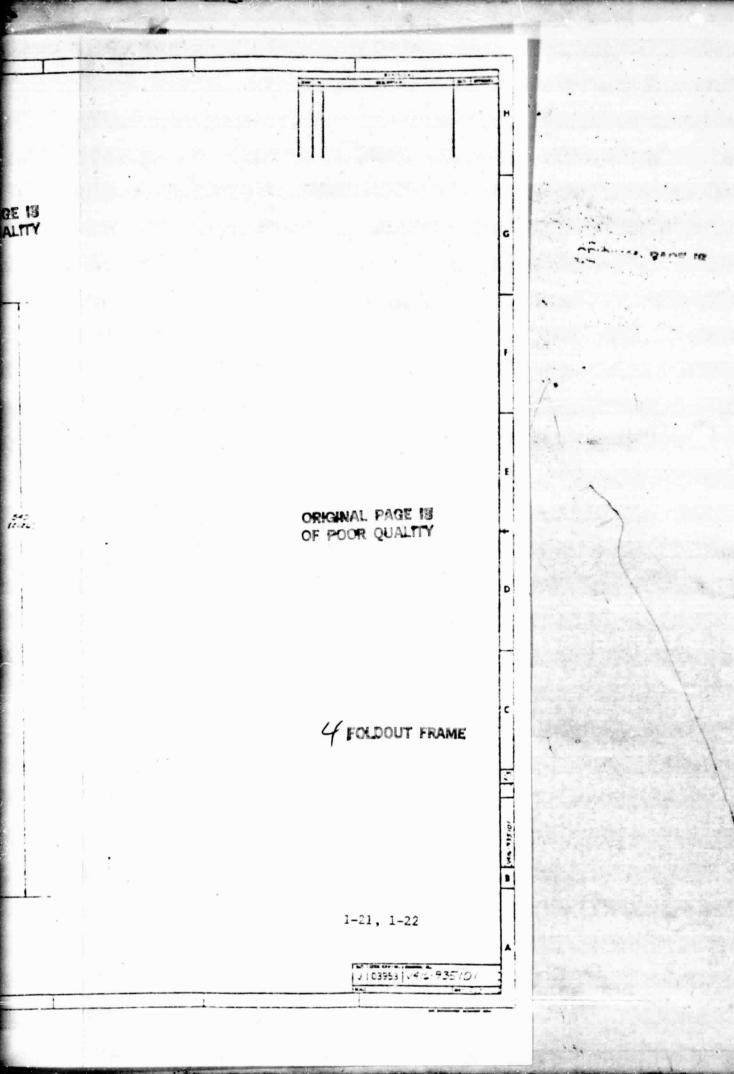


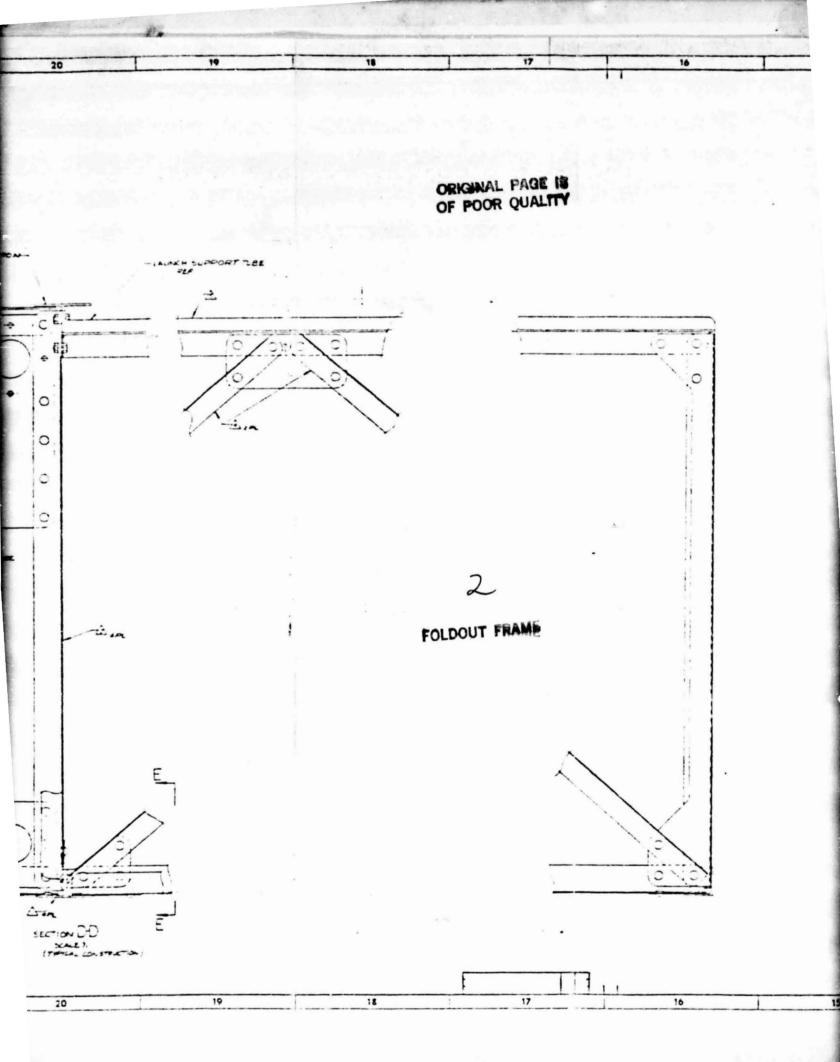


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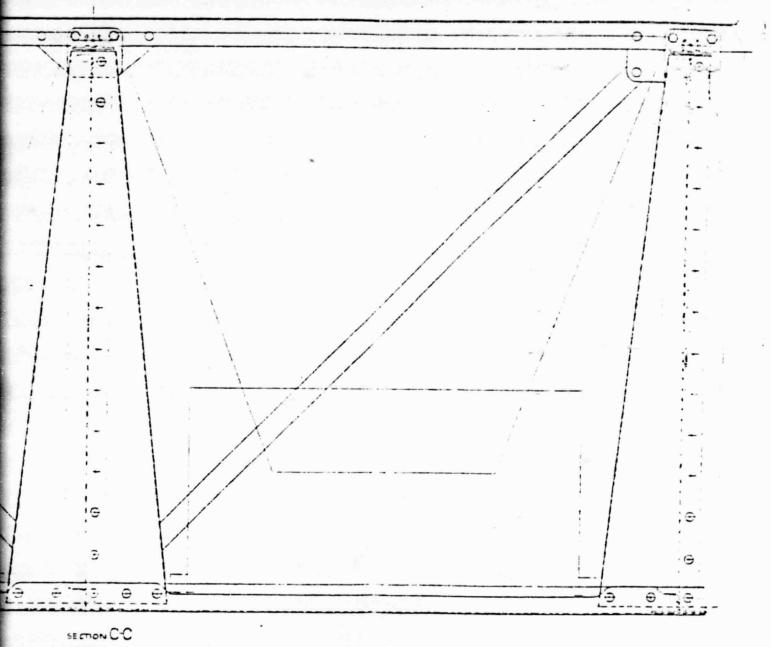




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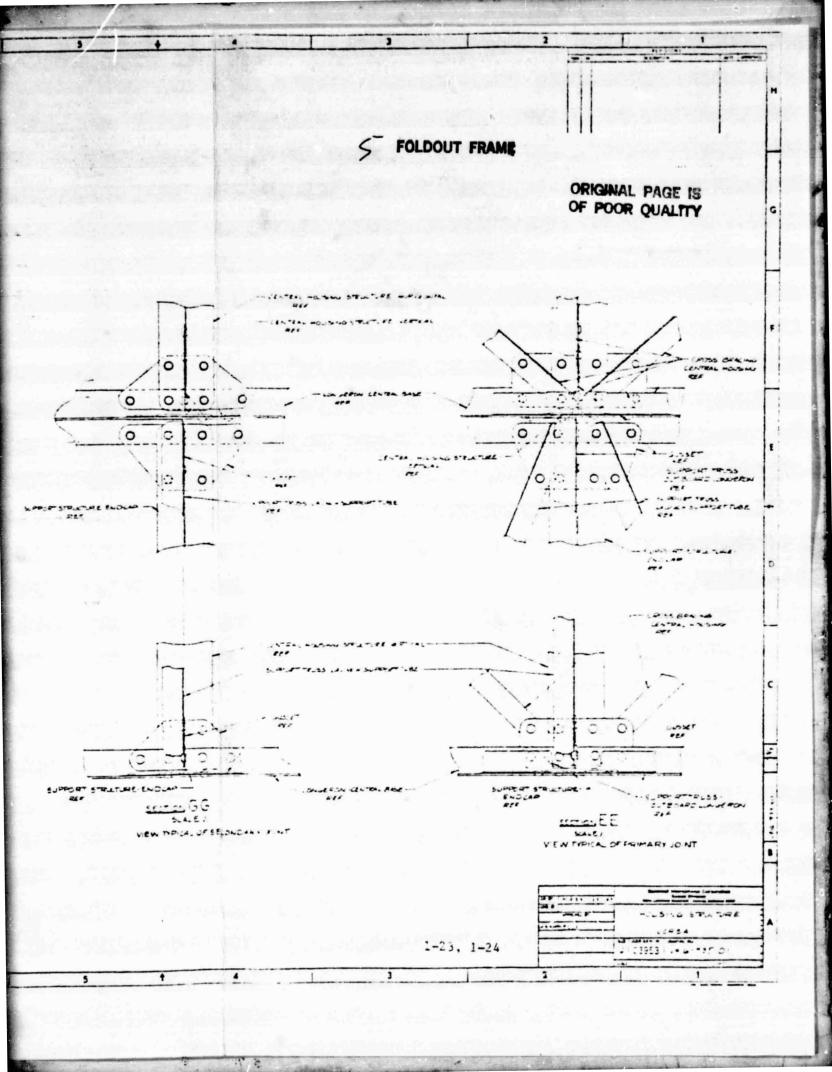
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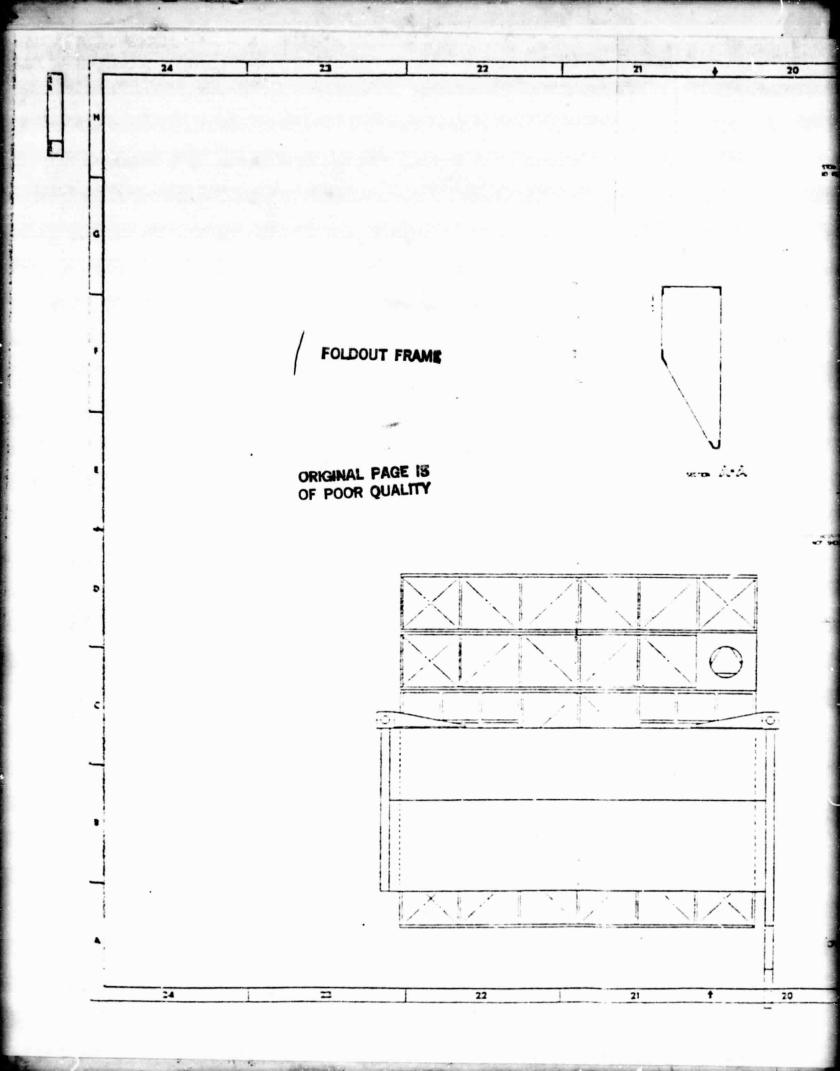
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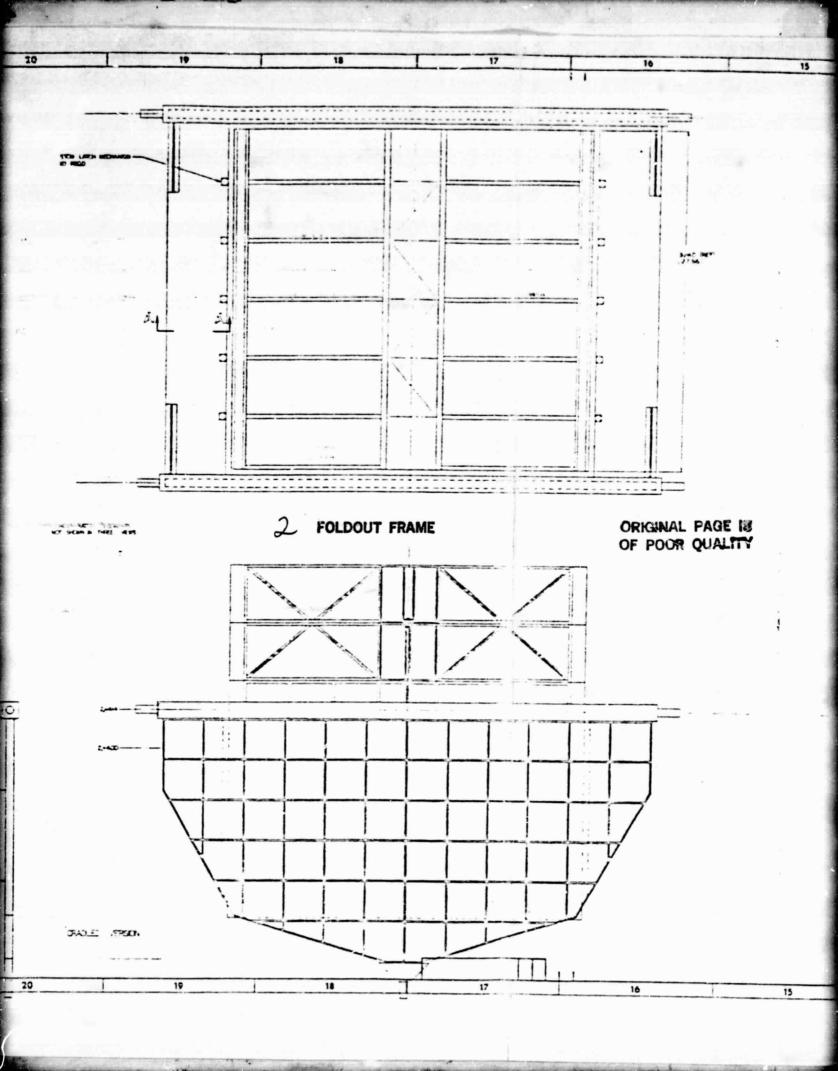


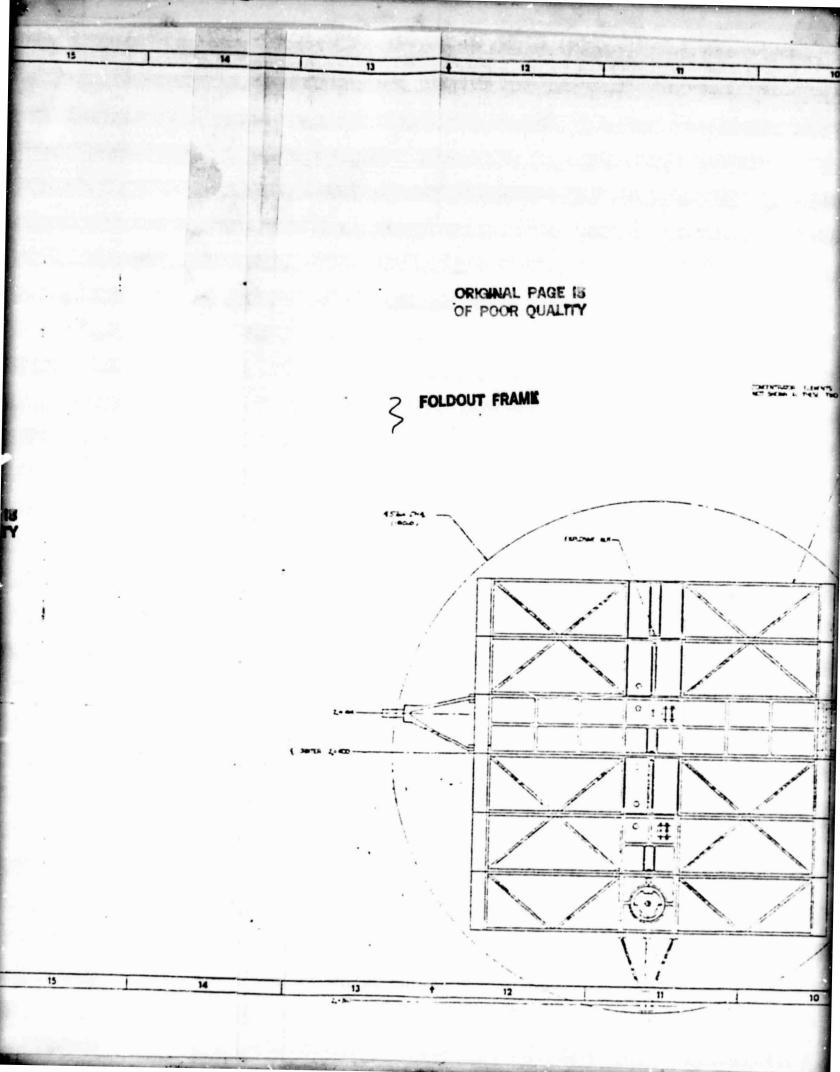
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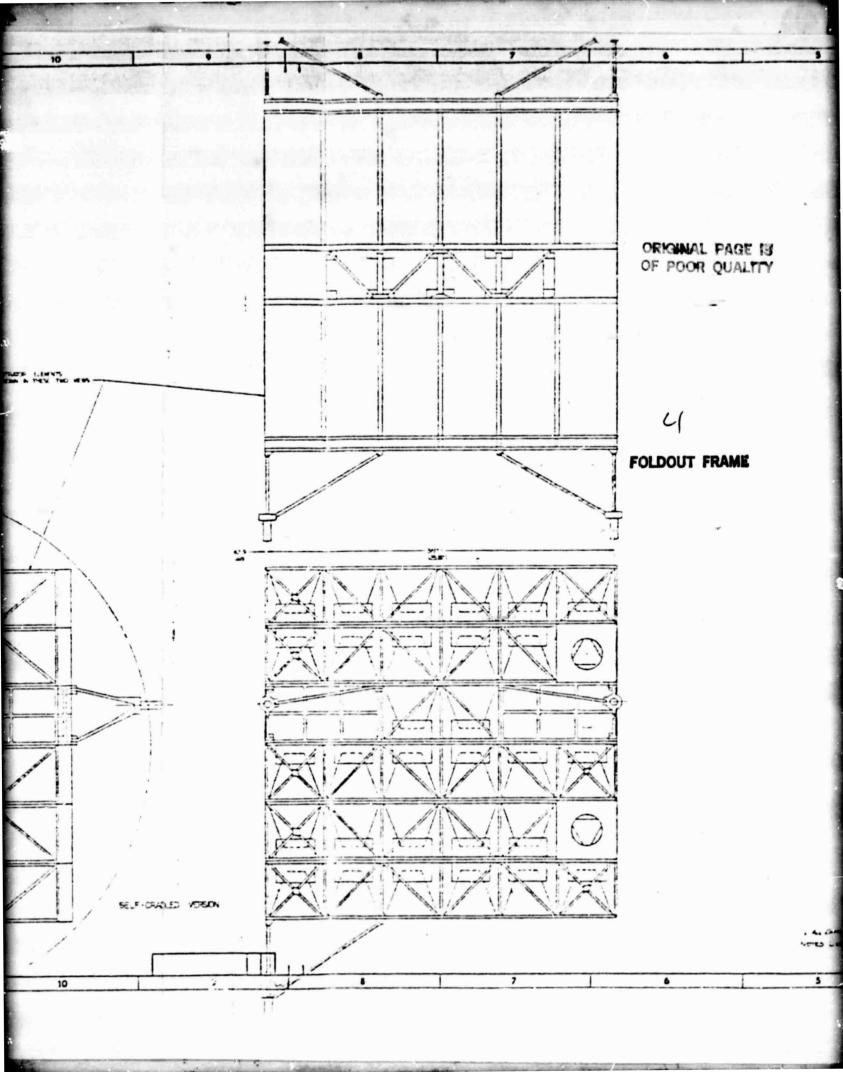
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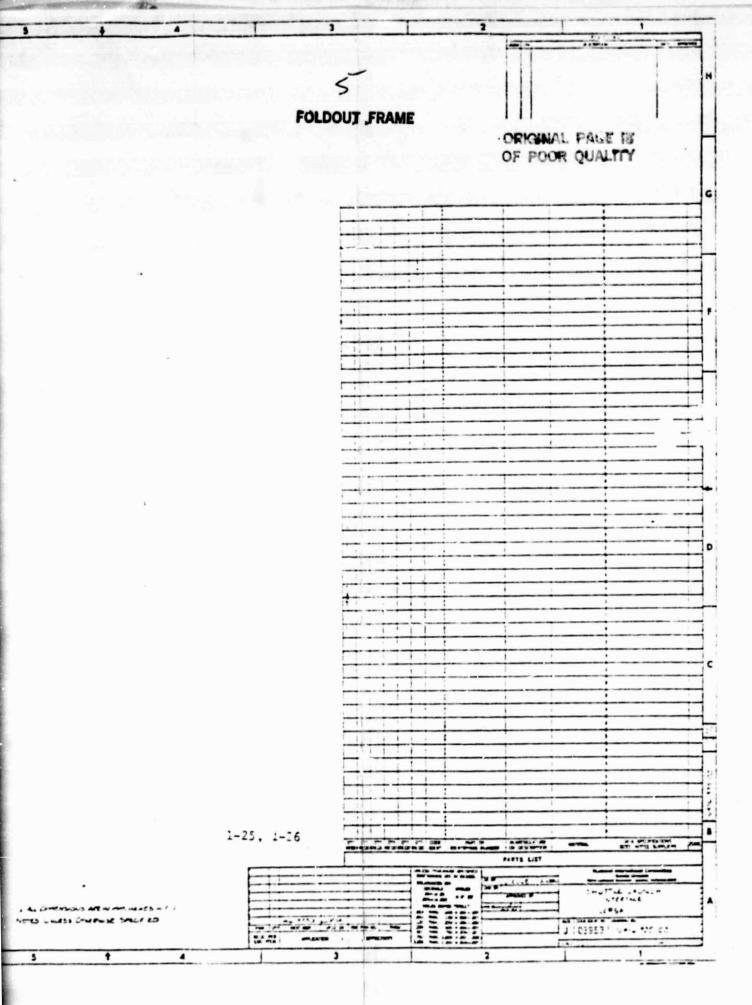


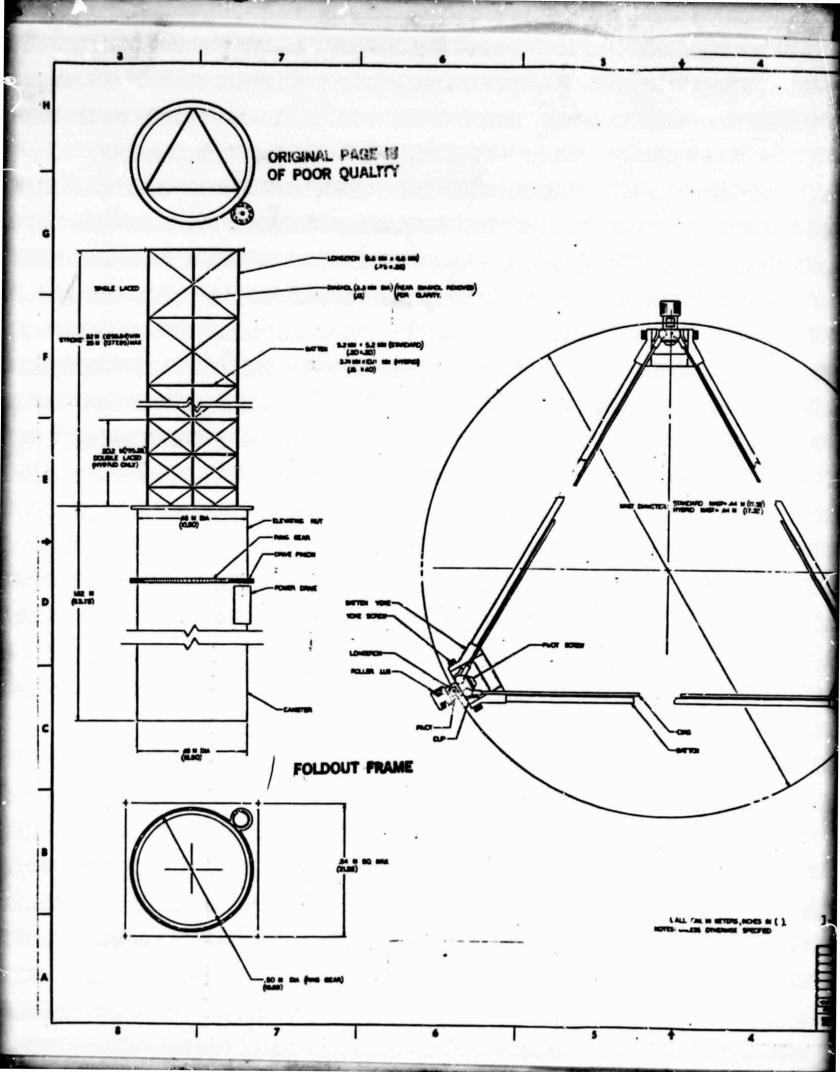


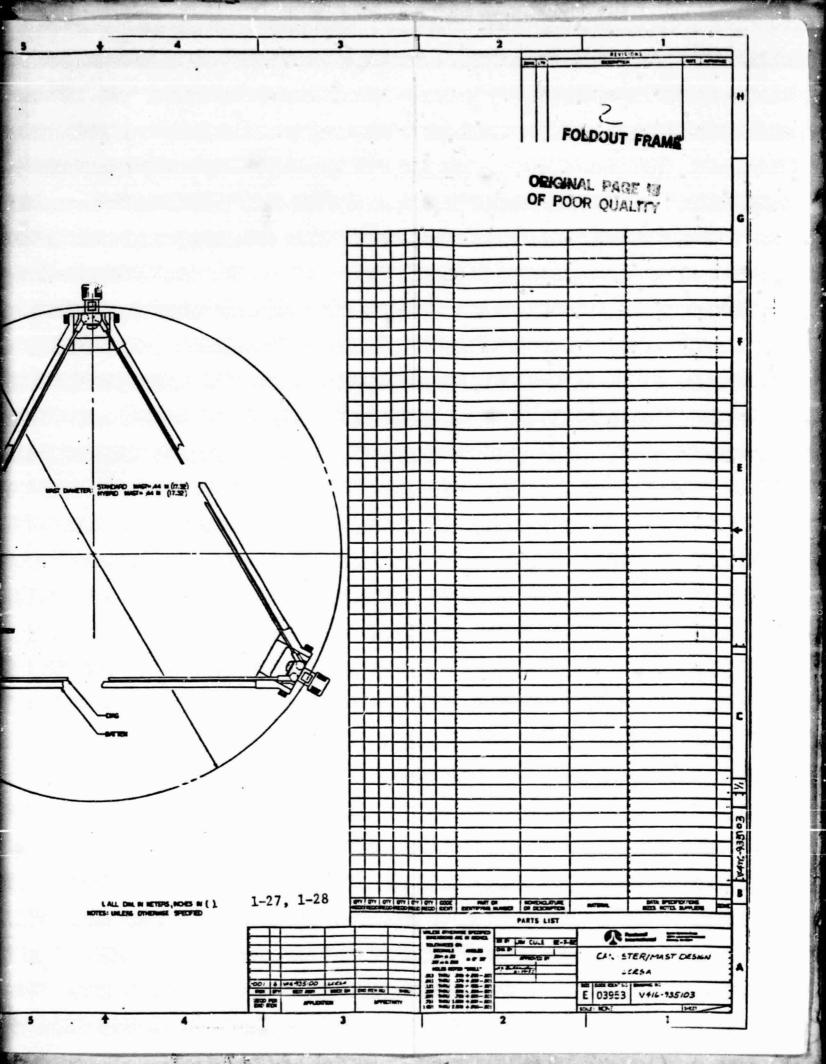


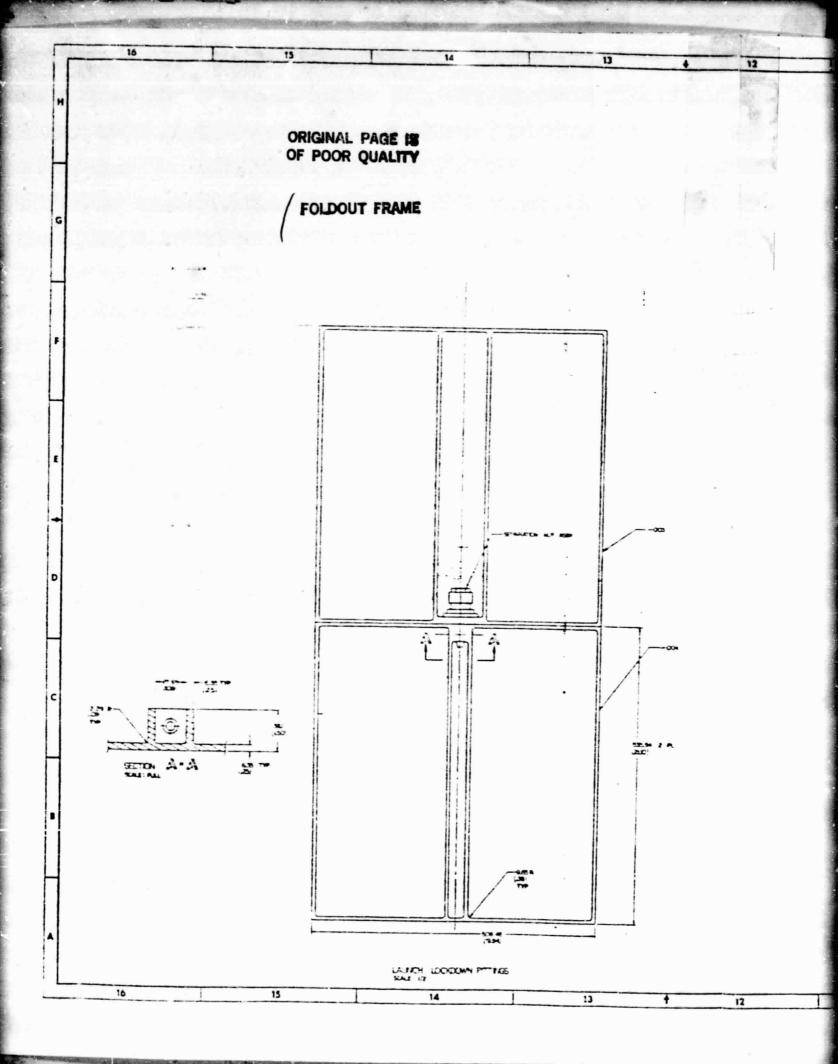


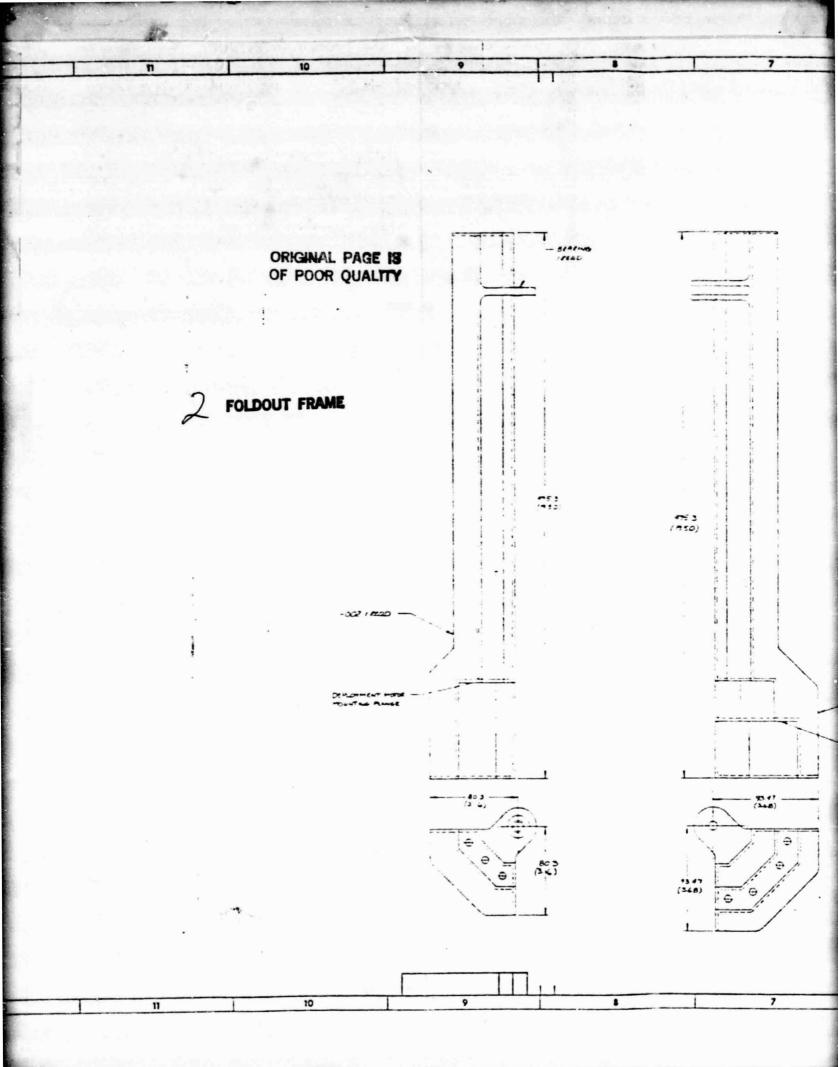


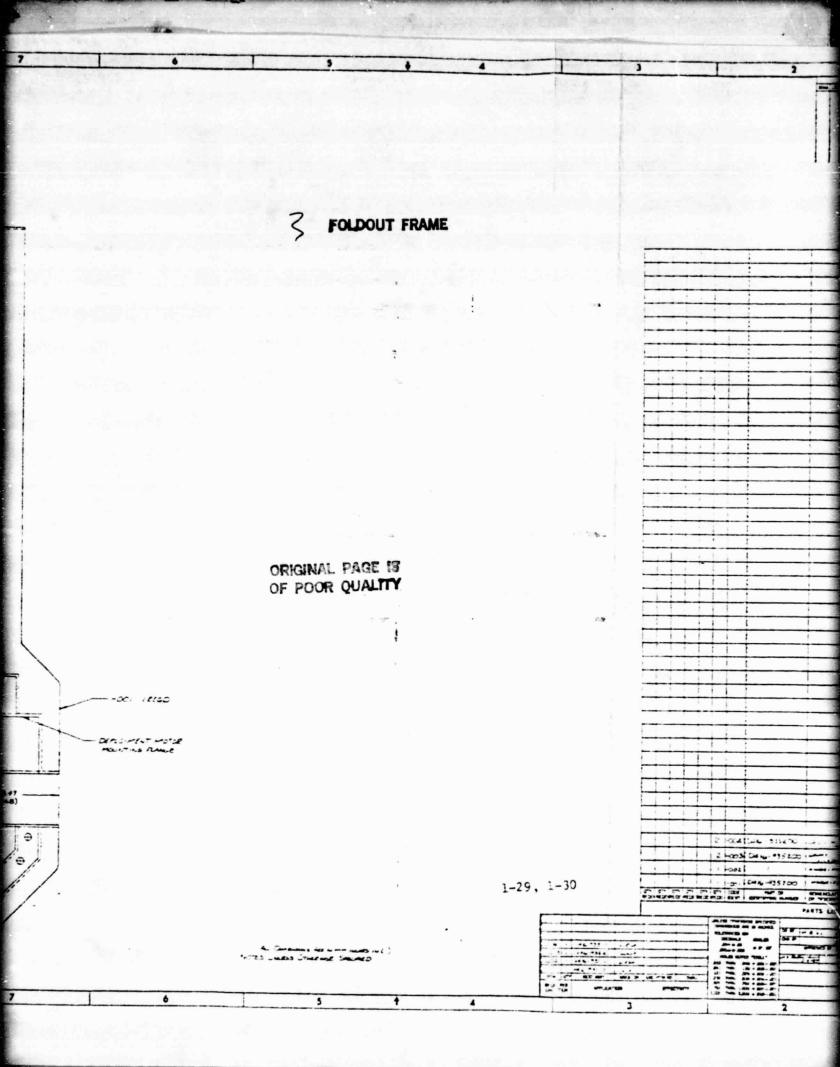


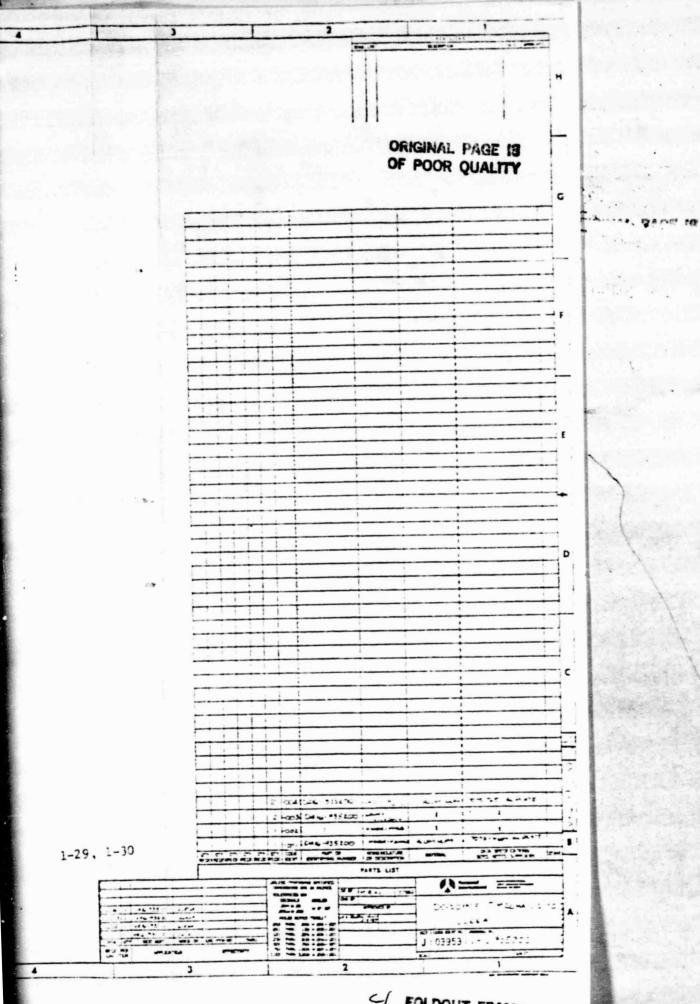




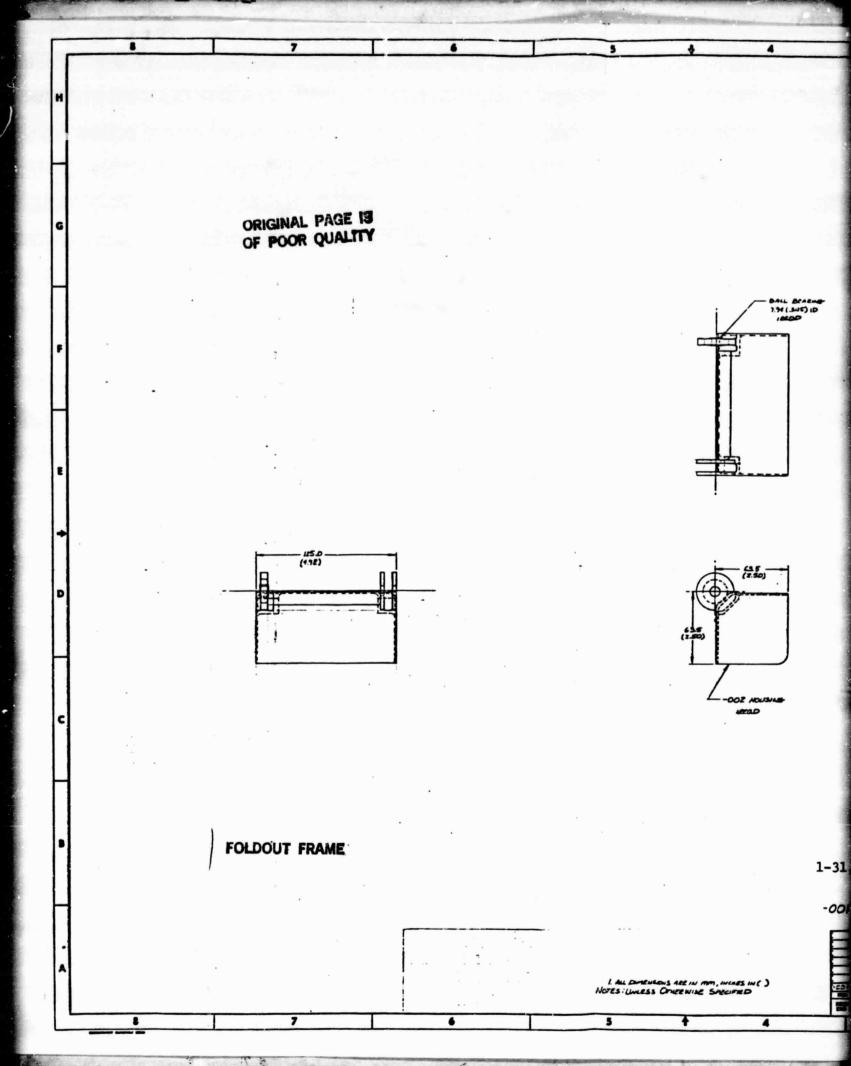


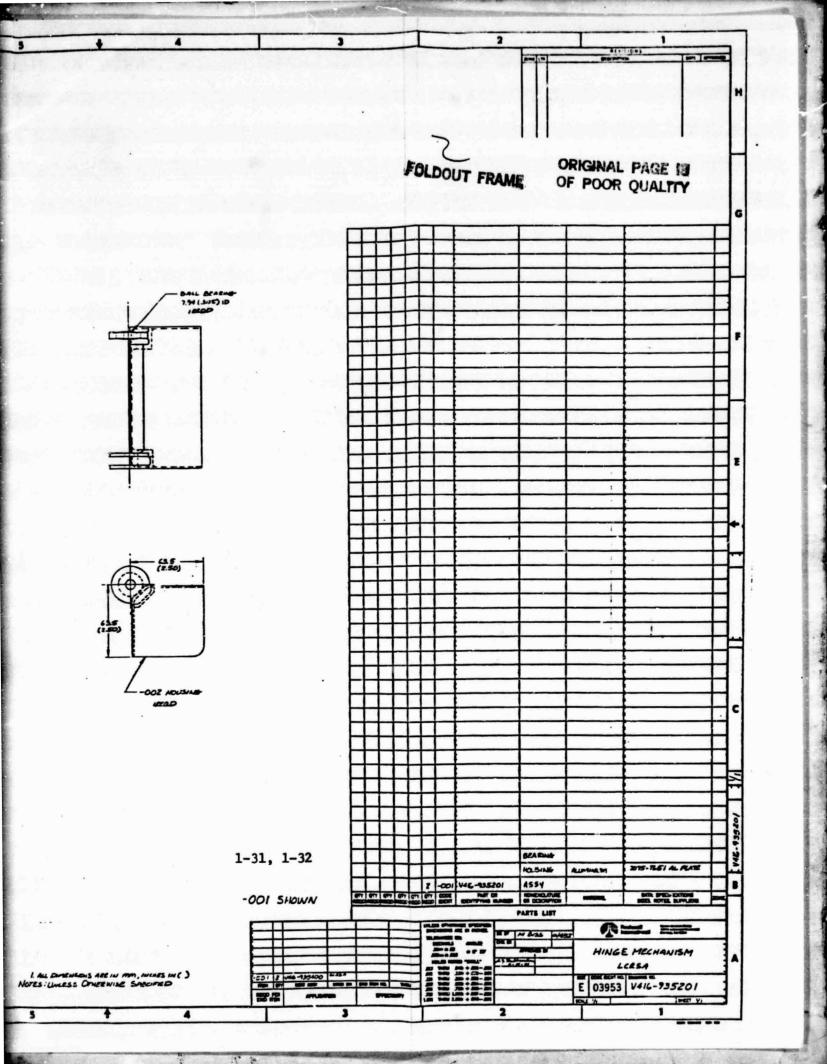




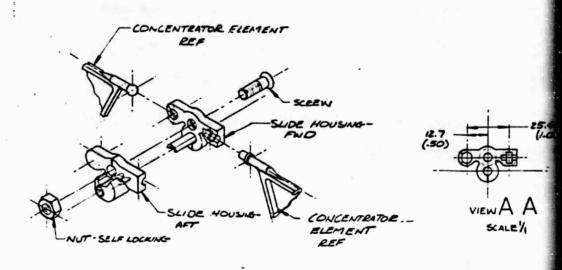


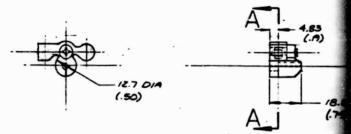
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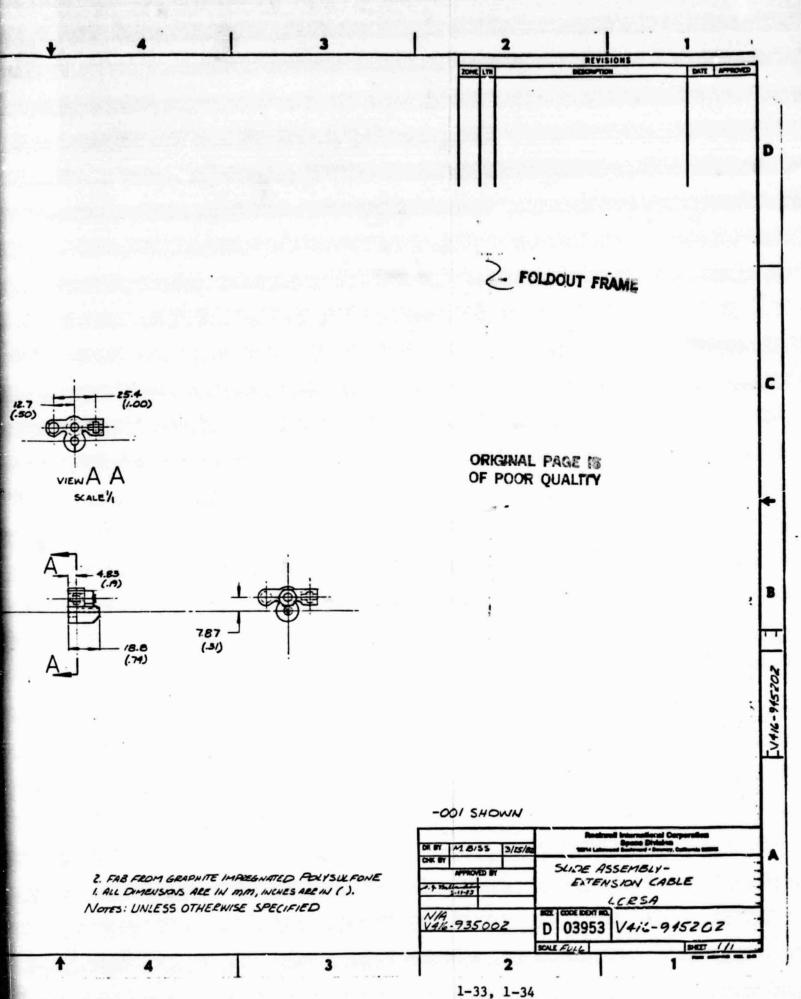
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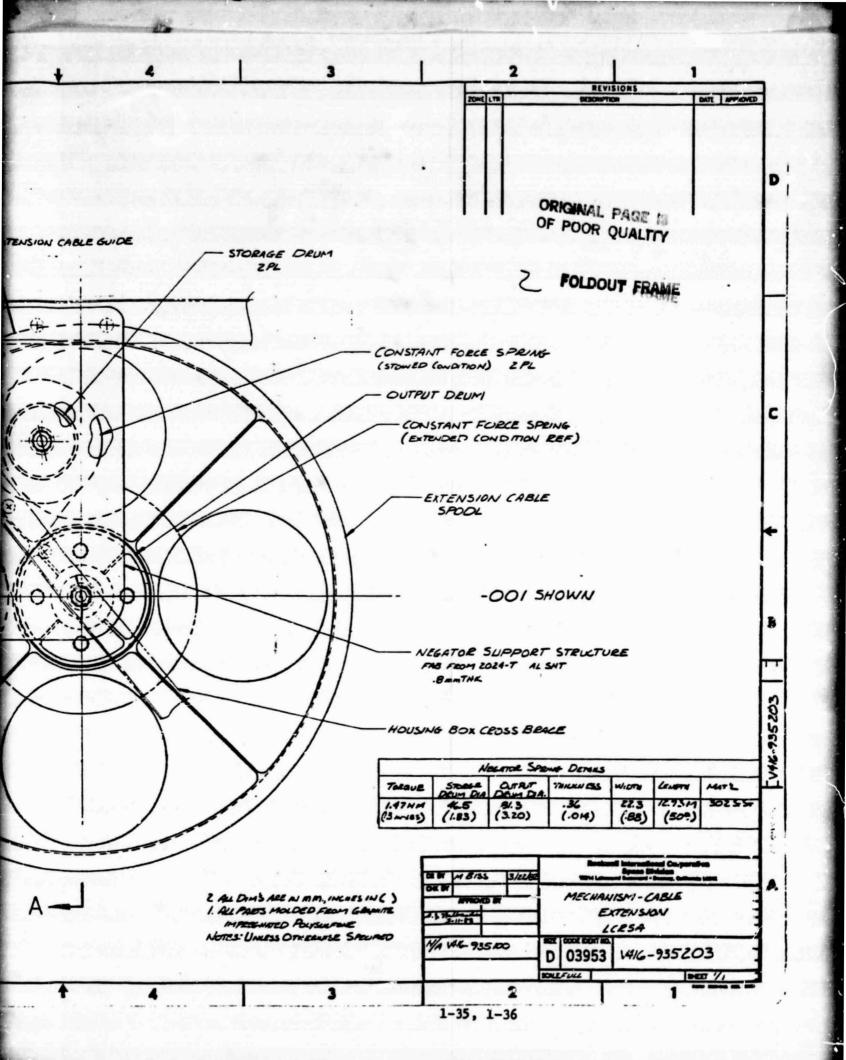


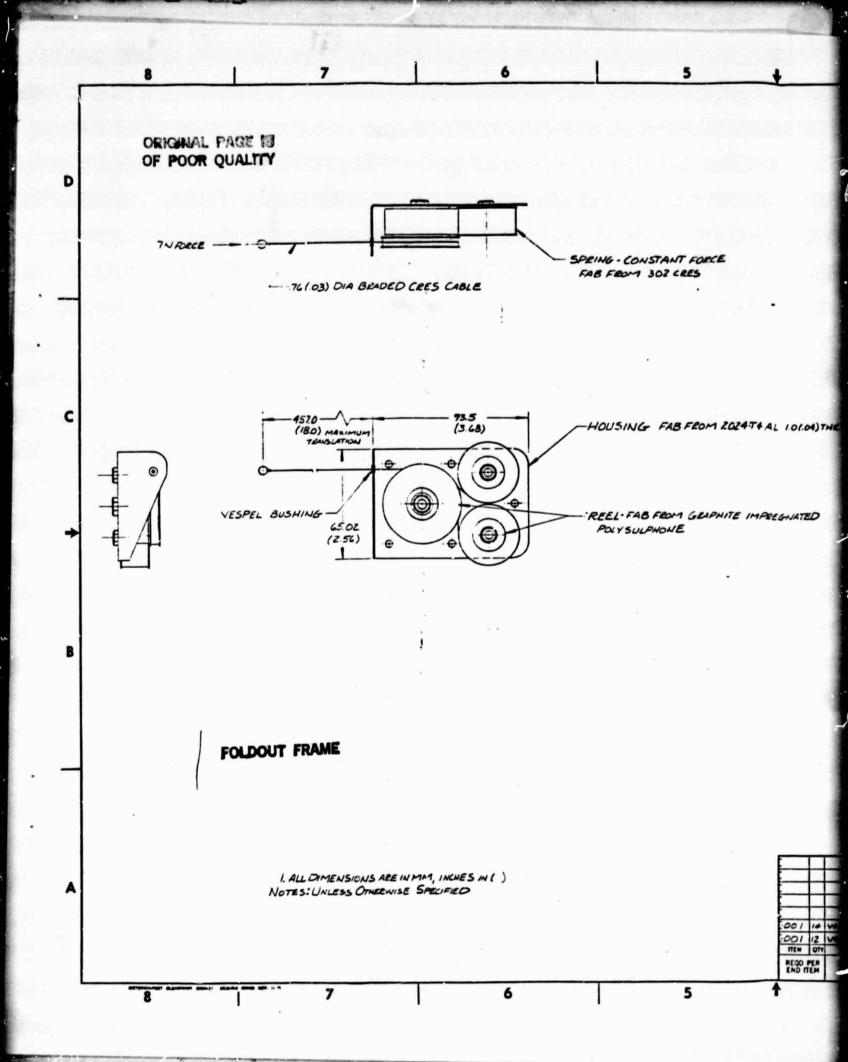


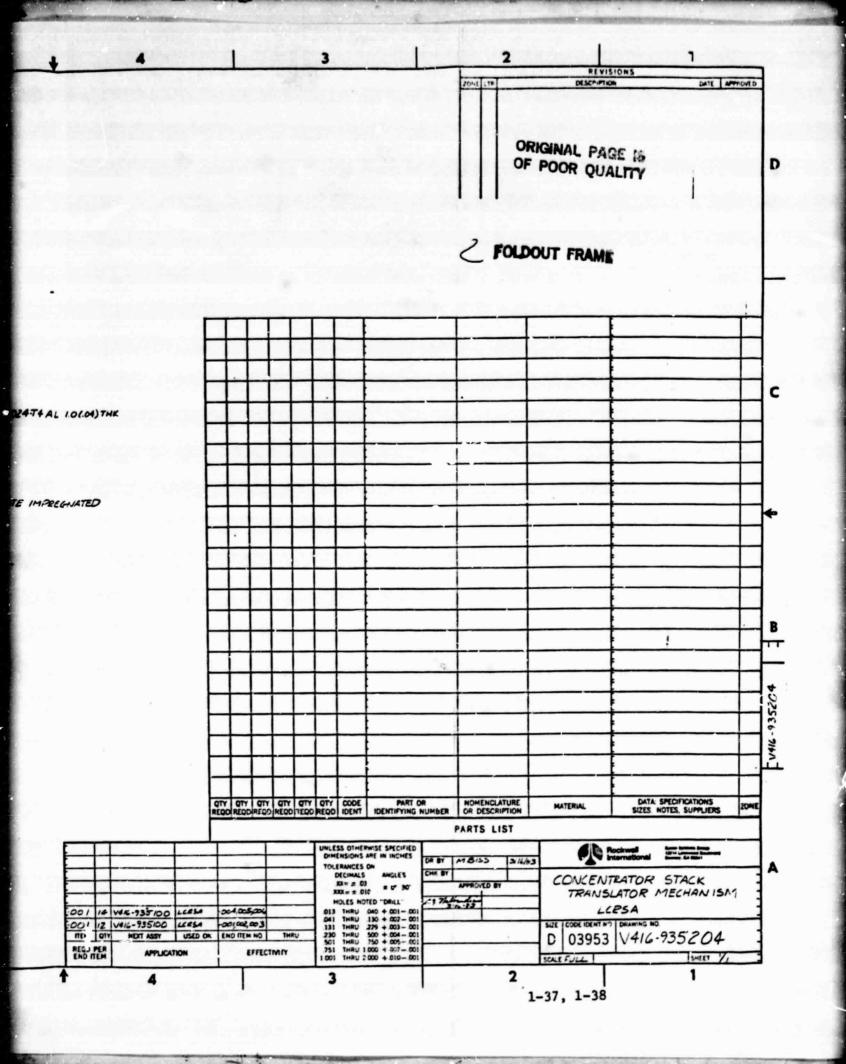
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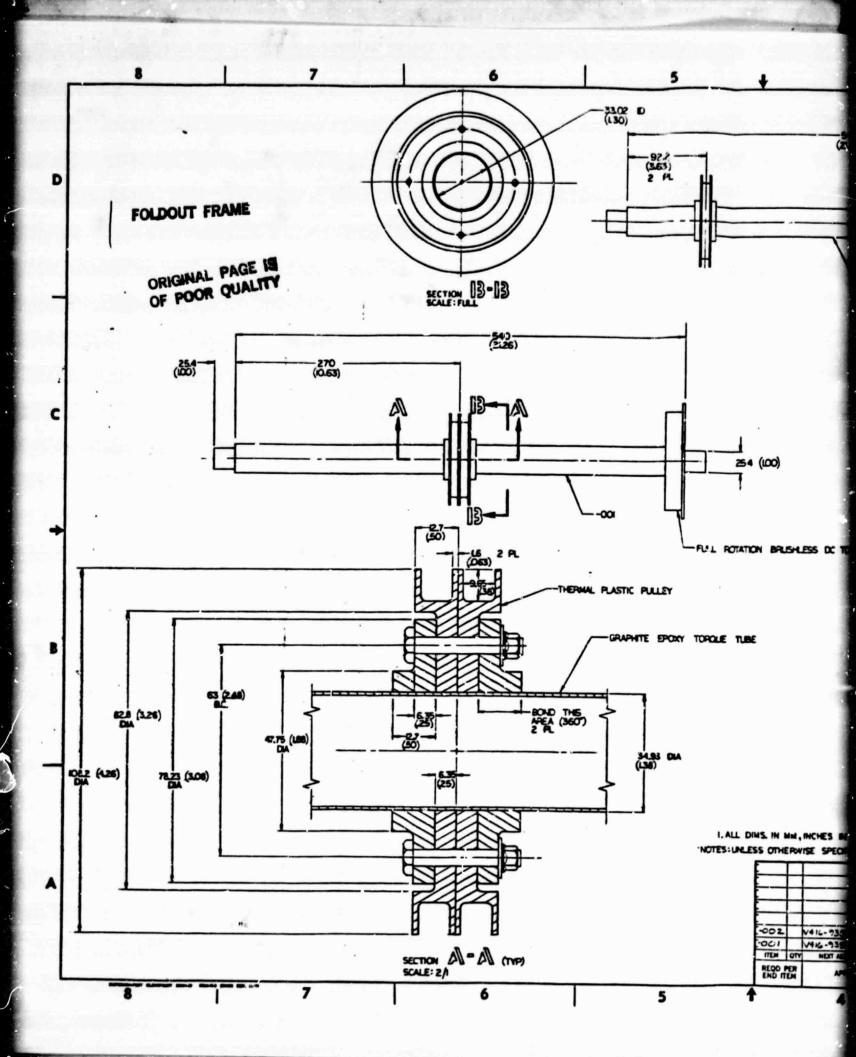
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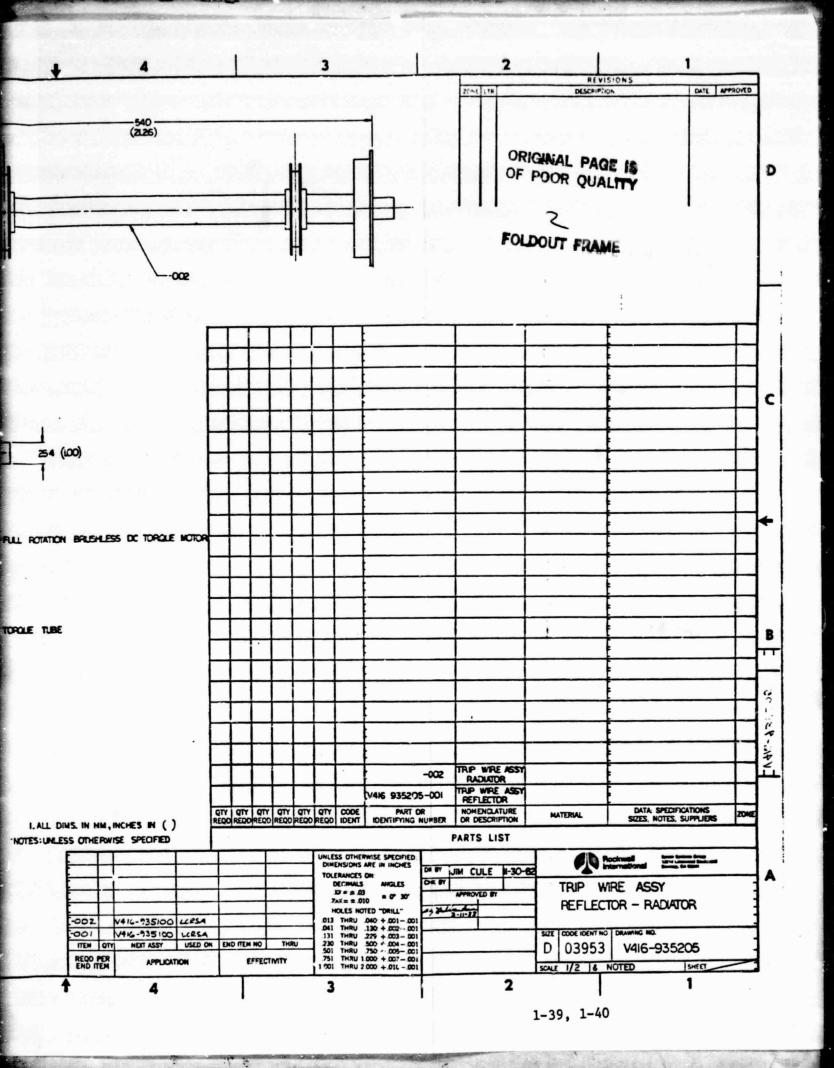


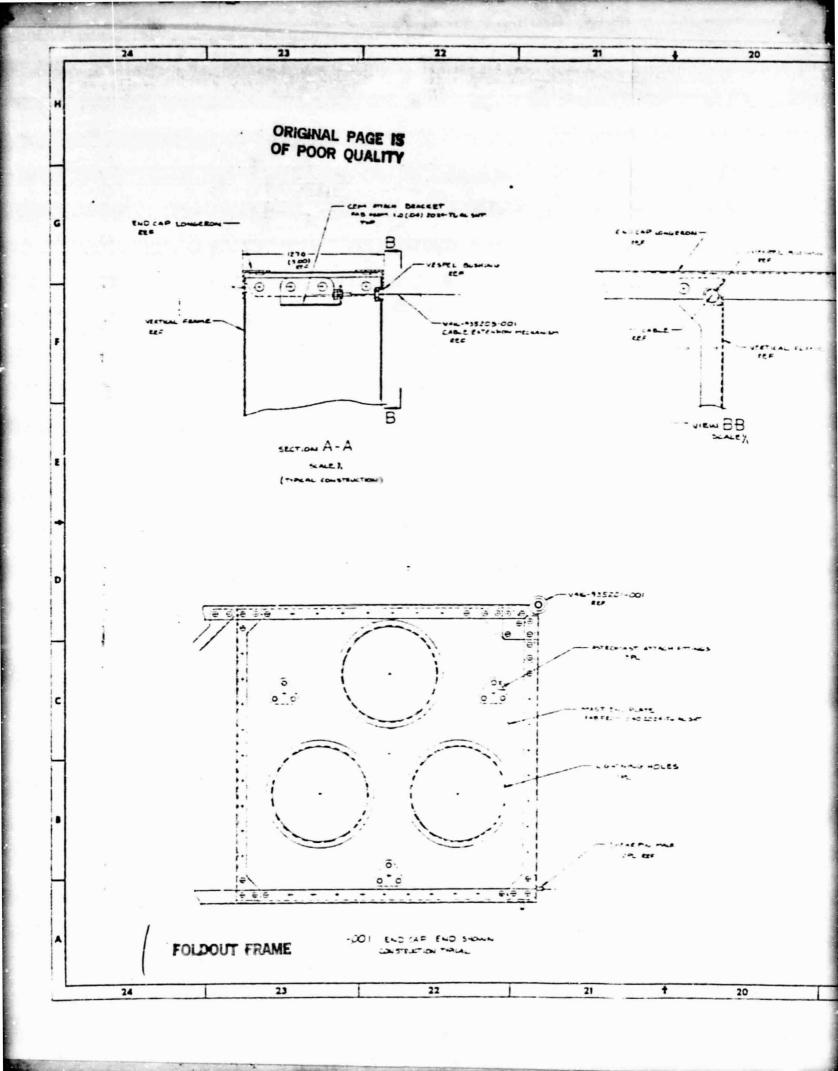


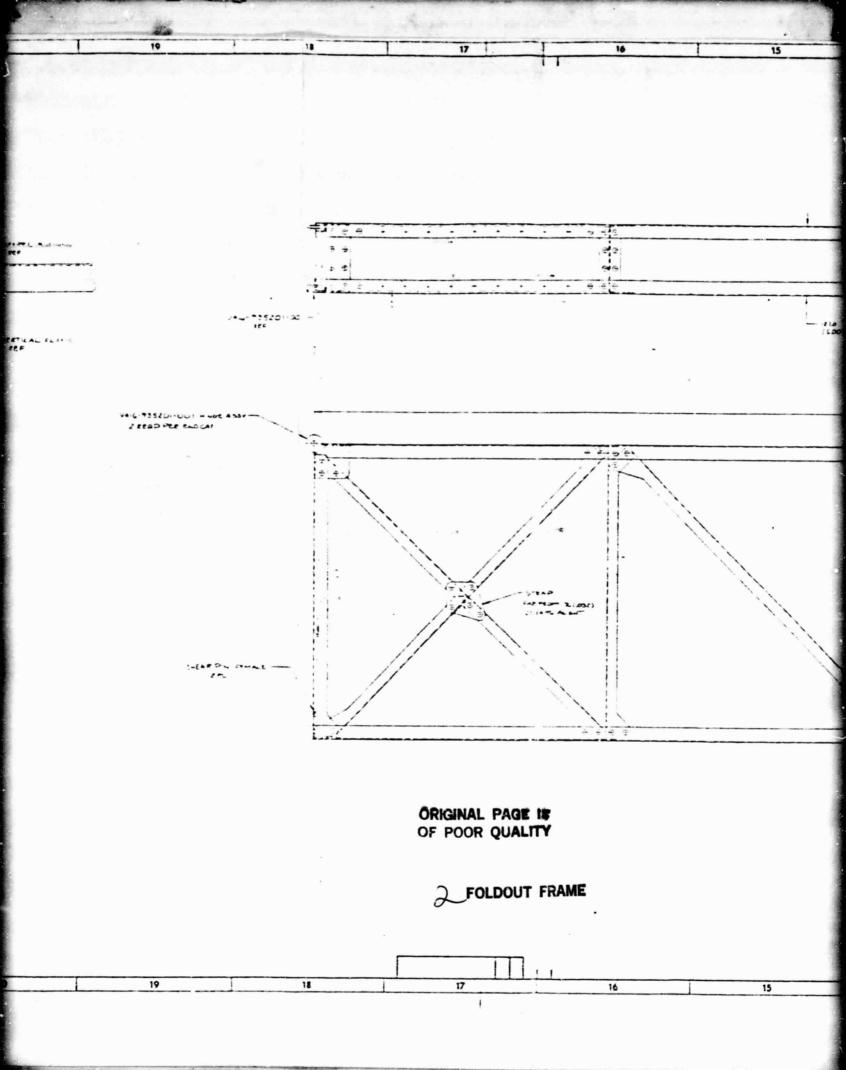


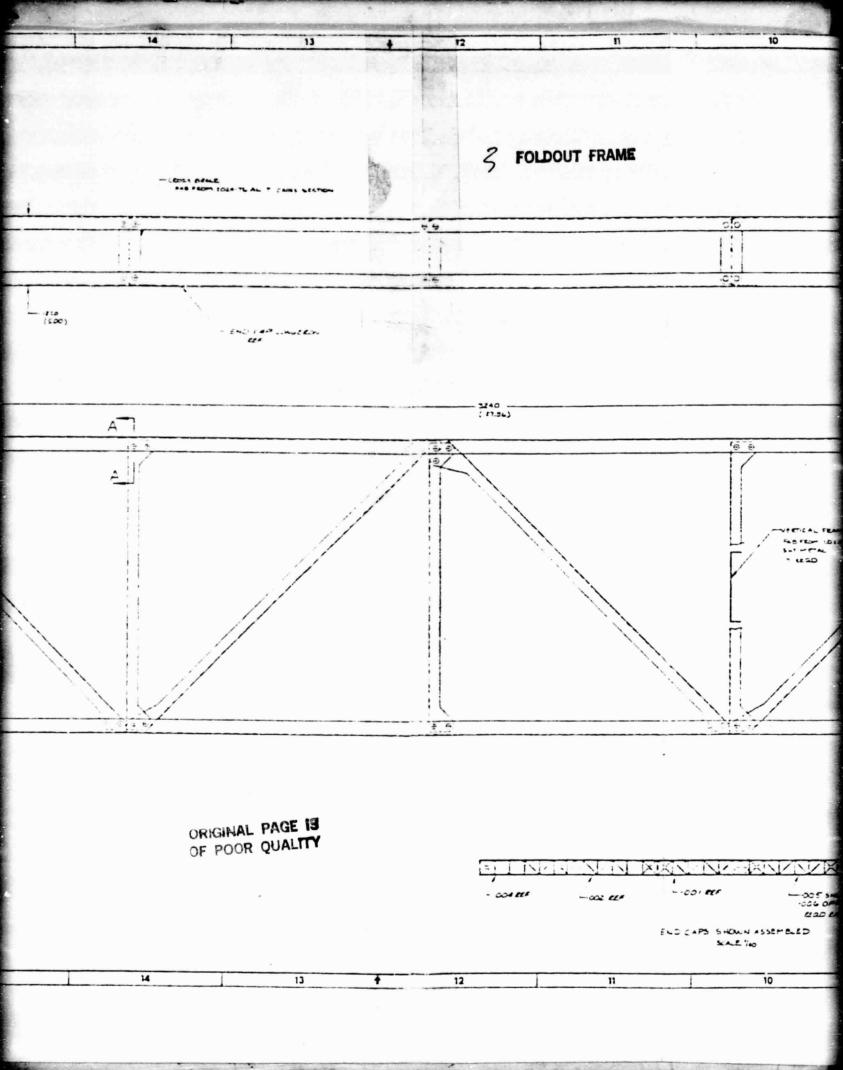




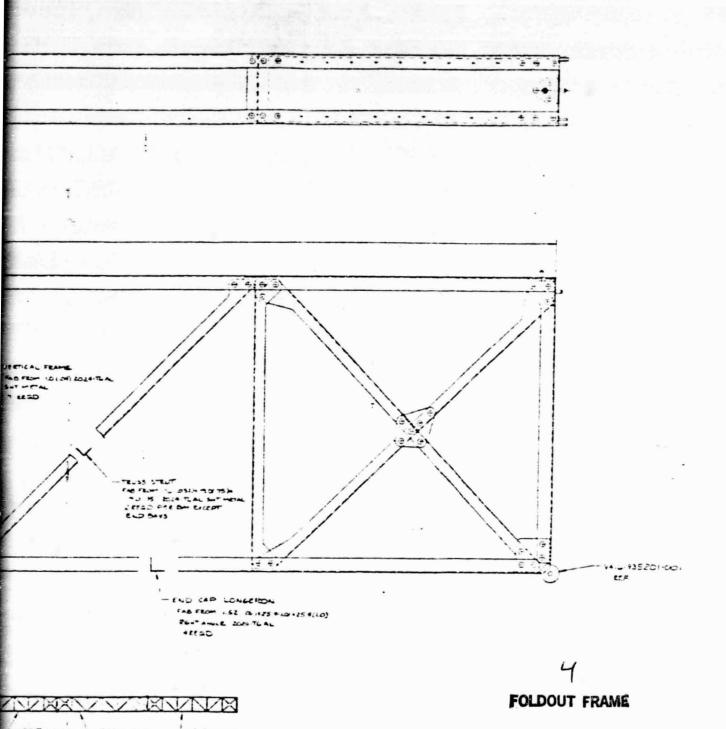


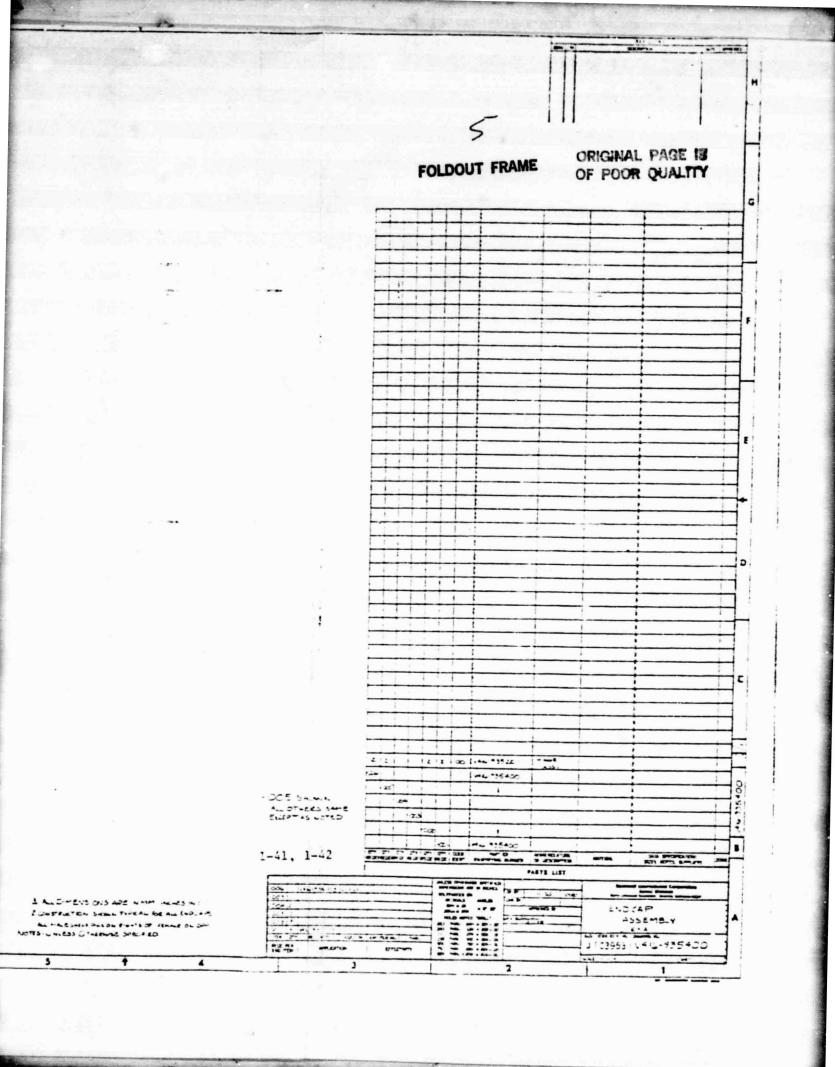


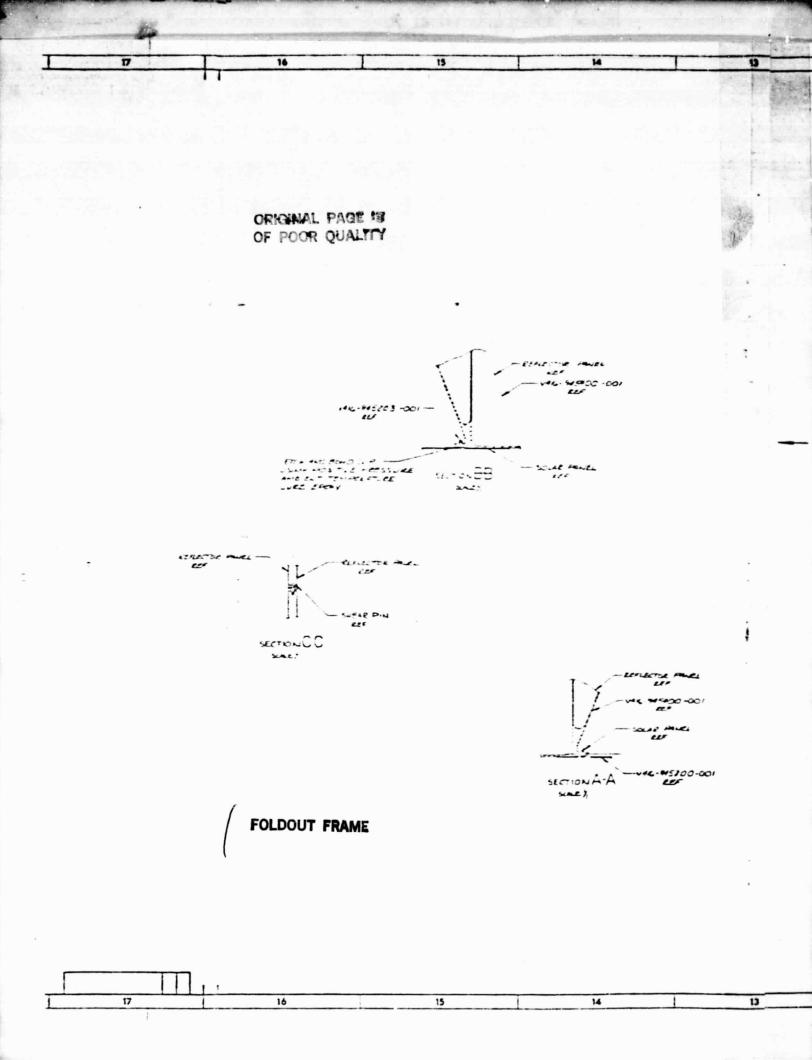


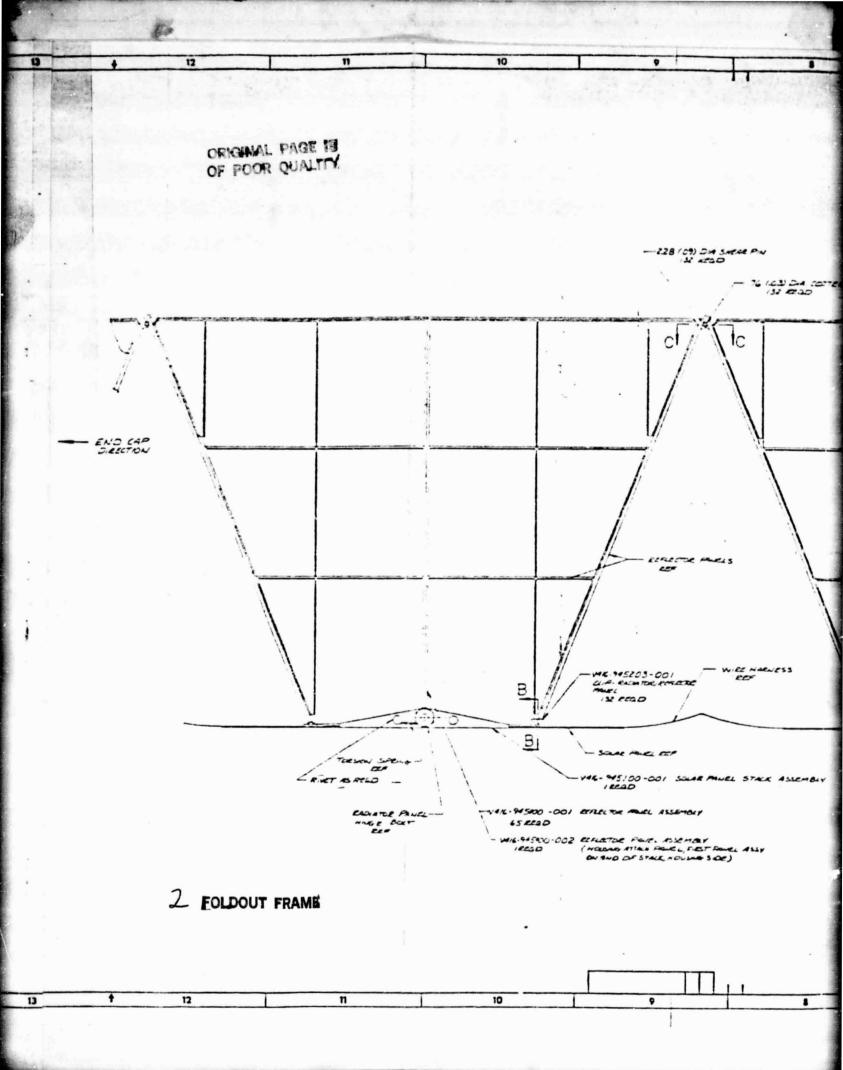


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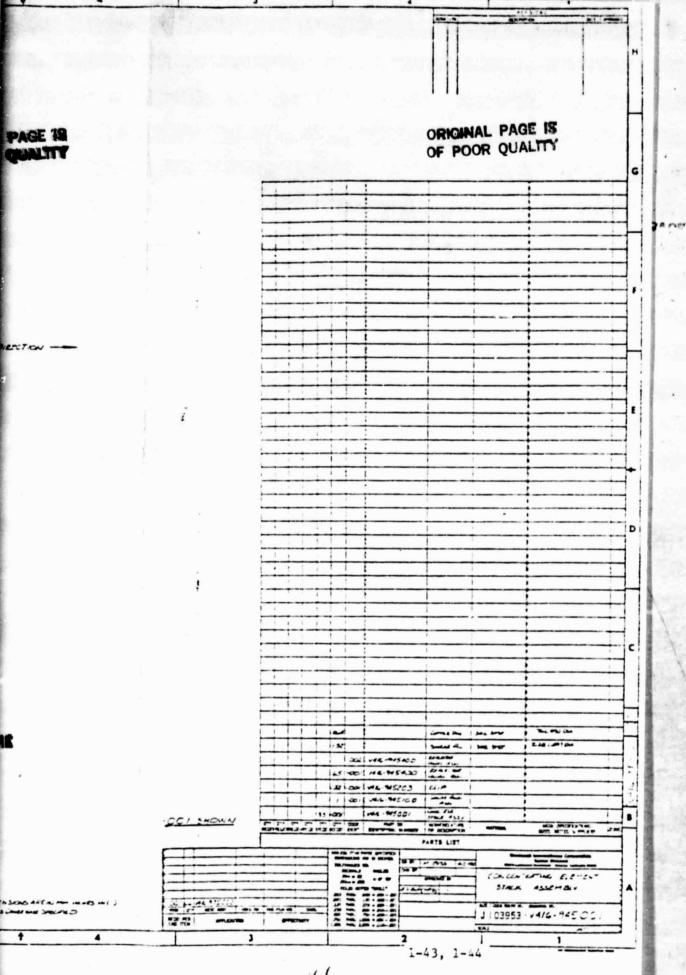




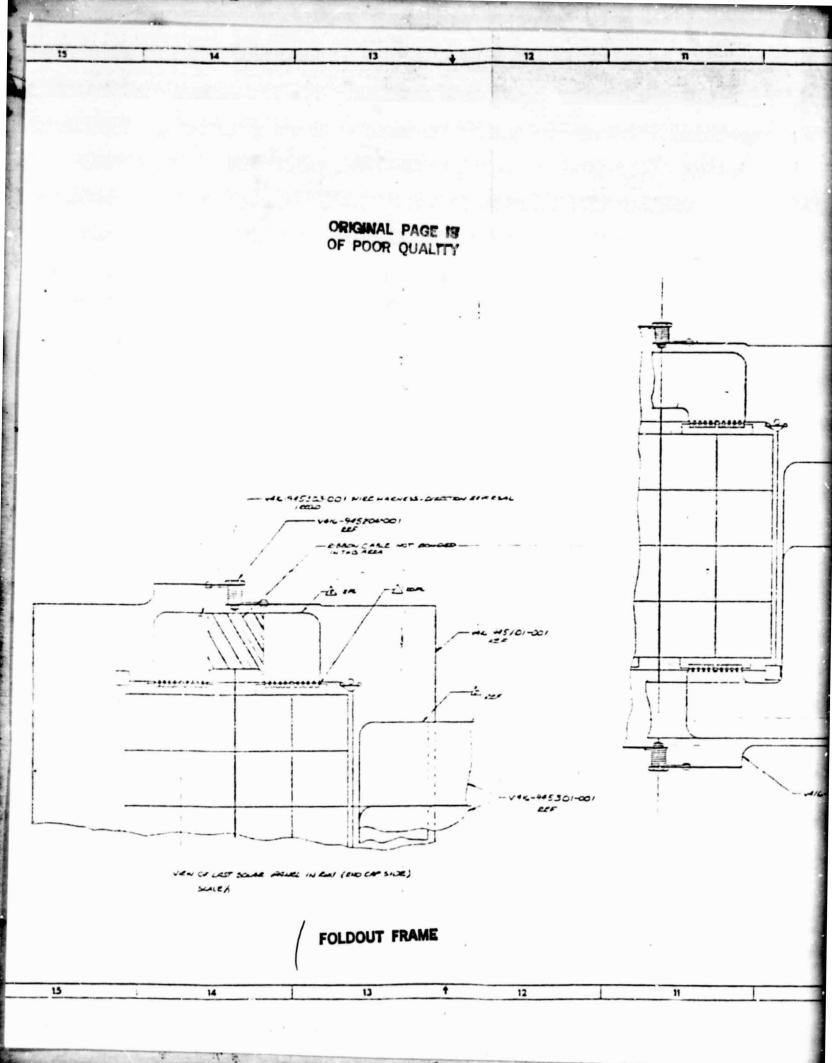


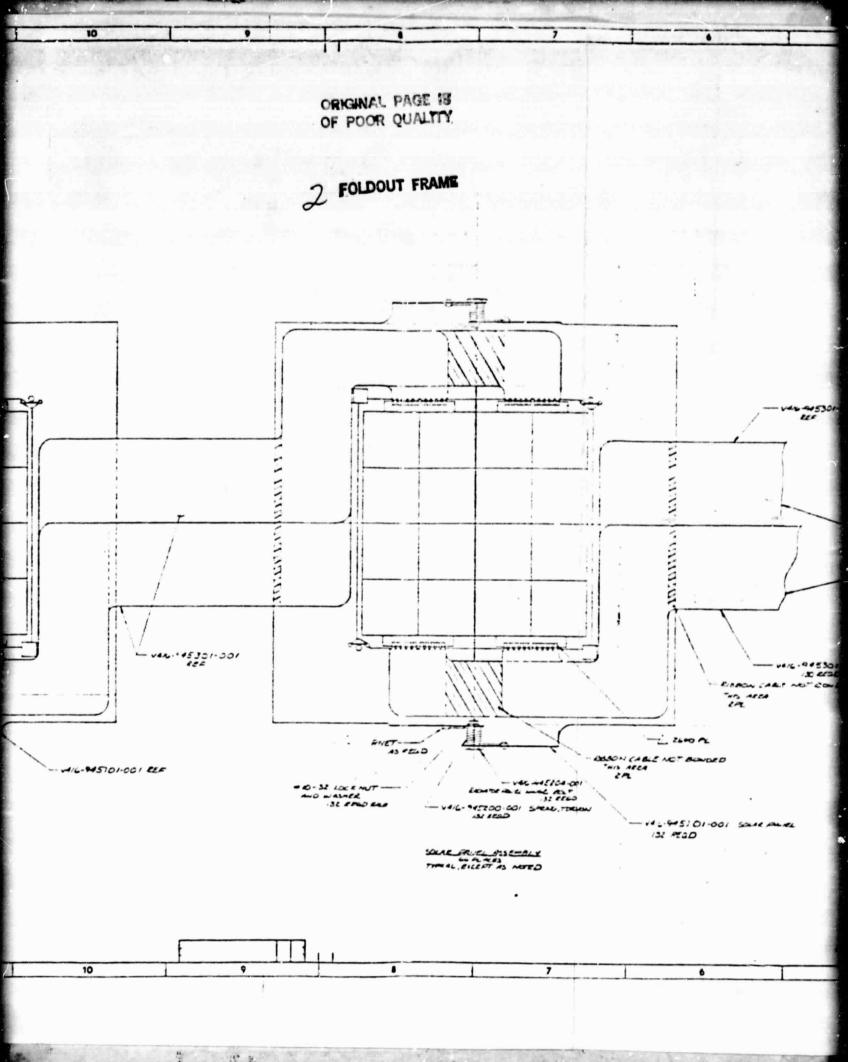


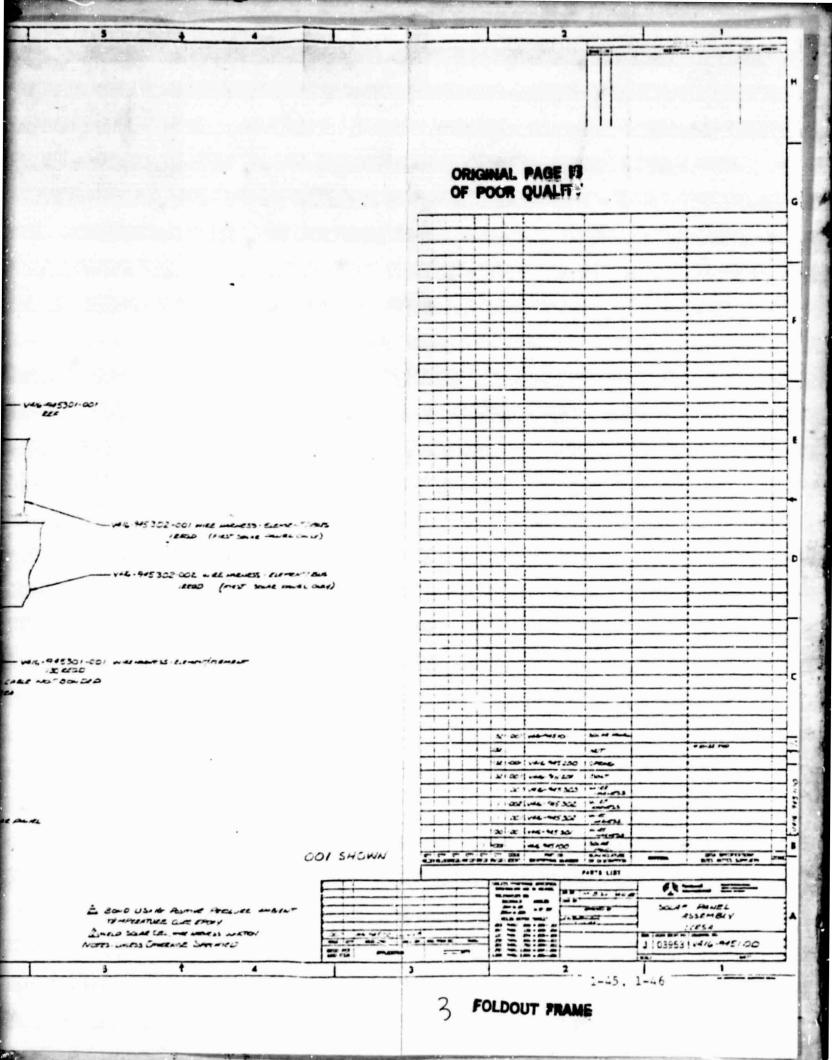
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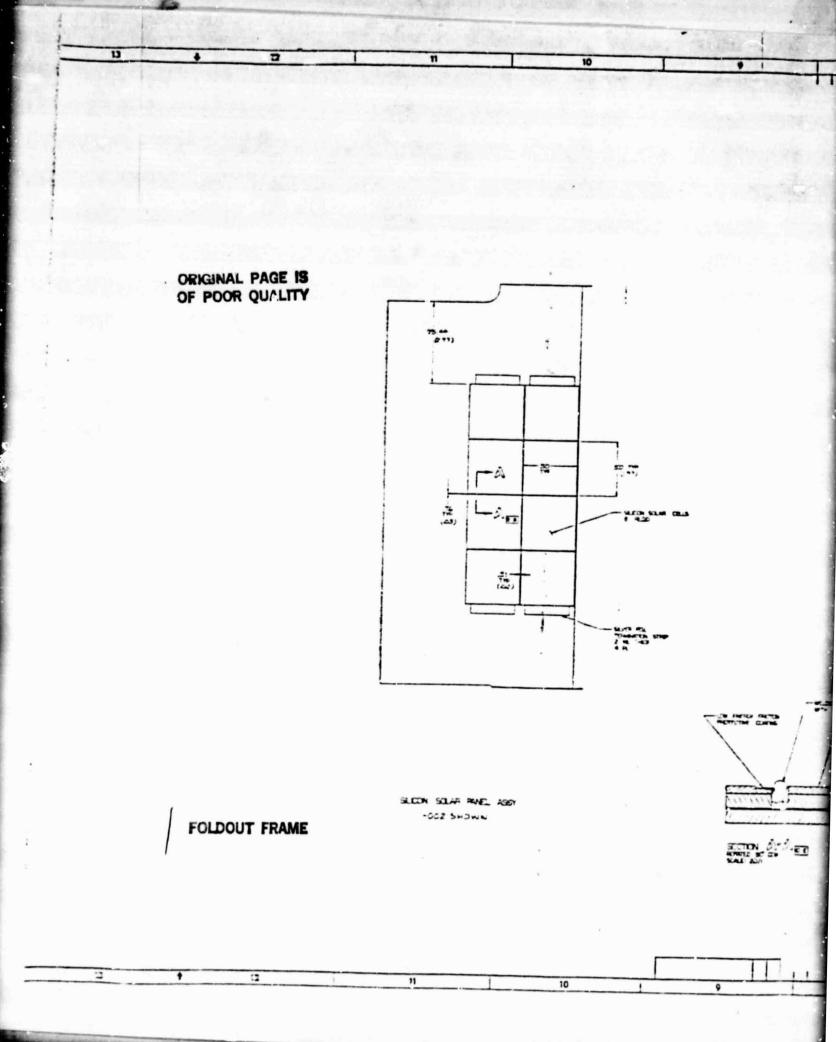


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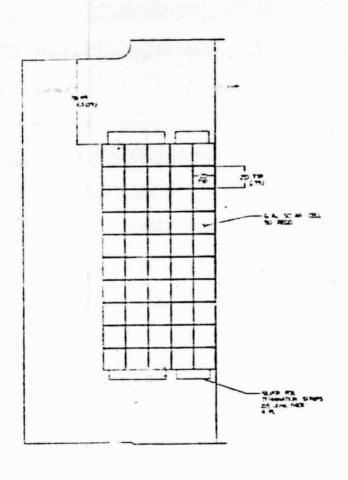


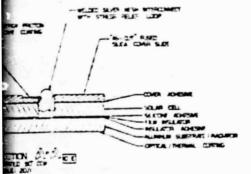






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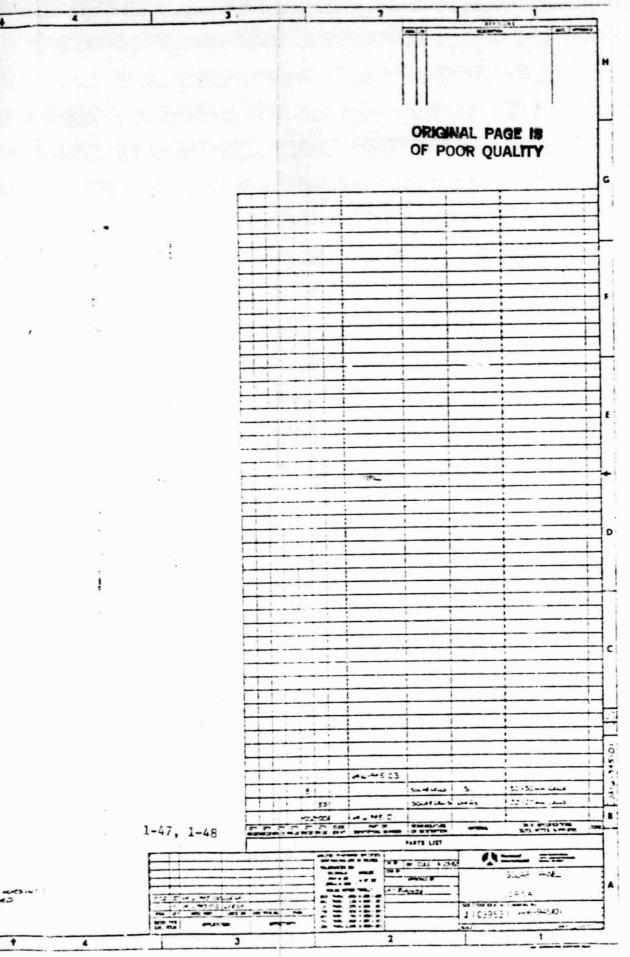
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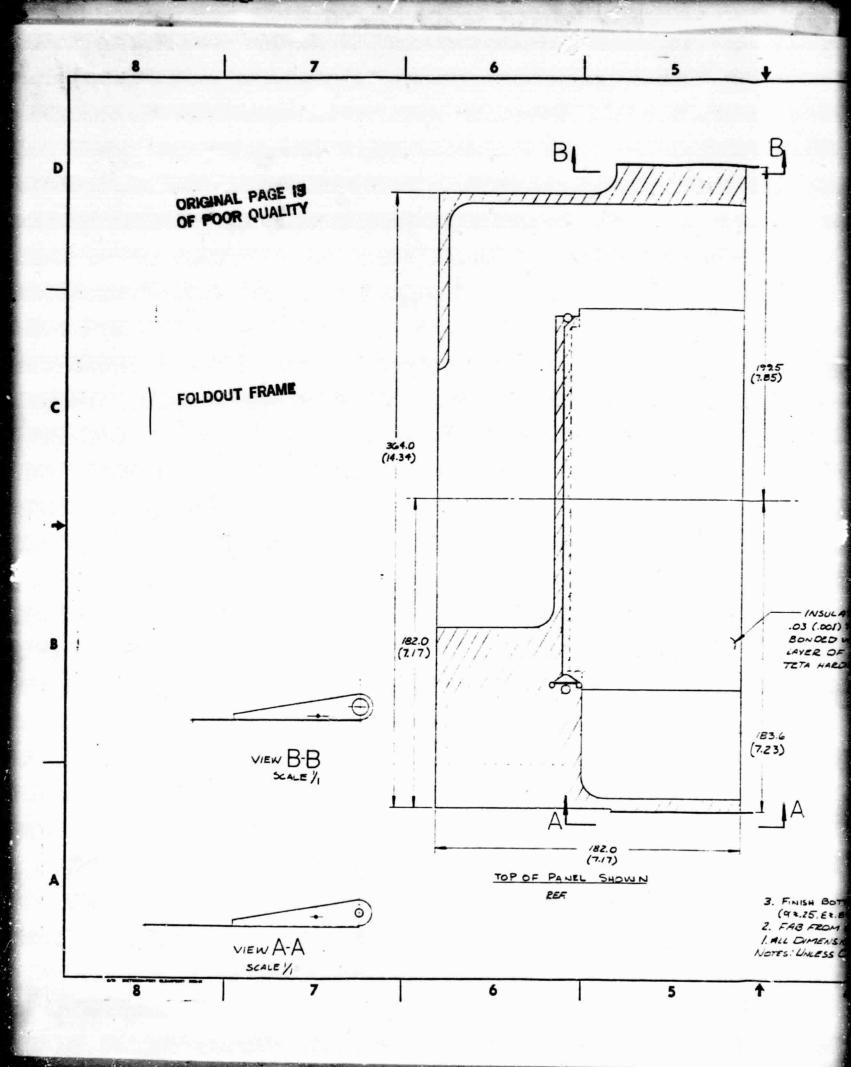
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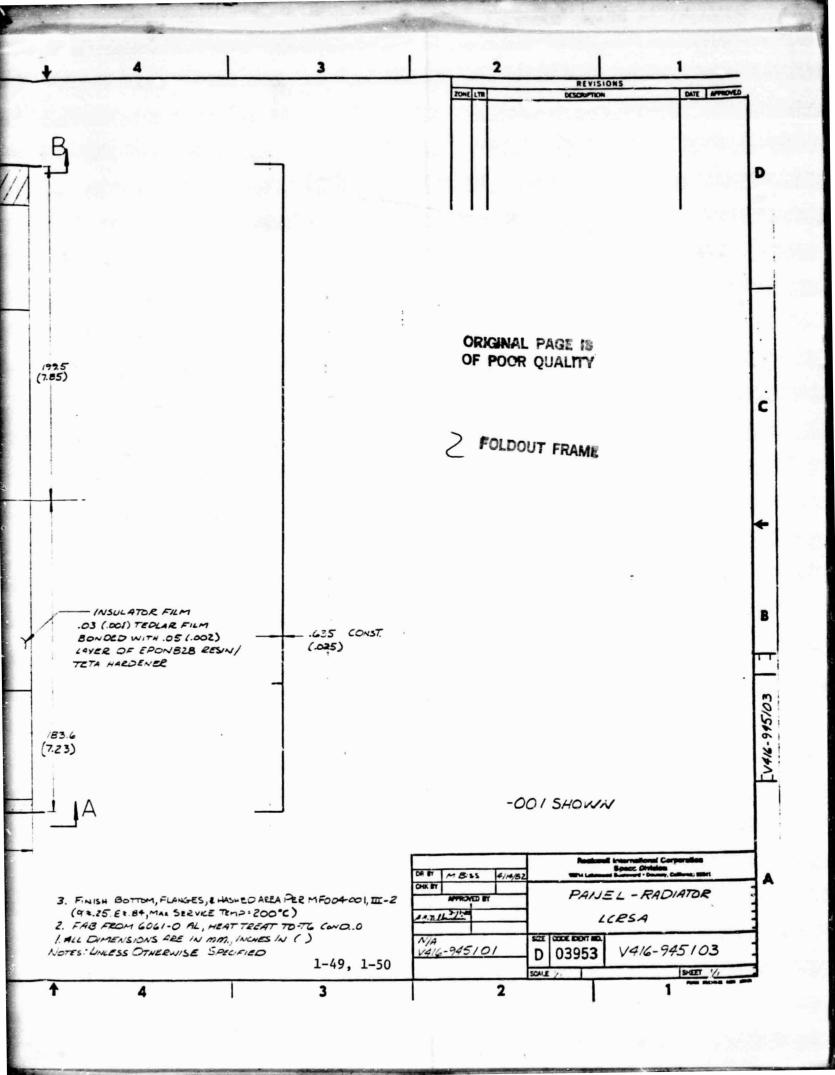
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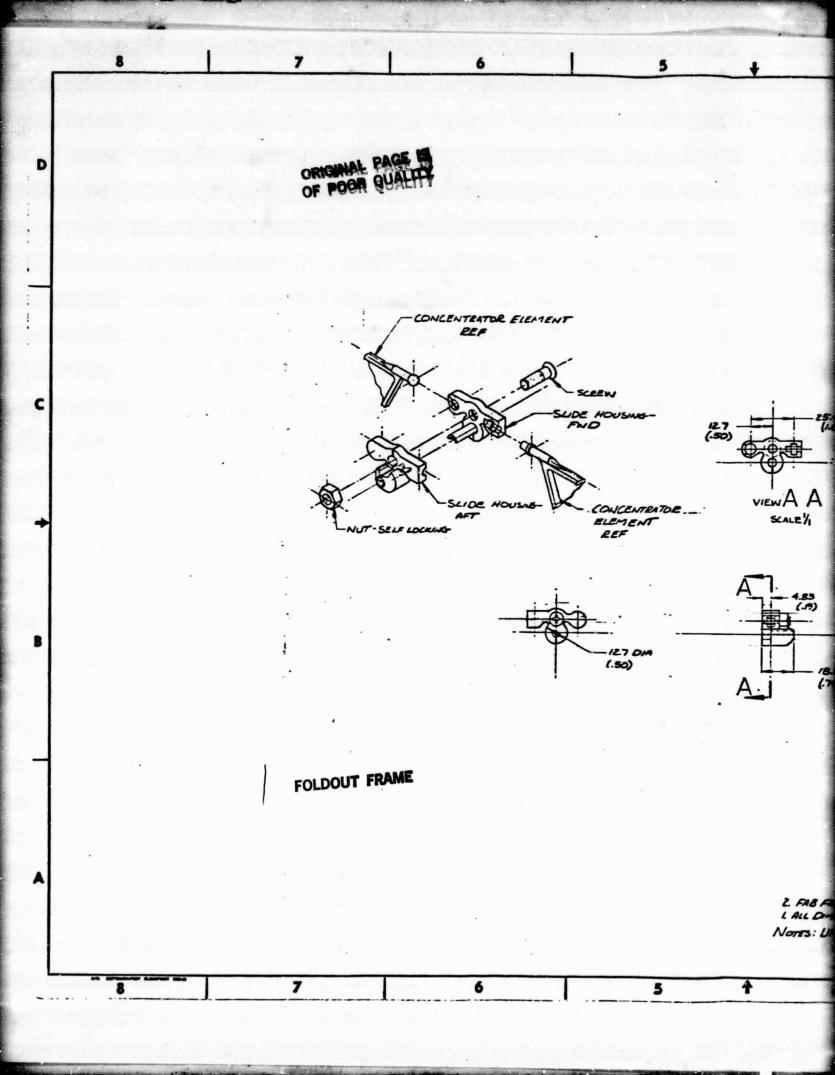
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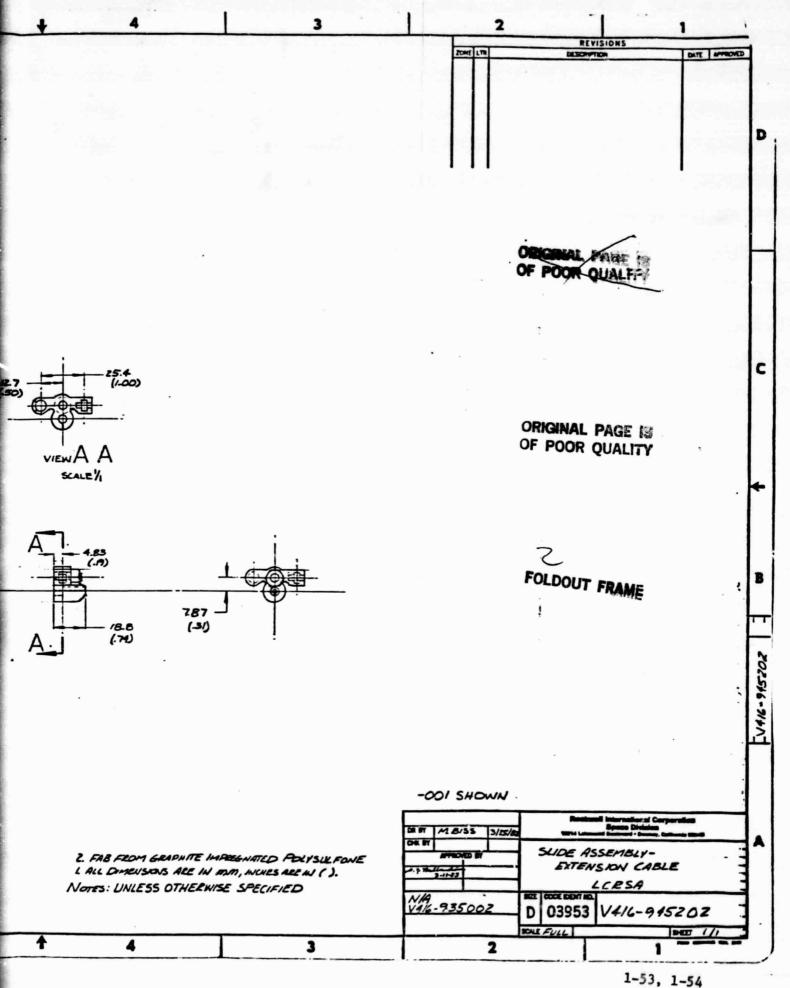


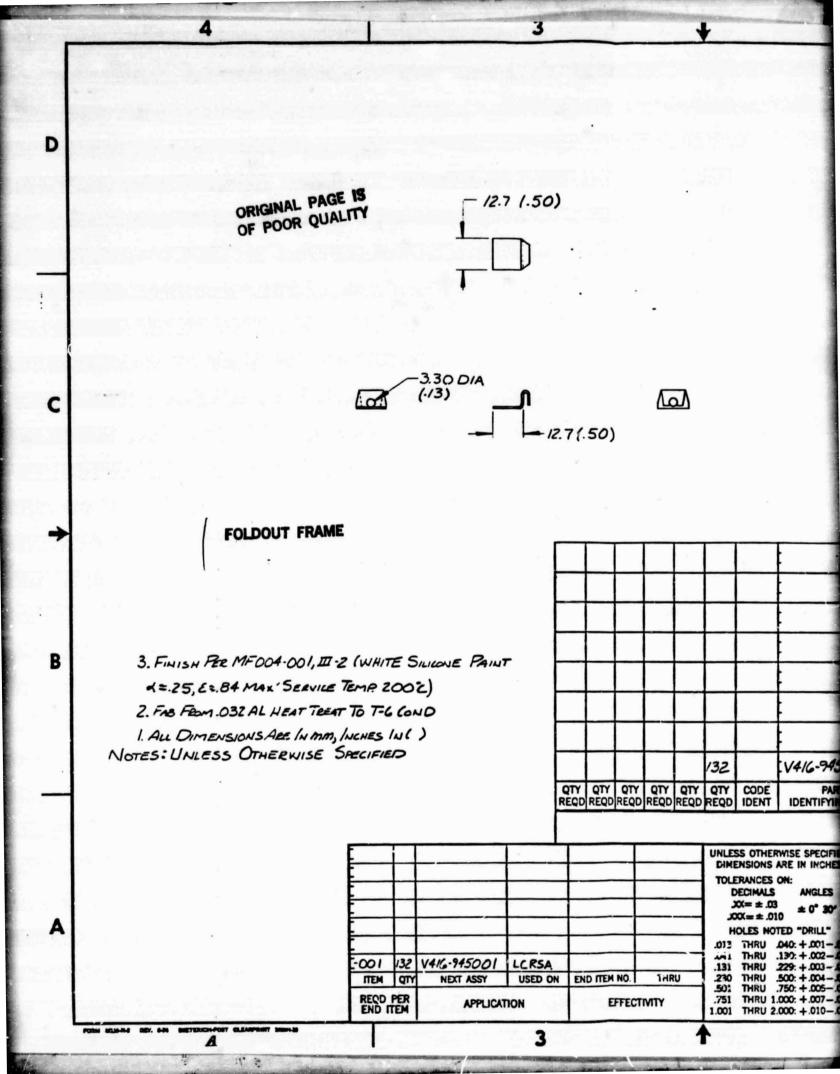


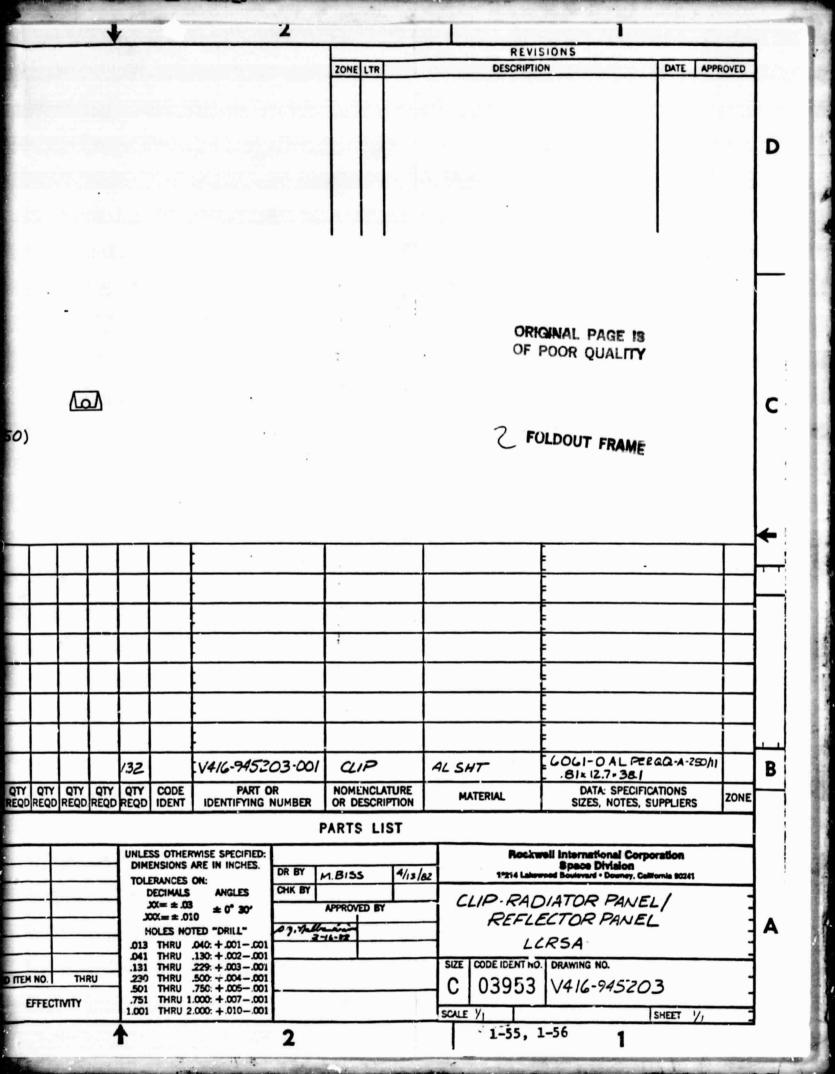
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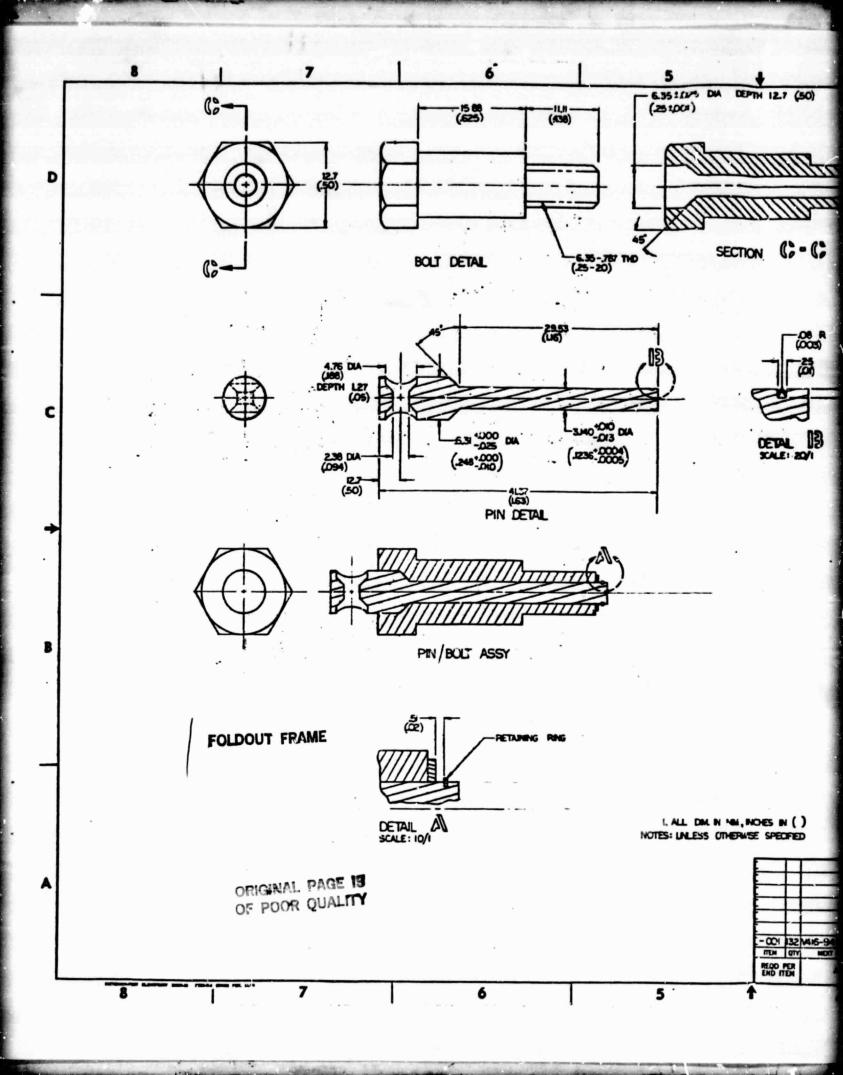
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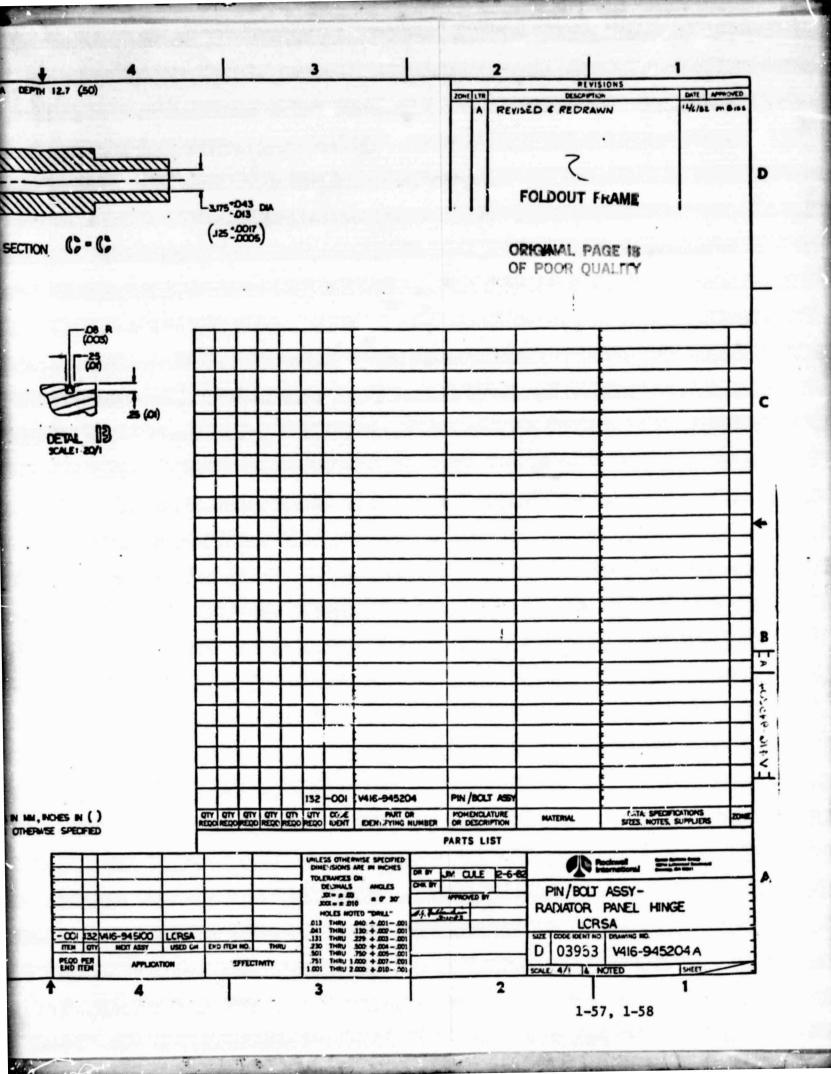


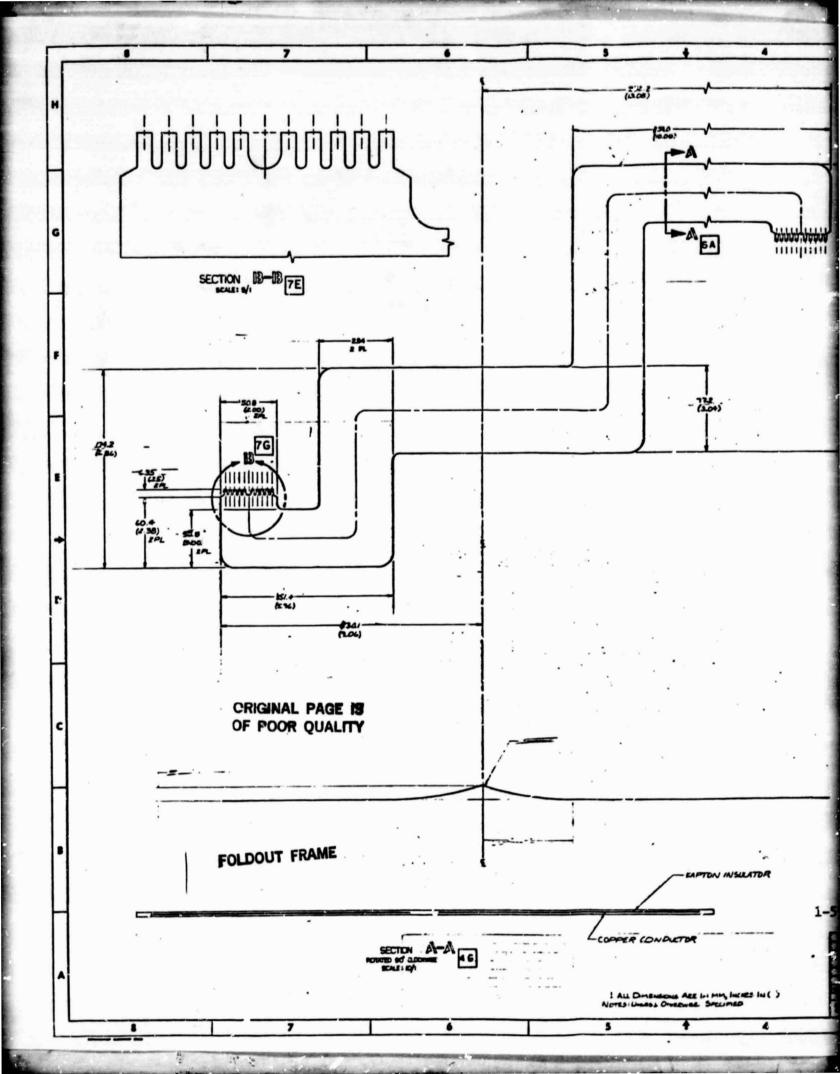


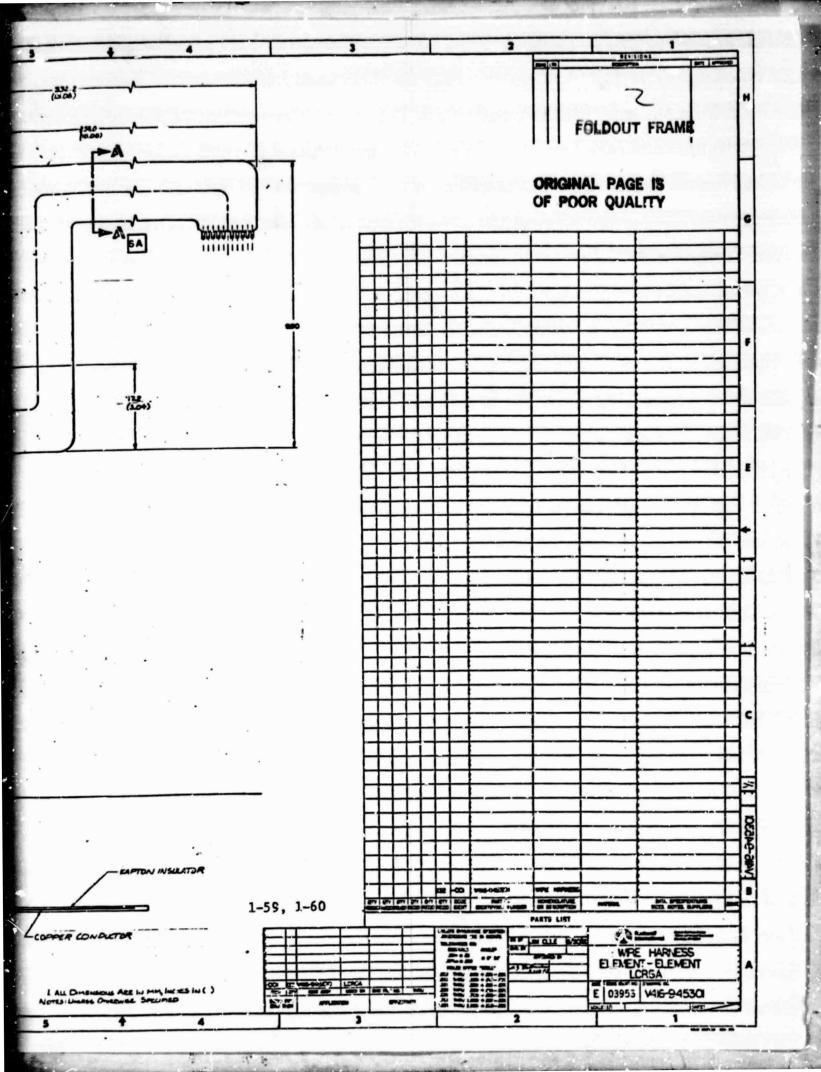


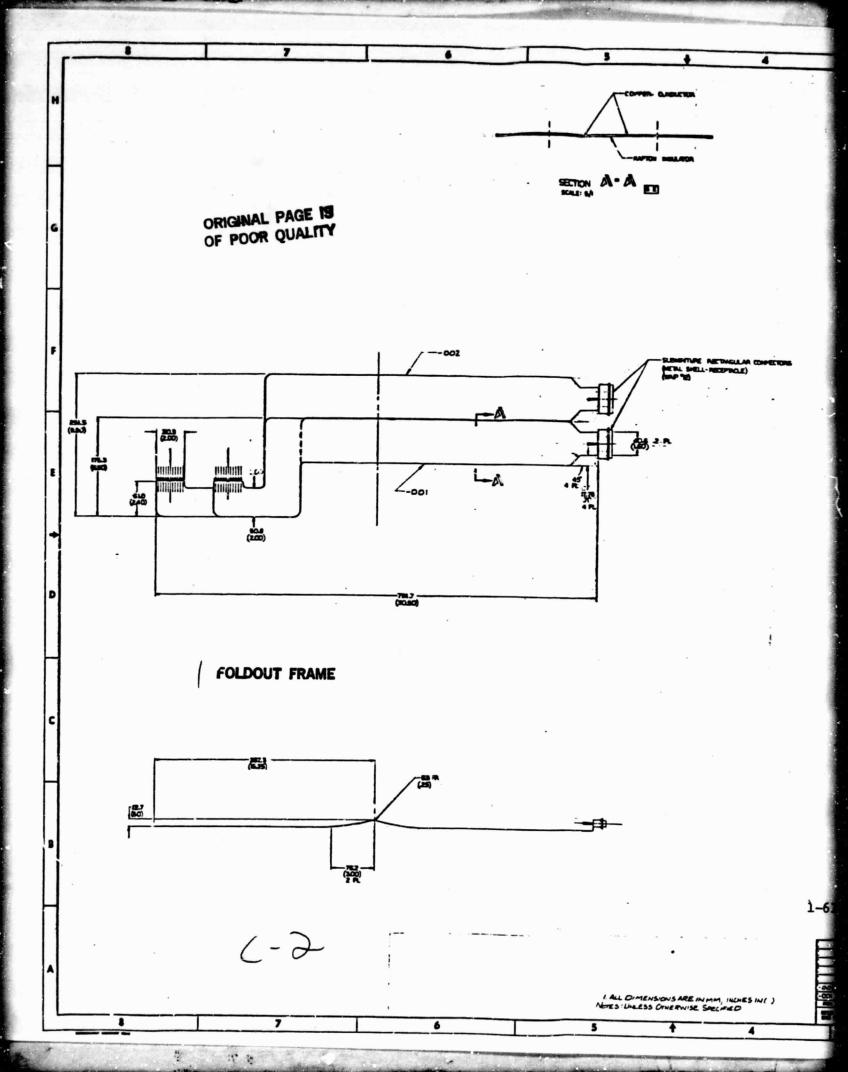


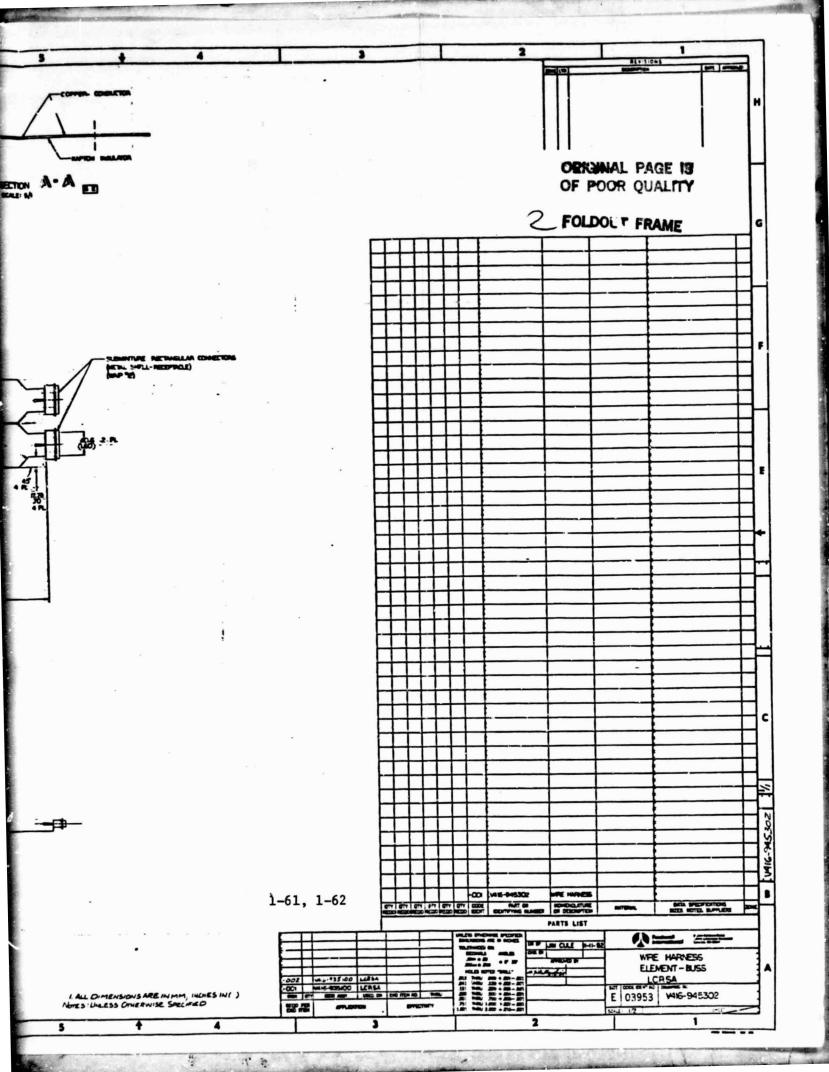


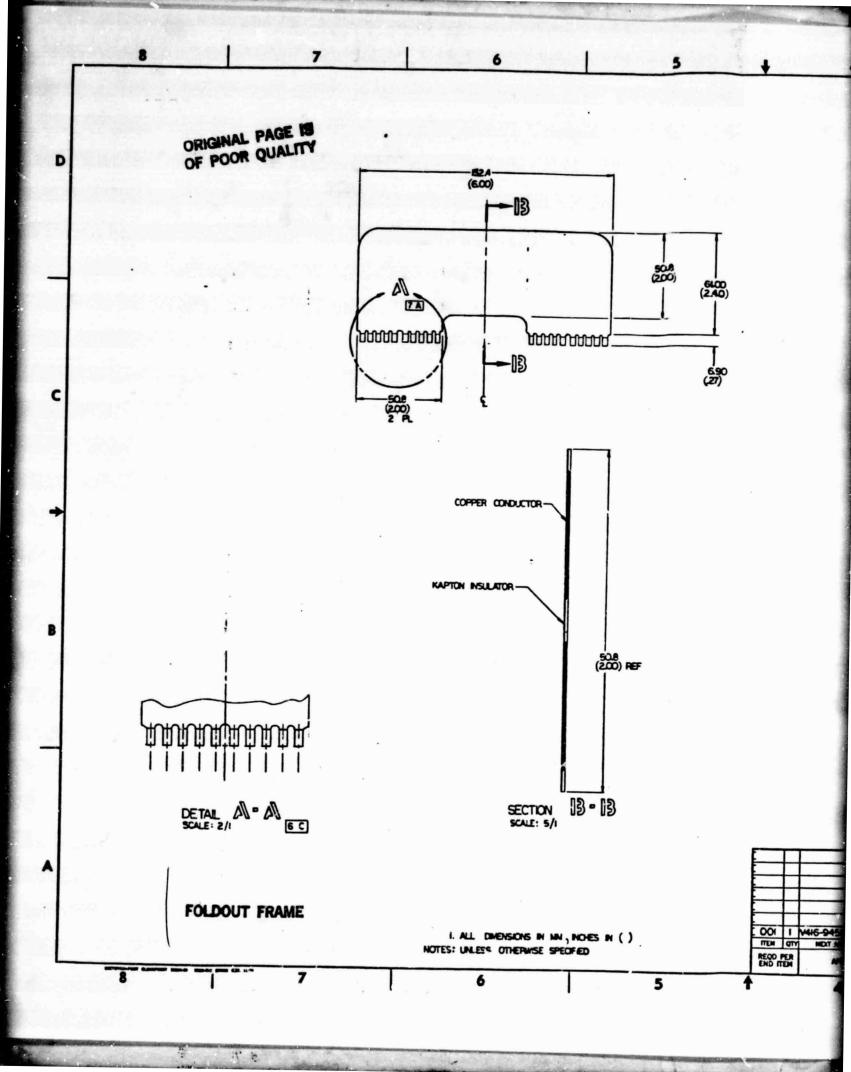


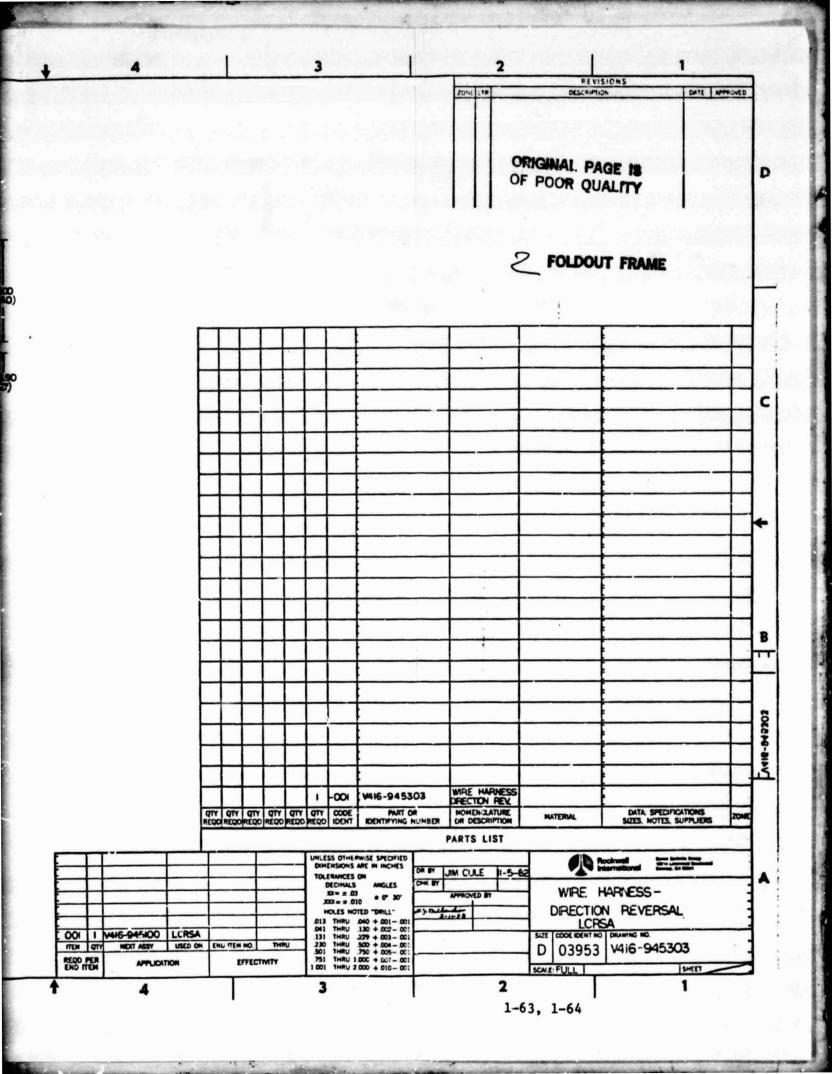


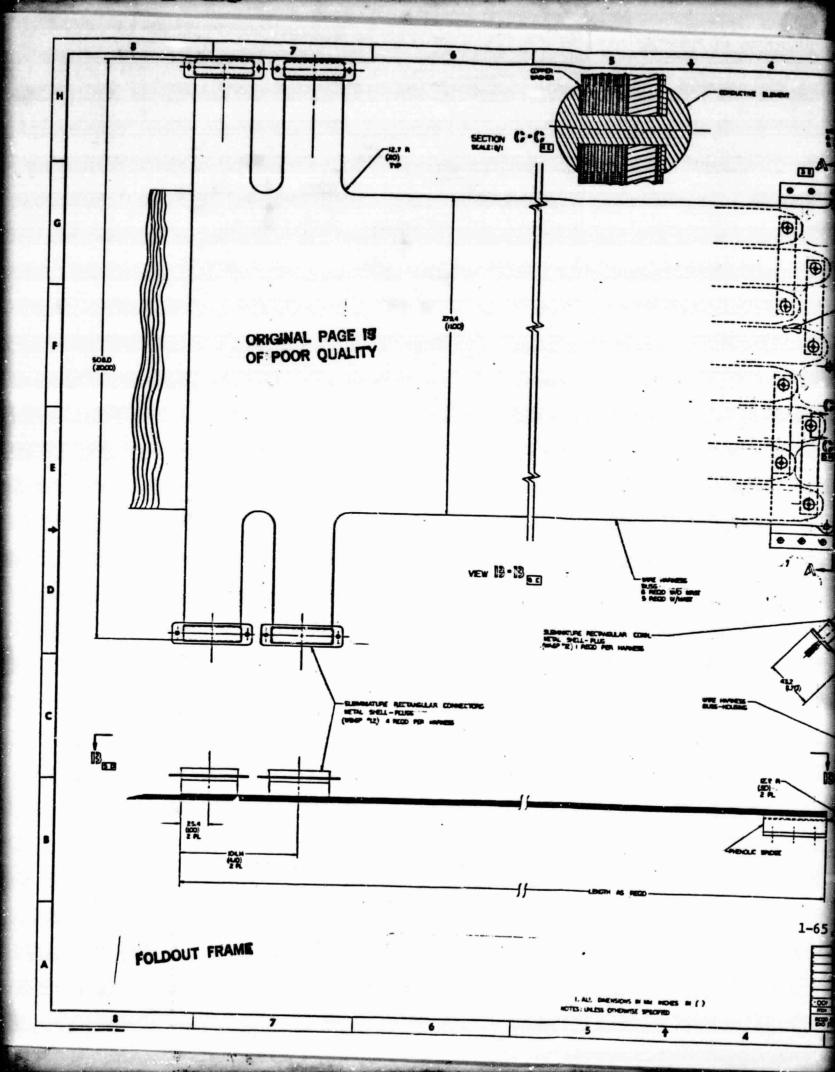


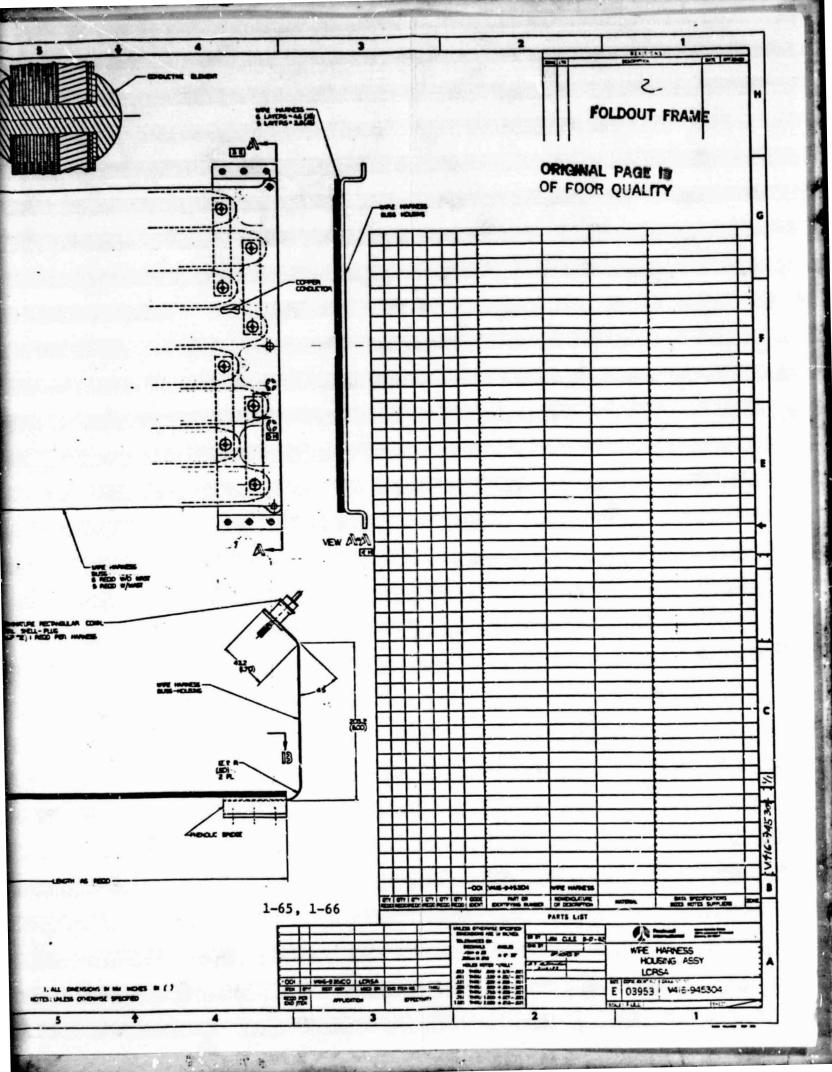


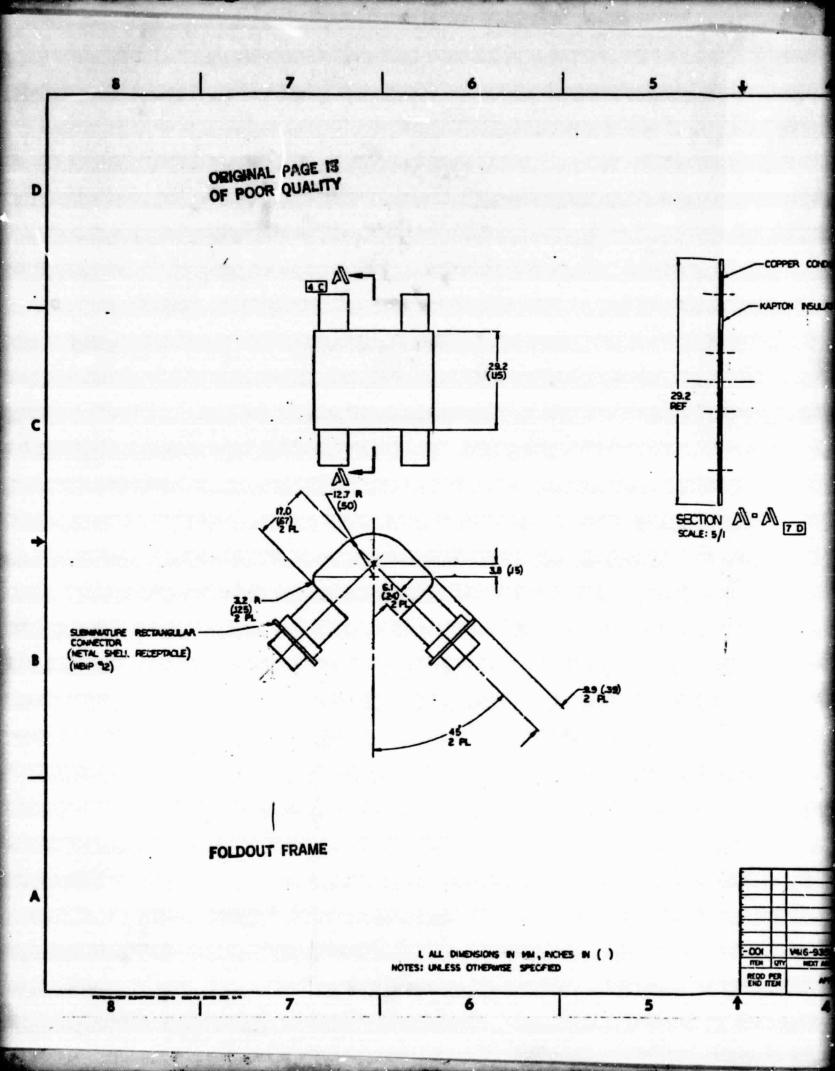


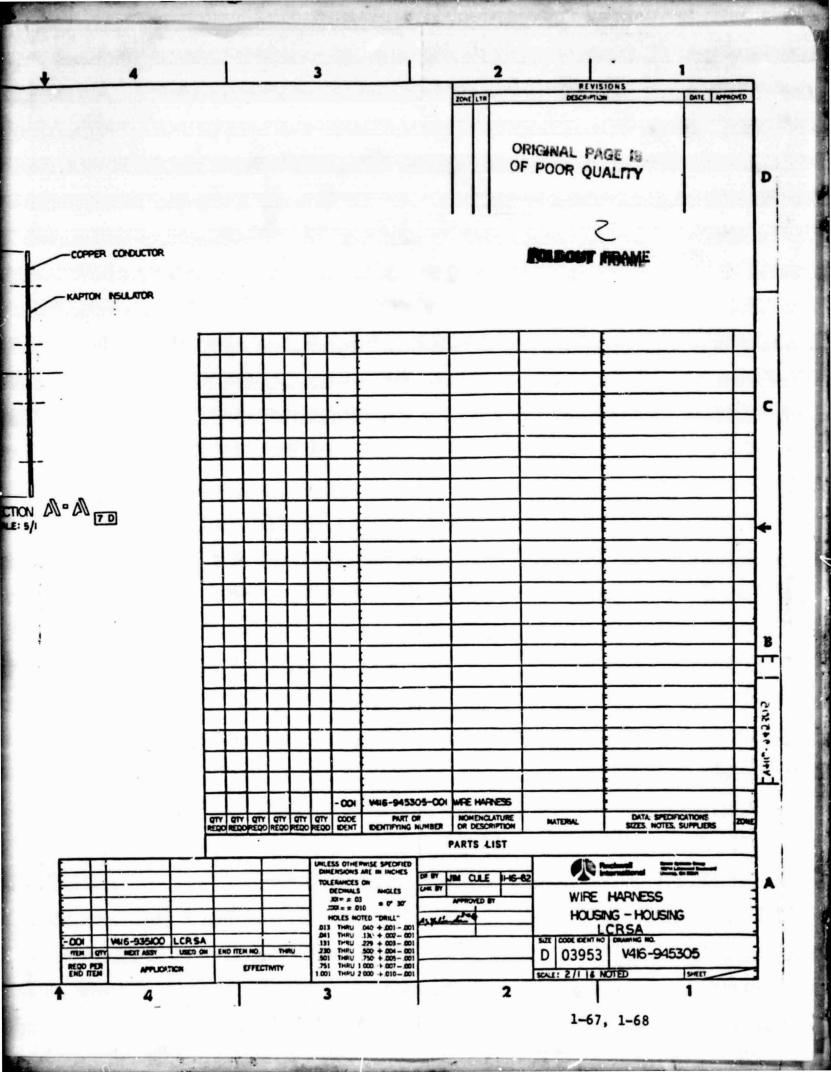


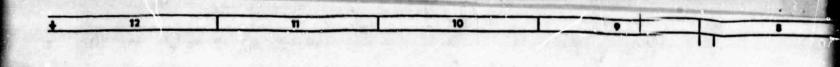




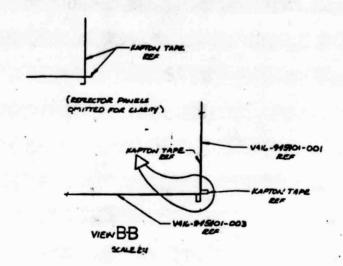




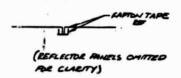


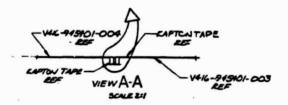


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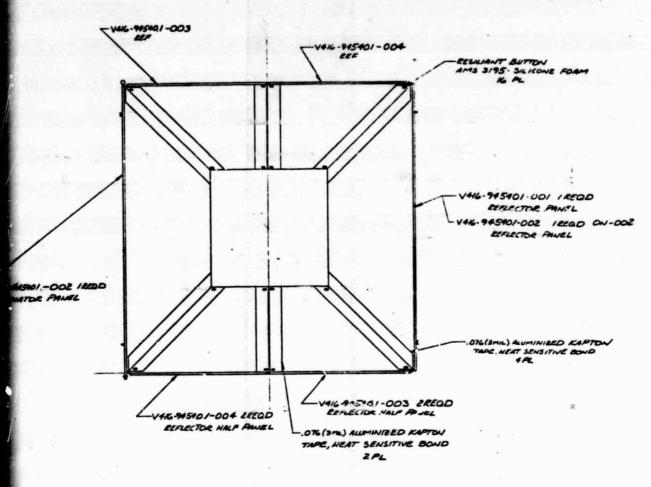


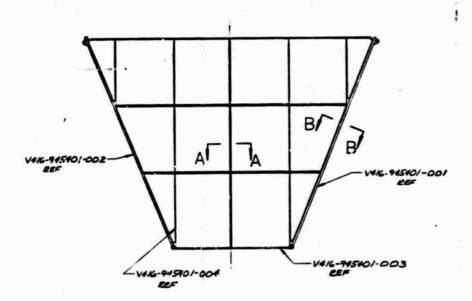
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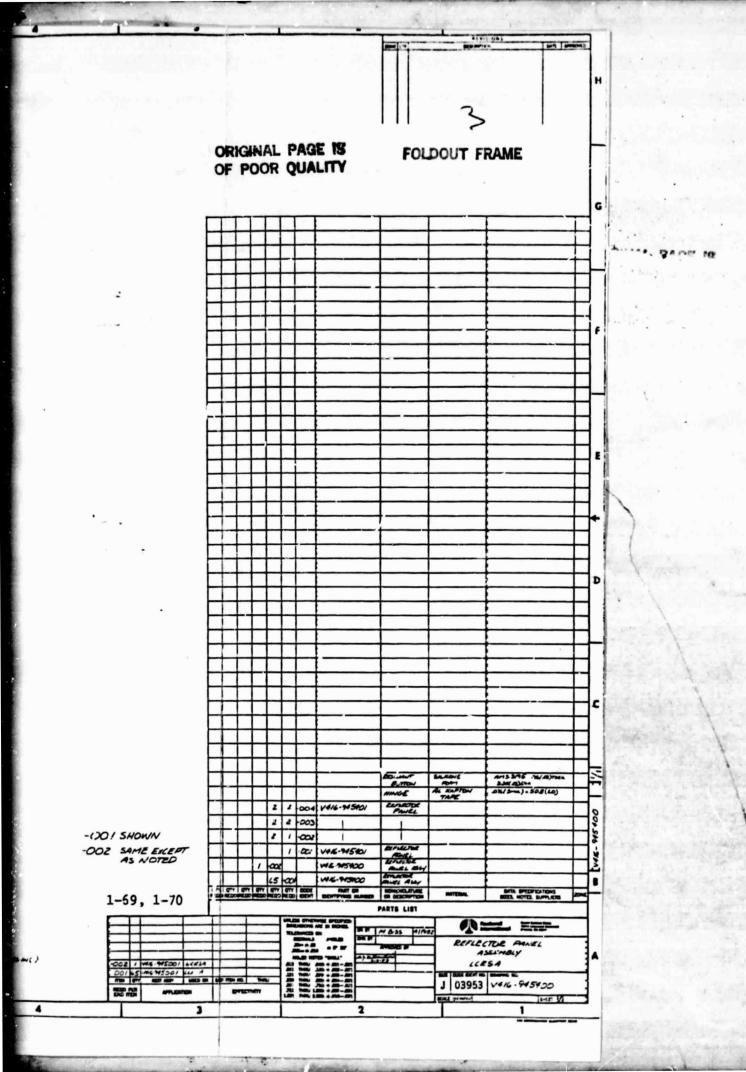
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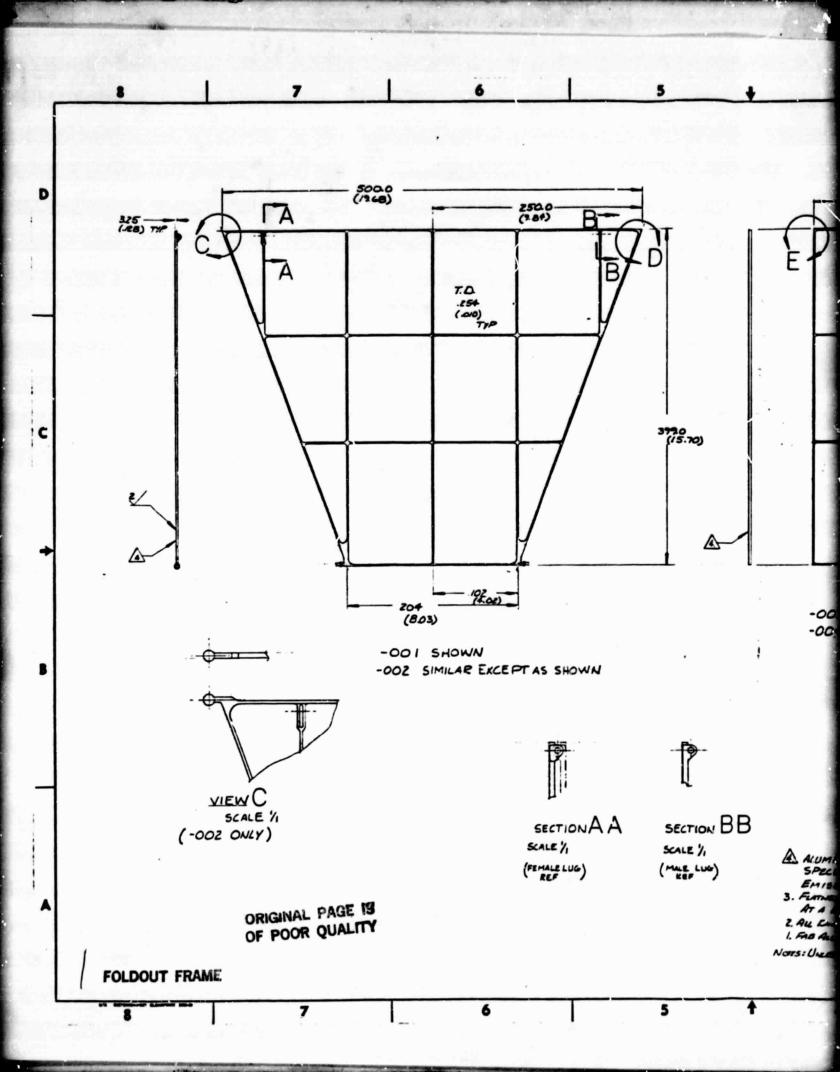
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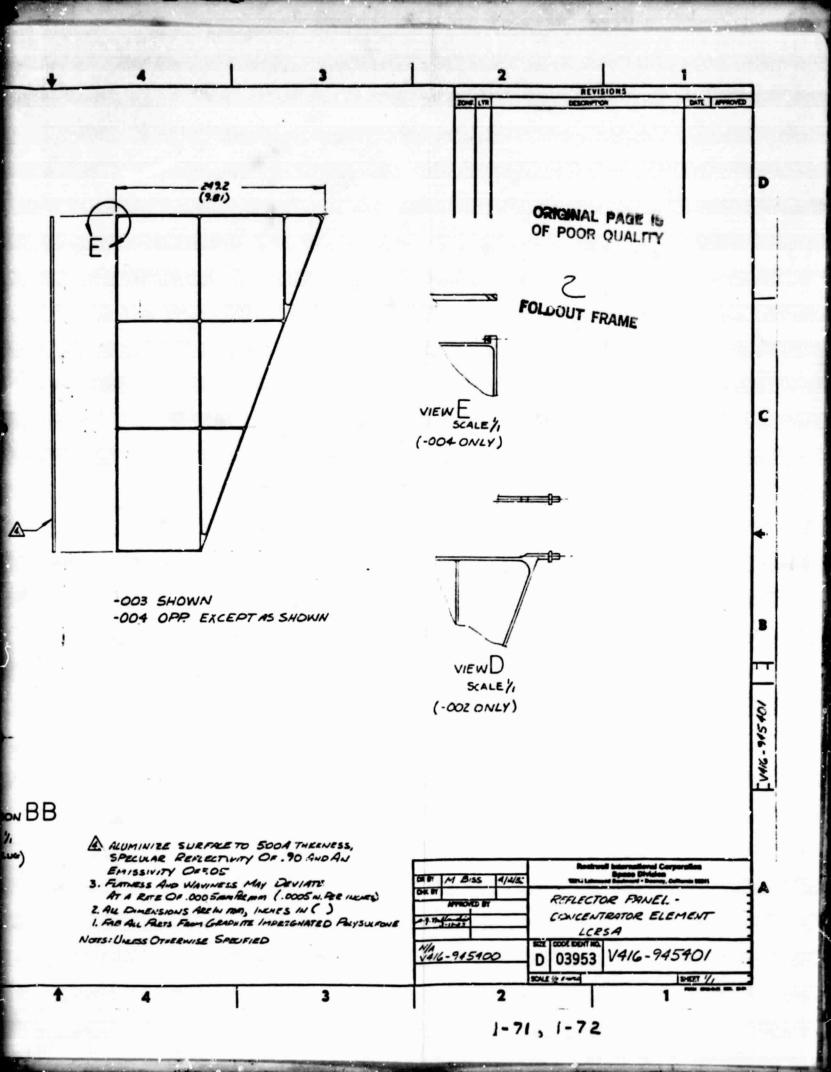
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2.0 TEST HARDWARE DRAWINGS

2.1 DISCUSSION

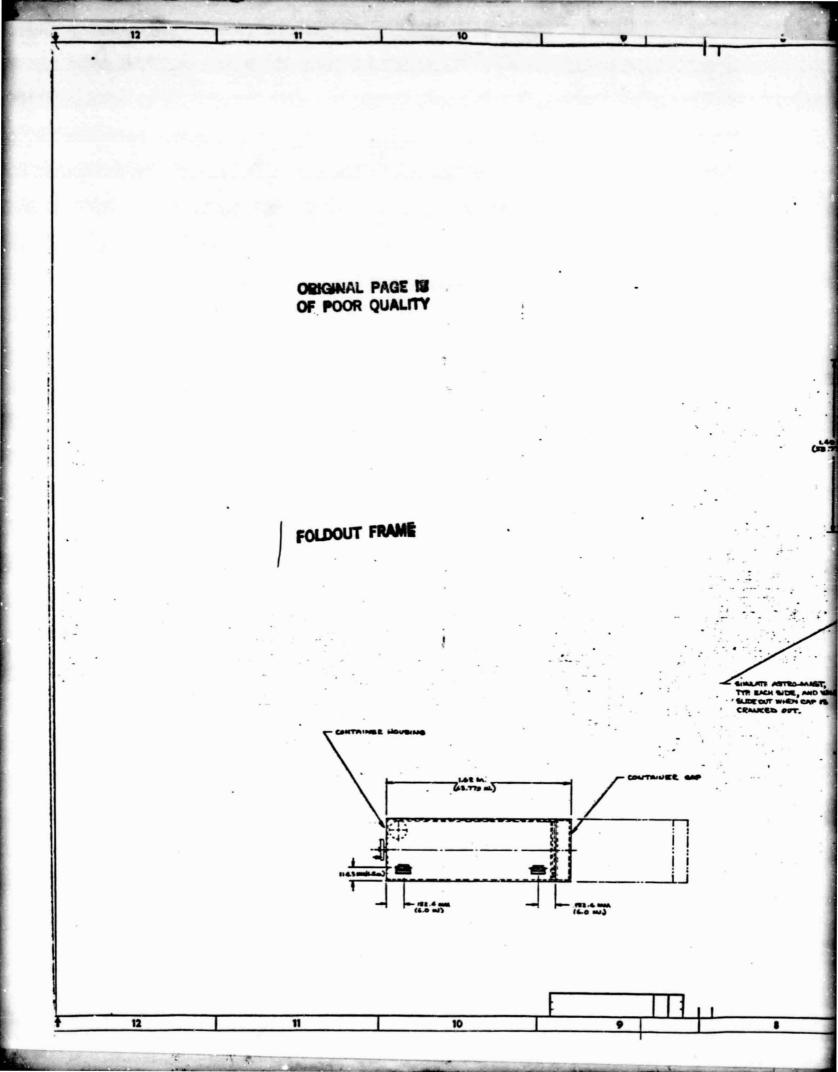
The hardware required to carry out experimental activities under this program are listed in the drawing tree depicted in Figure 2. There are a total of 21 drawings, of several types. Some provide information for subcontractors; others were used to construct Rockwell-fabricated components. The tests and experiments performed using the resulting hardware are fully described in Vol. 1 of this report.

2.2 TEST HARDWARE DRAWINGS

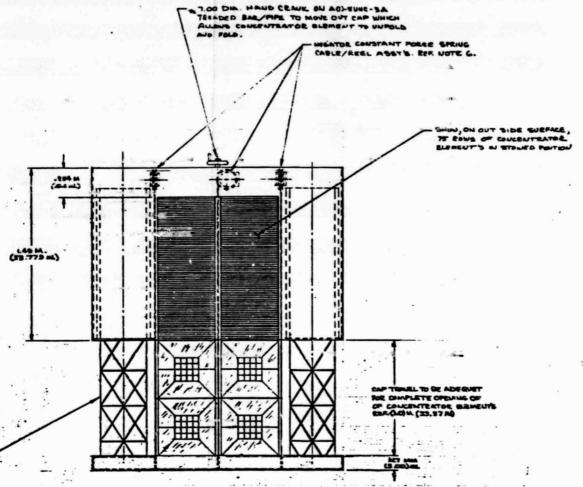
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- 5. D416-451000
- 6. D416-451001
- 7. D416-451092
- 8. D416-451003
- 9. D416-451004
- 10. D416-451005
- 11. D416-451006
- 12. D416-451007
- 13. D416-452000
- 14. D416-452001
- 15. D416-453000
- 16. D416-454000
- 17. D416-454001
- 18. D416-454002
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- 21. D416-455000

Test Hardware Drawing Tree

Figure 2.

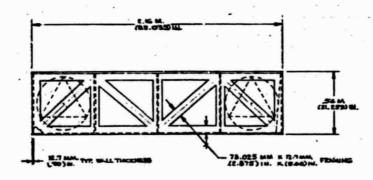


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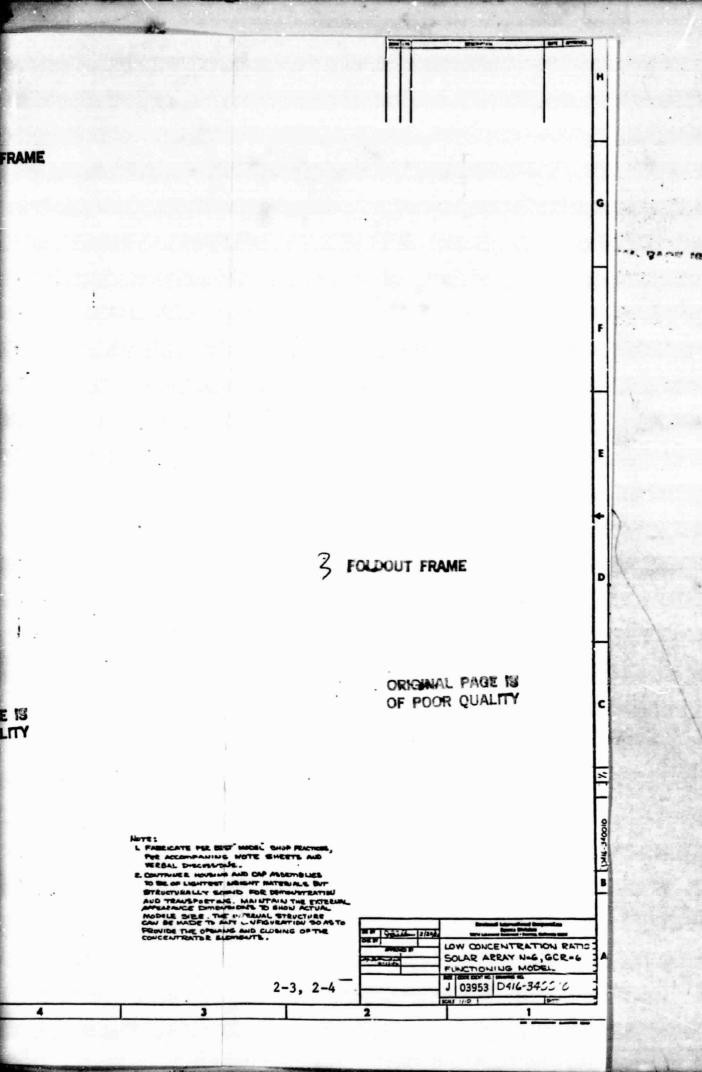
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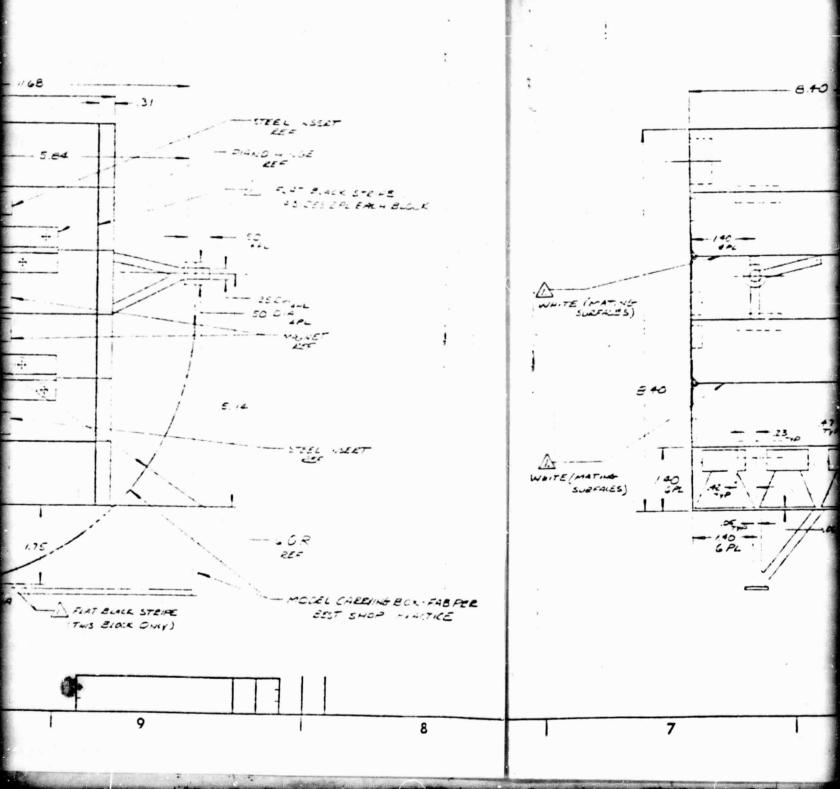
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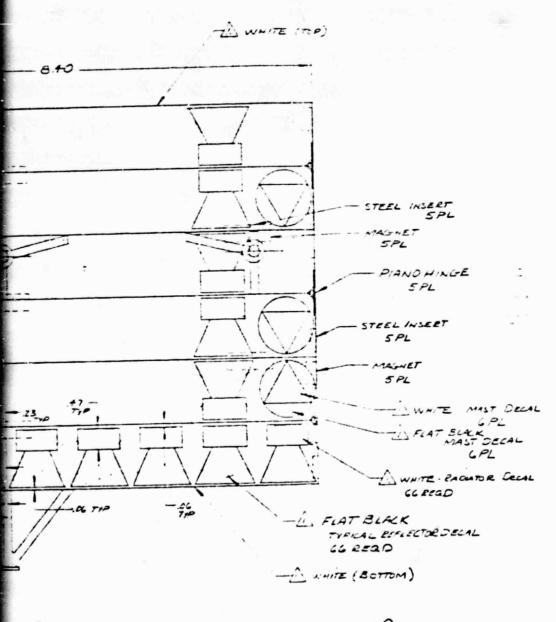
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- 5. REMOVEABLE SURE 4. MATERIALS - METAL
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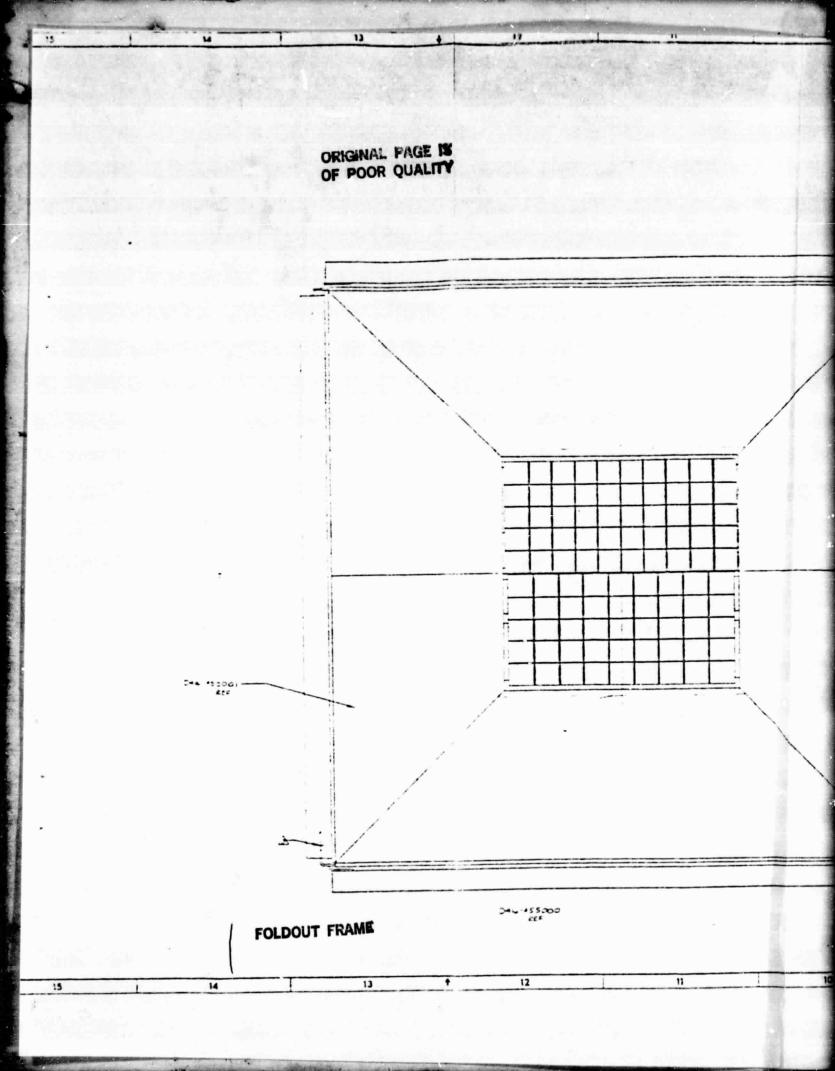
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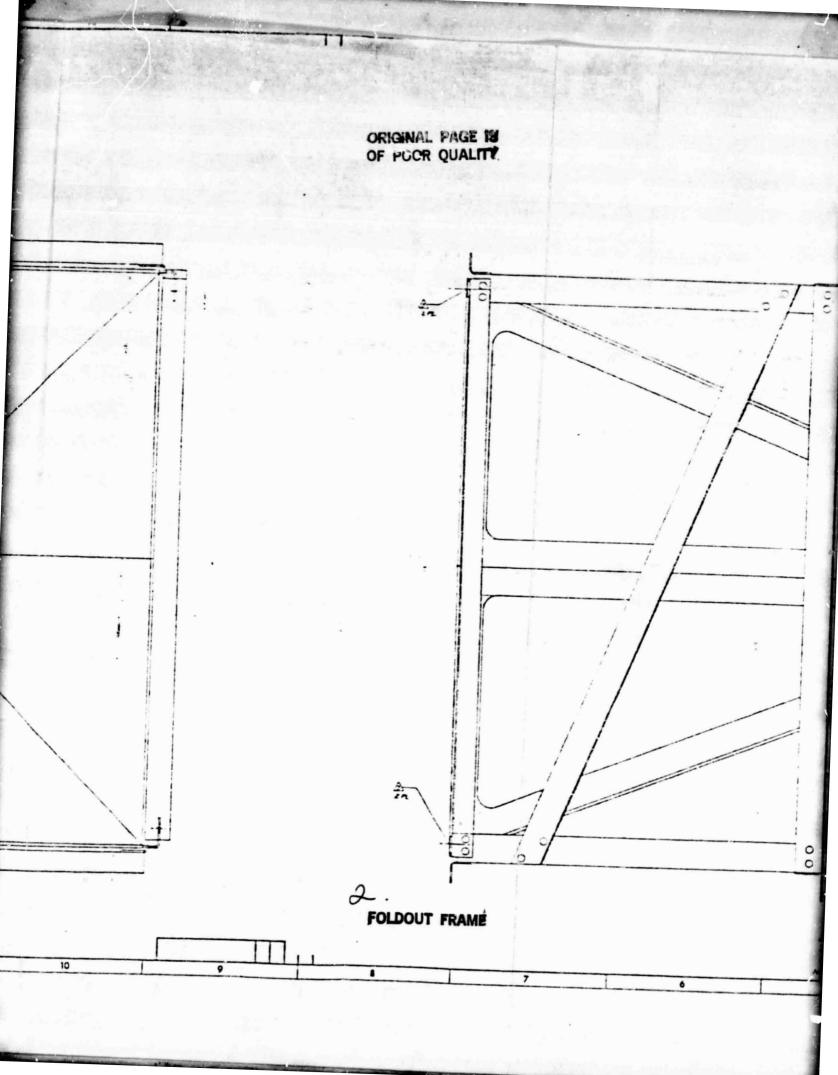
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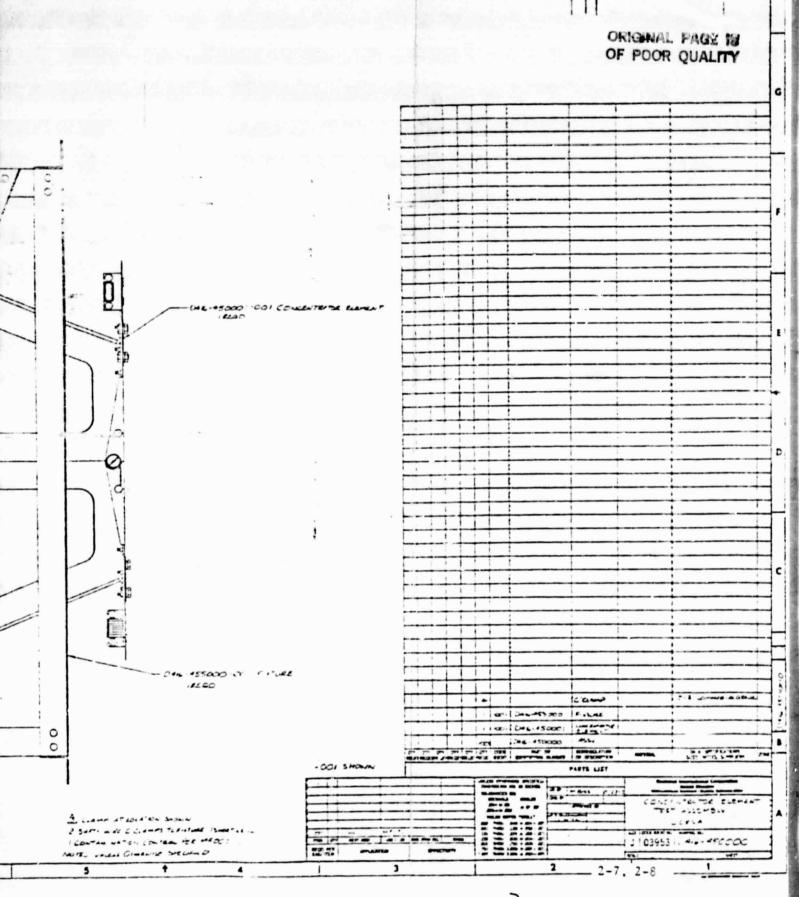
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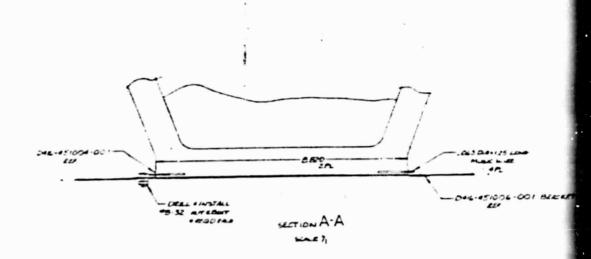
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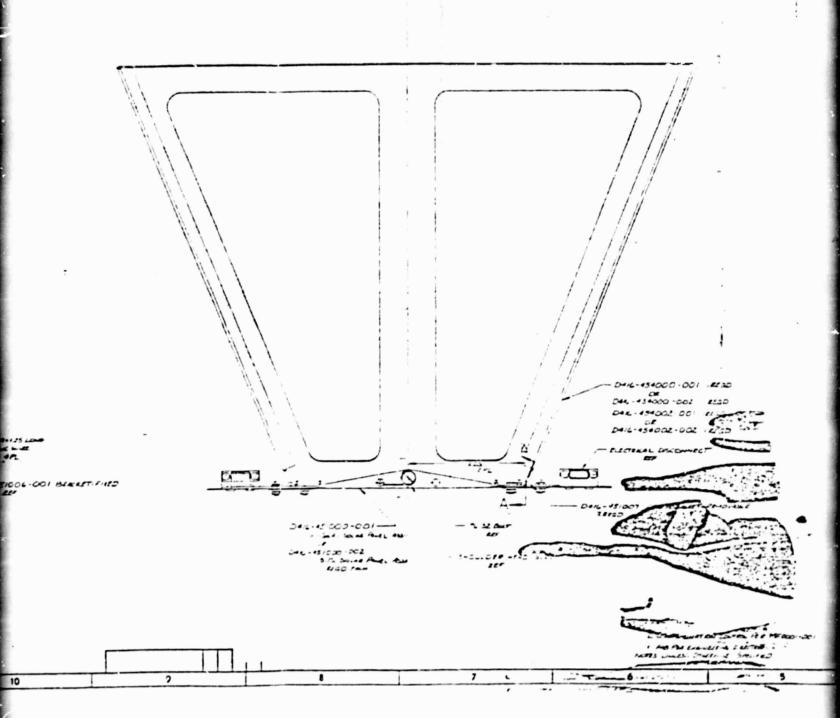
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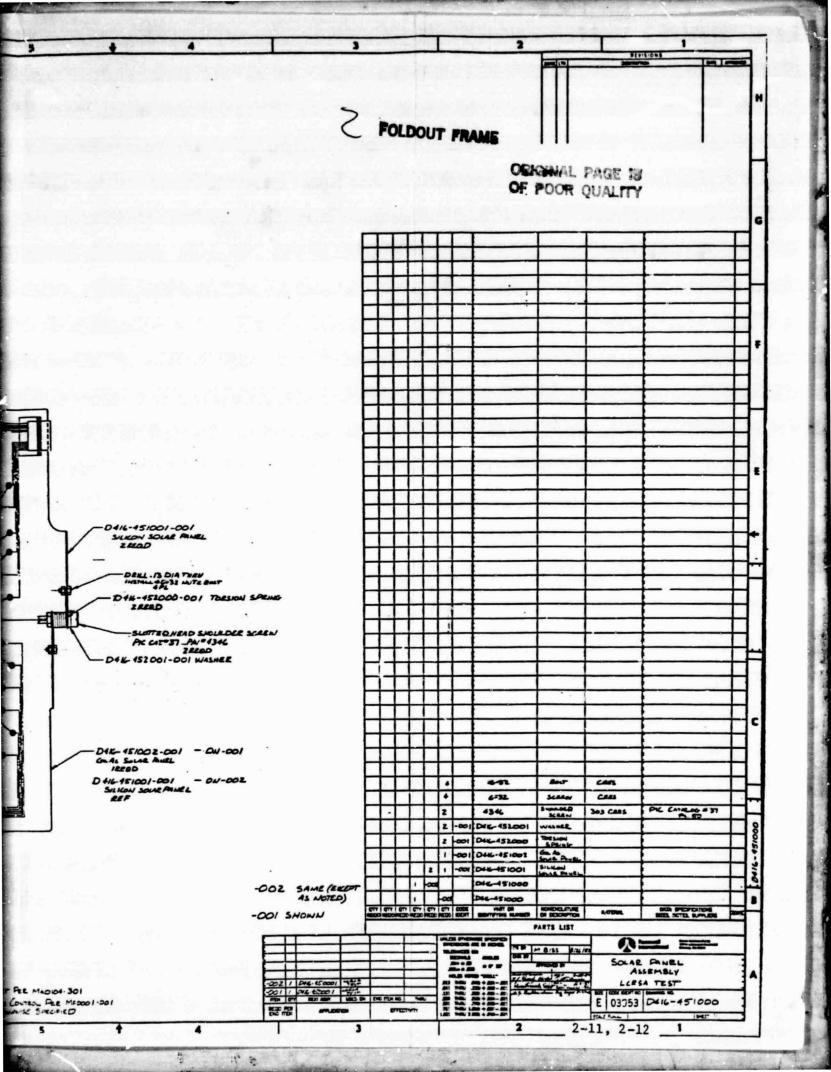
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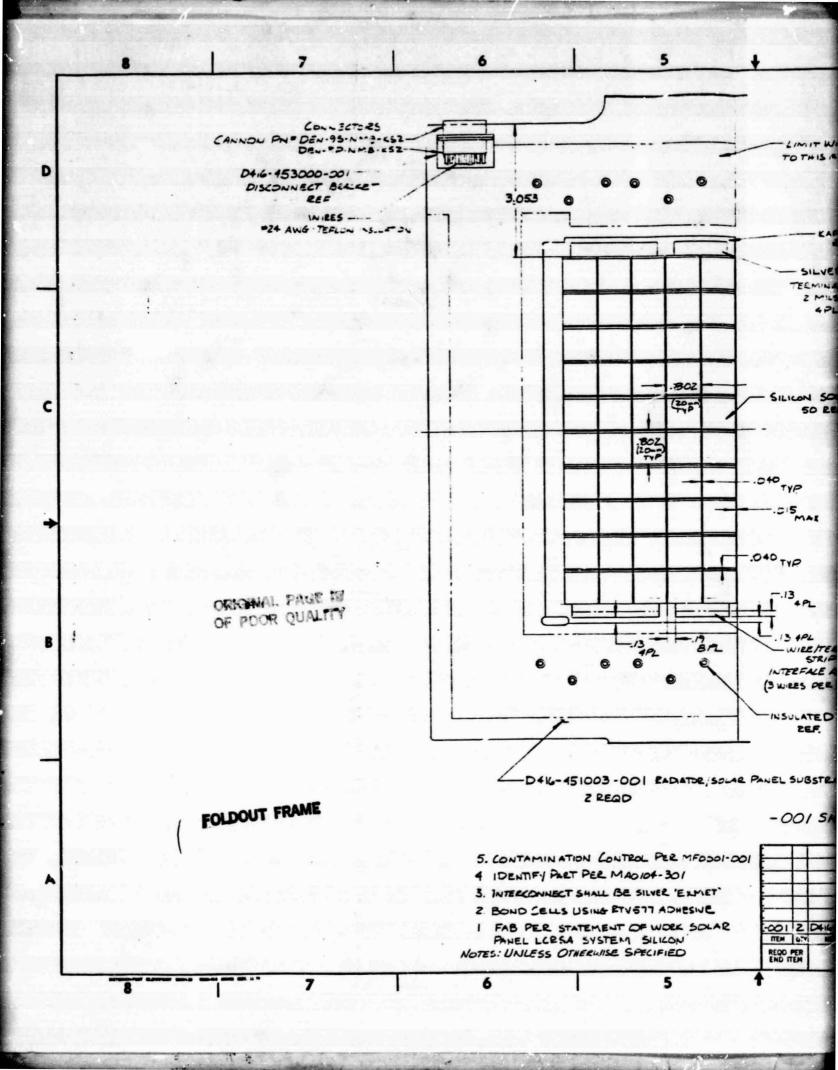


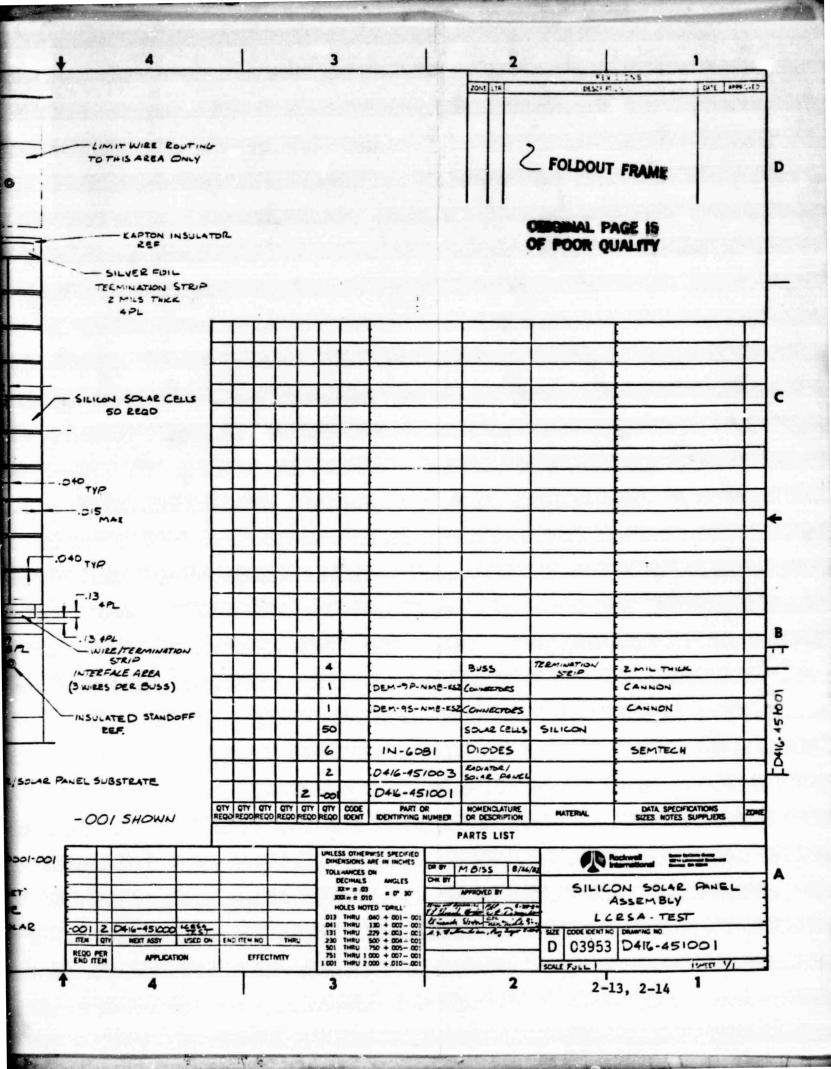
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1 1-001 Jab-15400 Inc. 10 2-9, 2-10 -COI SHOWN MATERIAL STATE OF STA 小二 == CONCUTATOR MEASURE STORES J 103953 1 244-45000 1 FOLDOUT FRAME

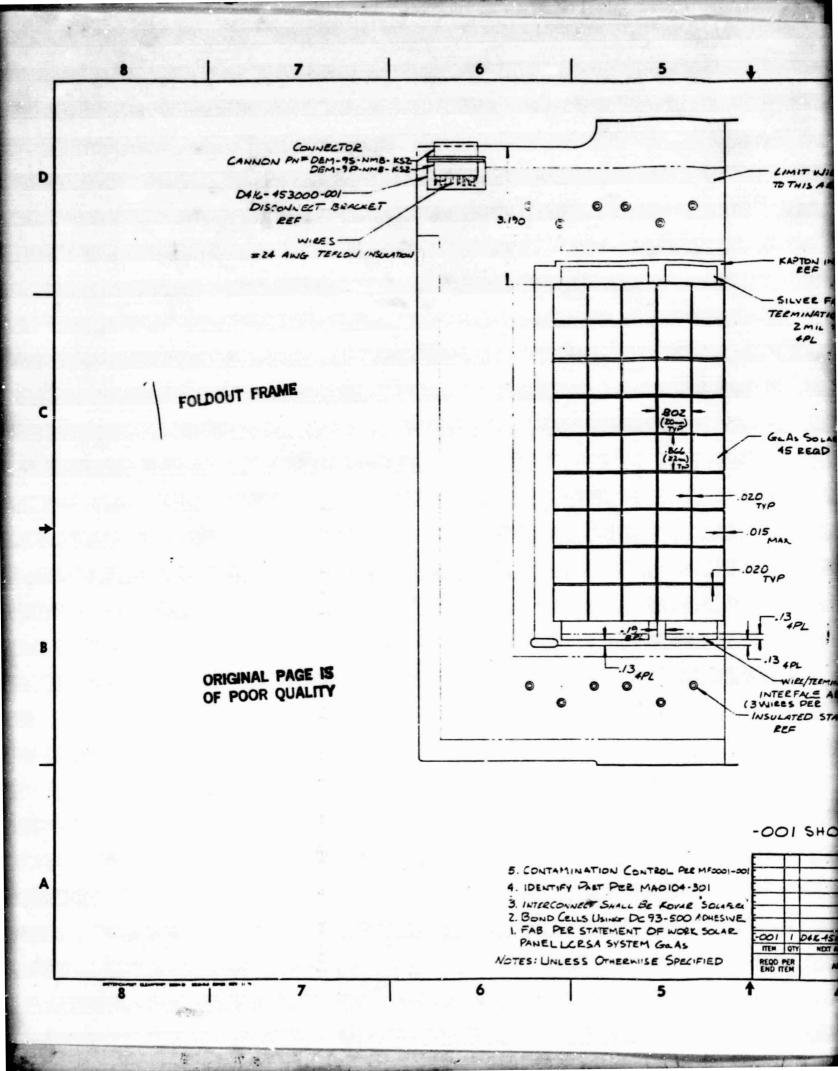
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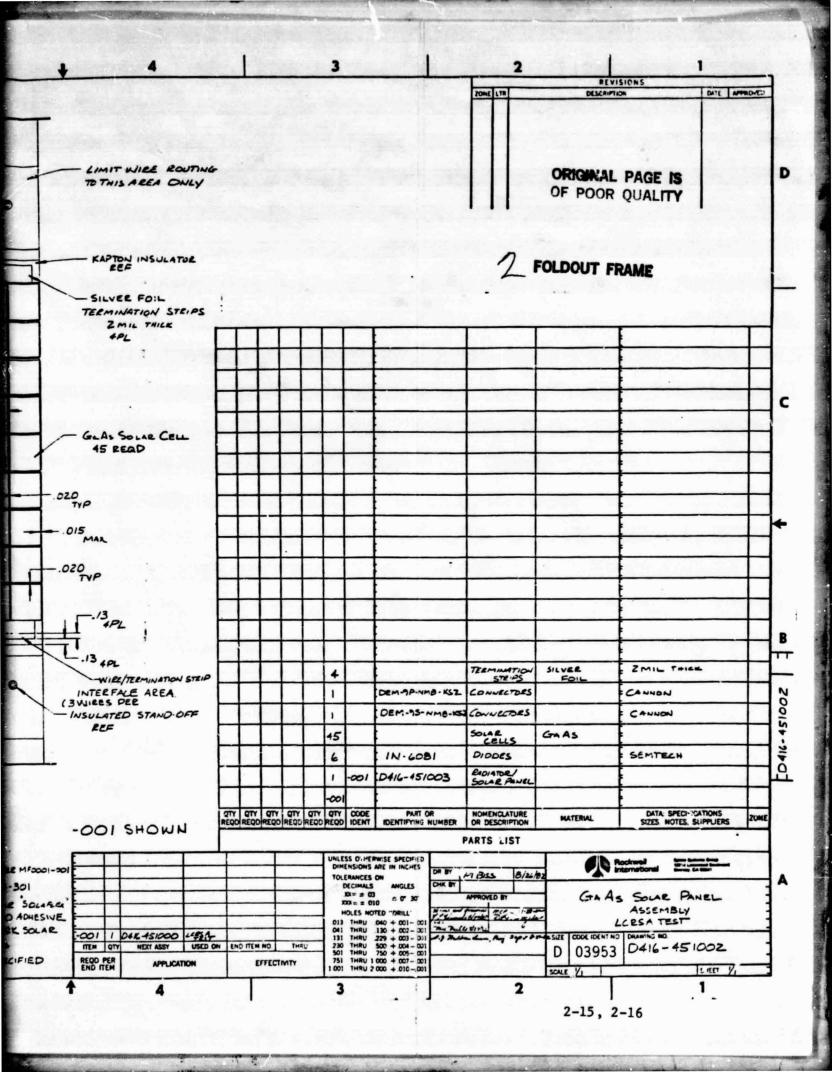
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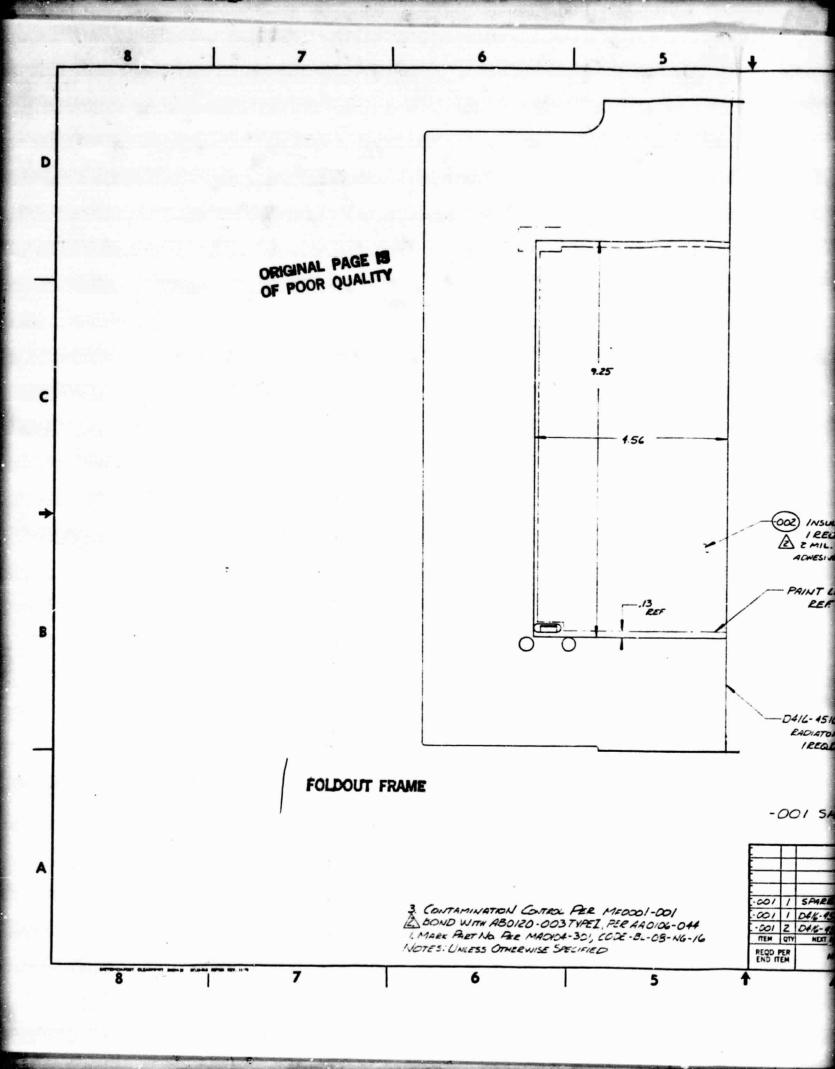


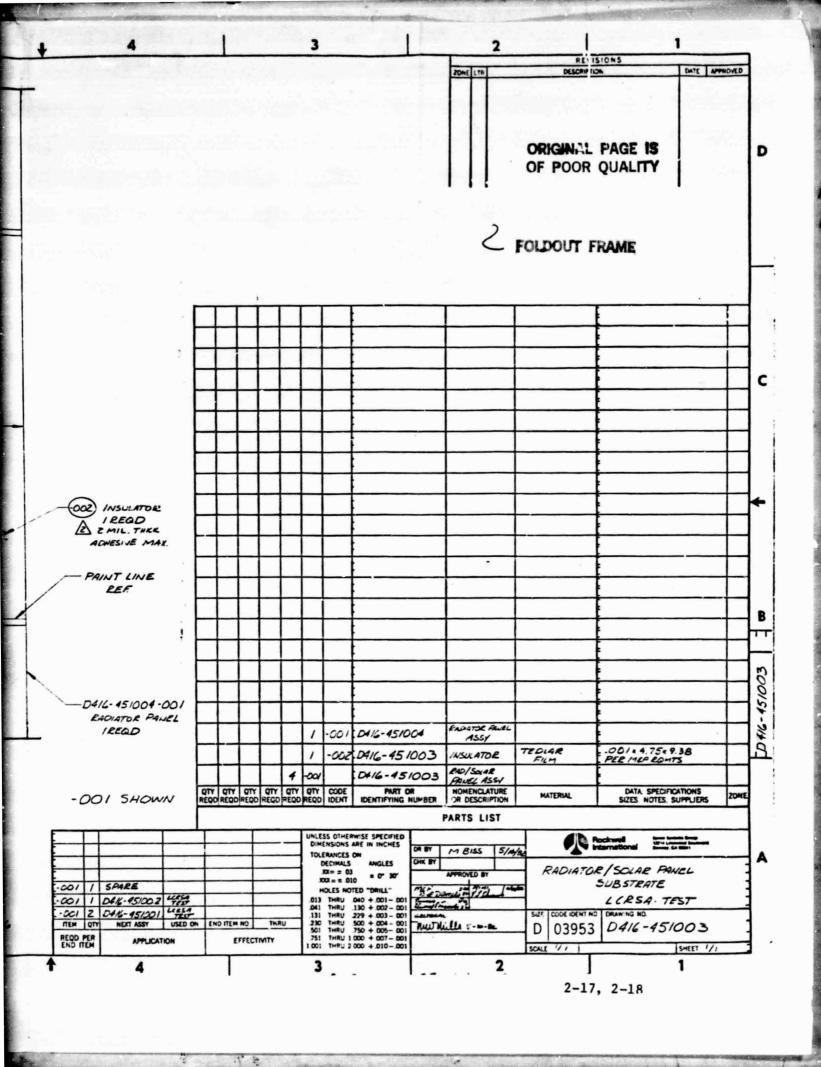


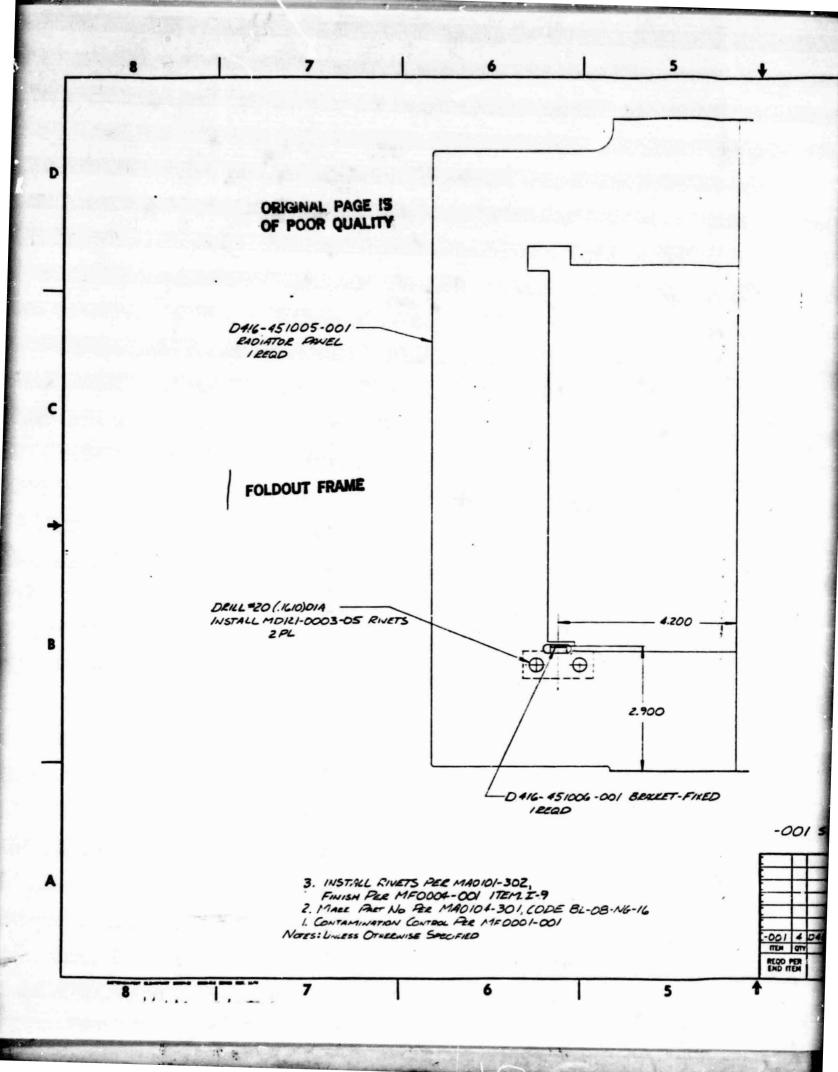




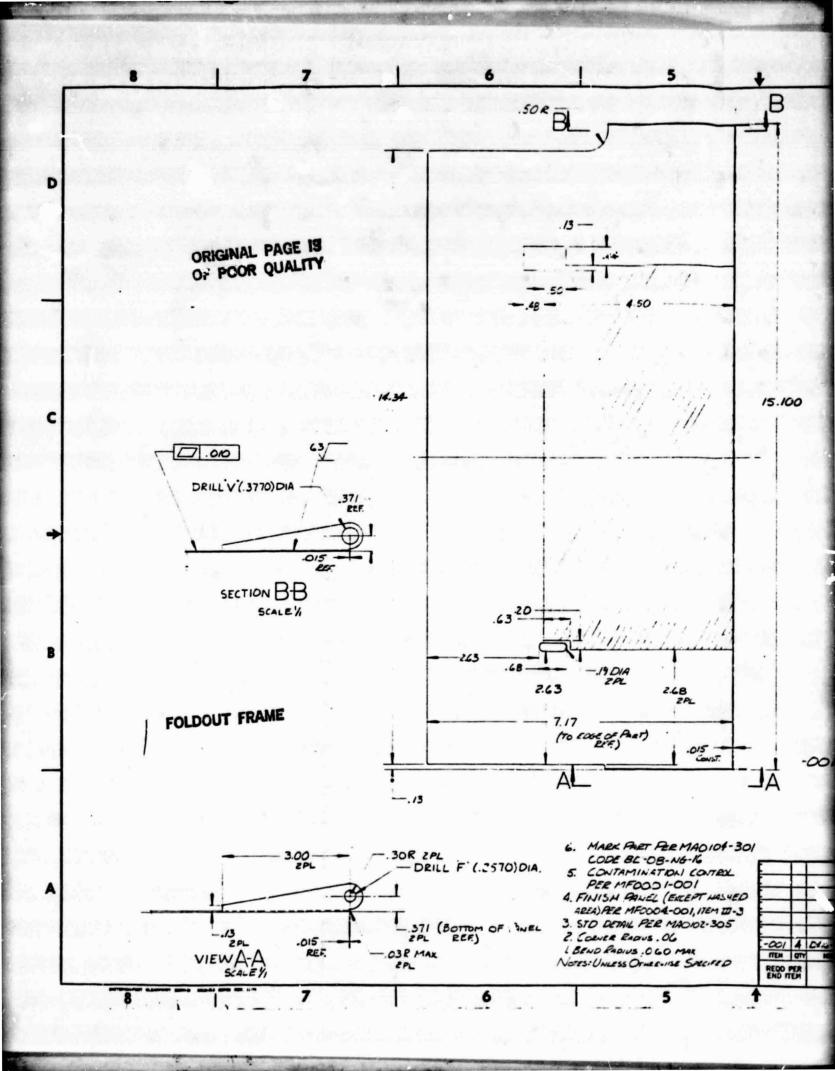


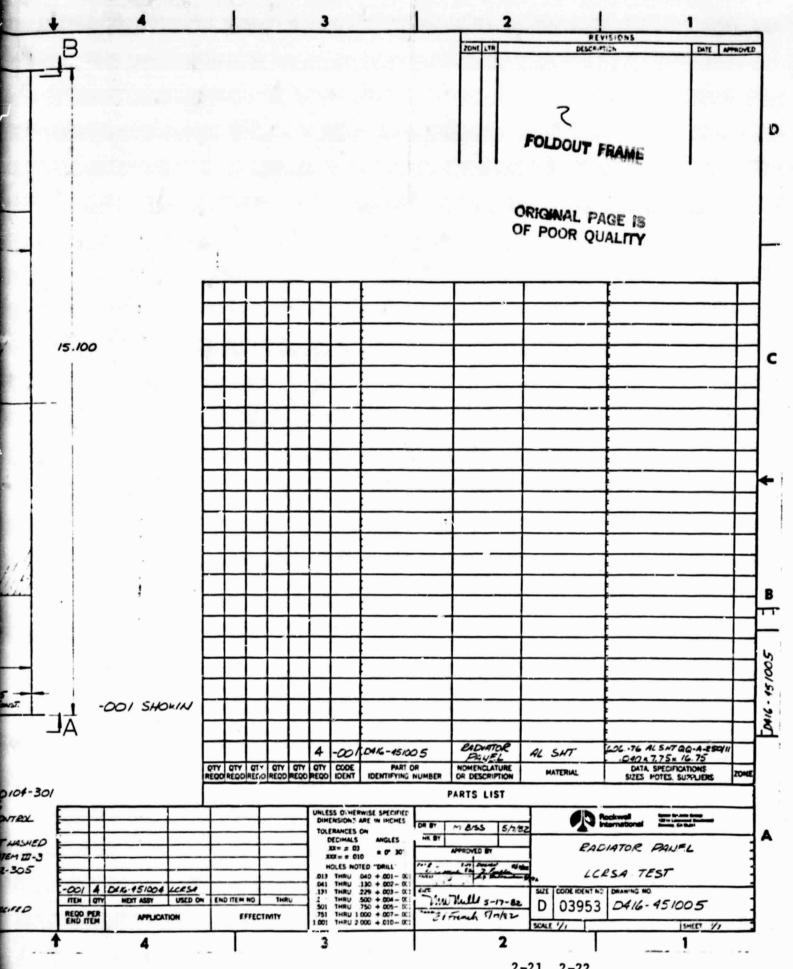


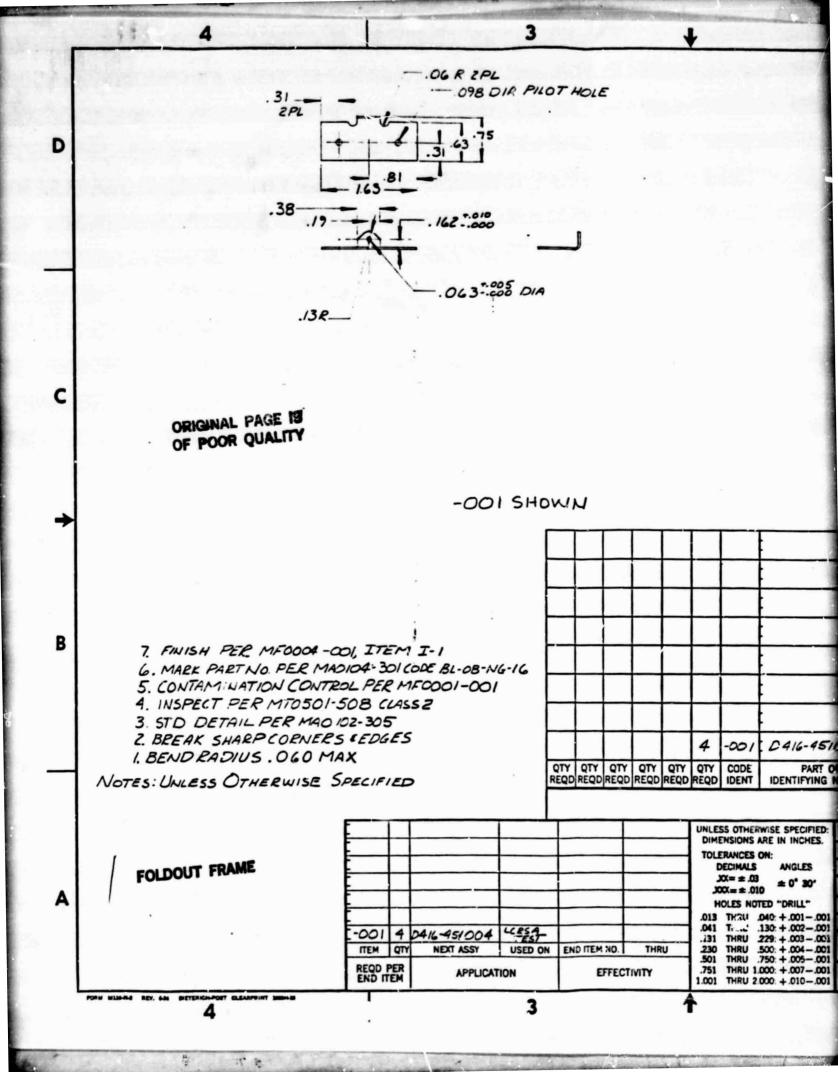


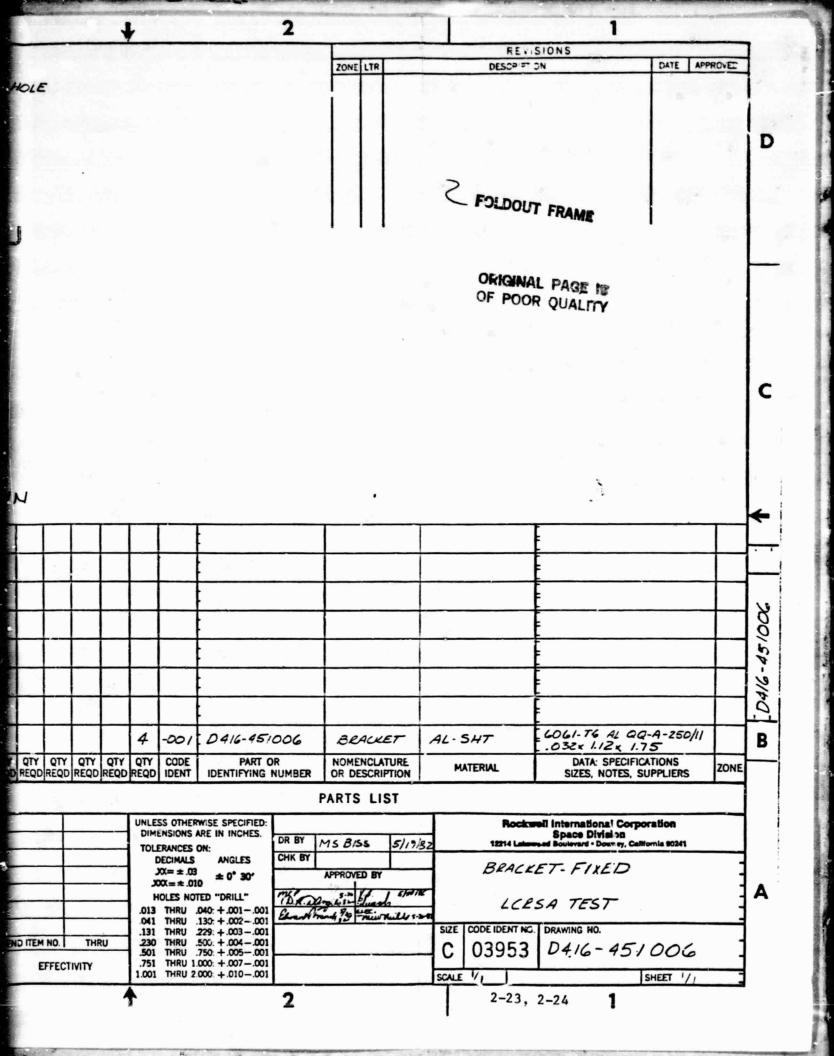


7. 3







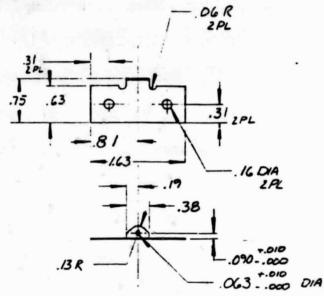


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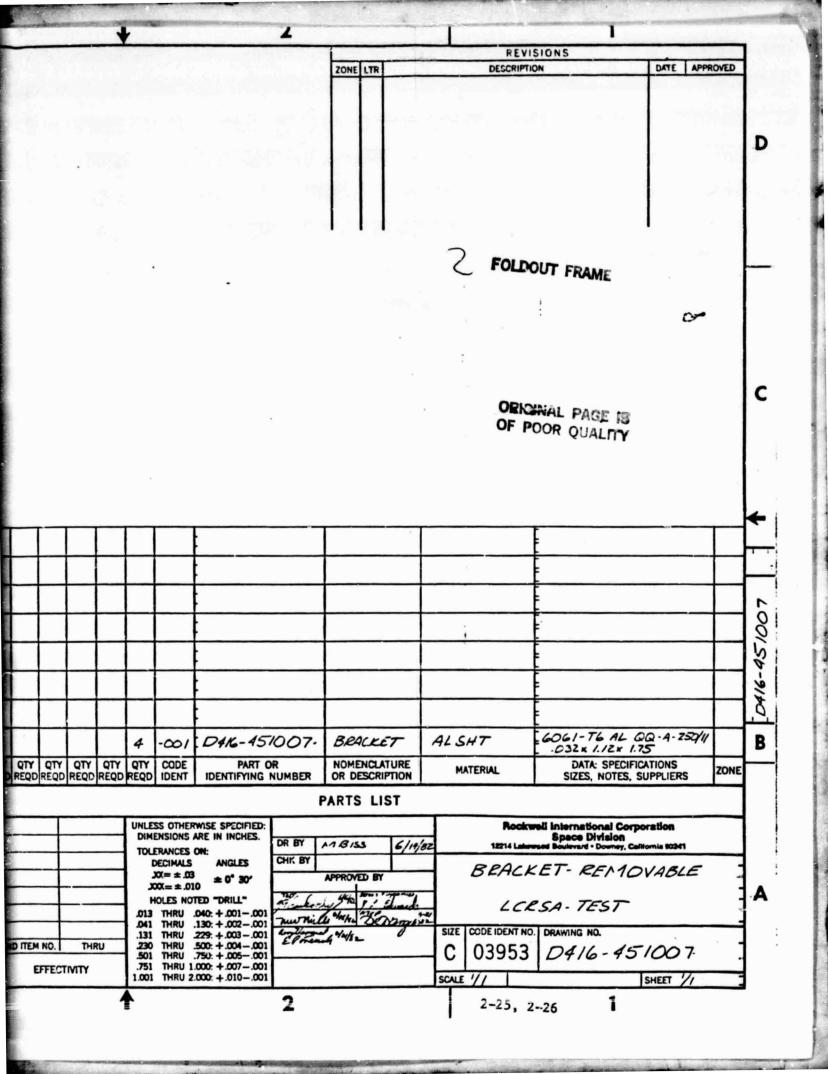
- 6. FINISH PANEL PER MF0004-001, ITEM III-3
- 5. MARK PART NO PER MADIO4-301 COX BL-08-NG-16
- 4. CONTAMINATION CONTROL PER MECOOI-DOI
- 3. STO DETAIL PER MACIOZ-305
- 2. BREAK SHALP CORNERS & EDGES
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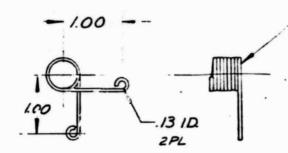
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- 6. STD DETAIL PER MADIOZ-305
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- 3. FINISH PER MF0004-001, ITEM I-23
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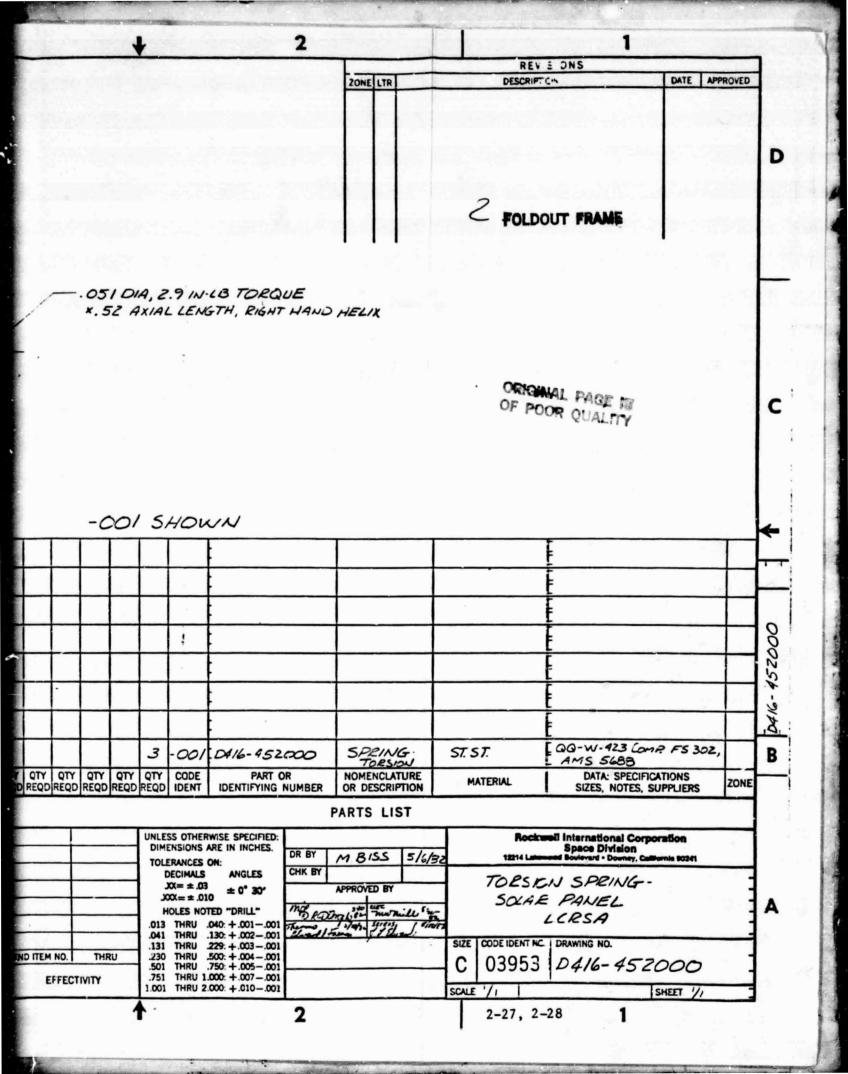
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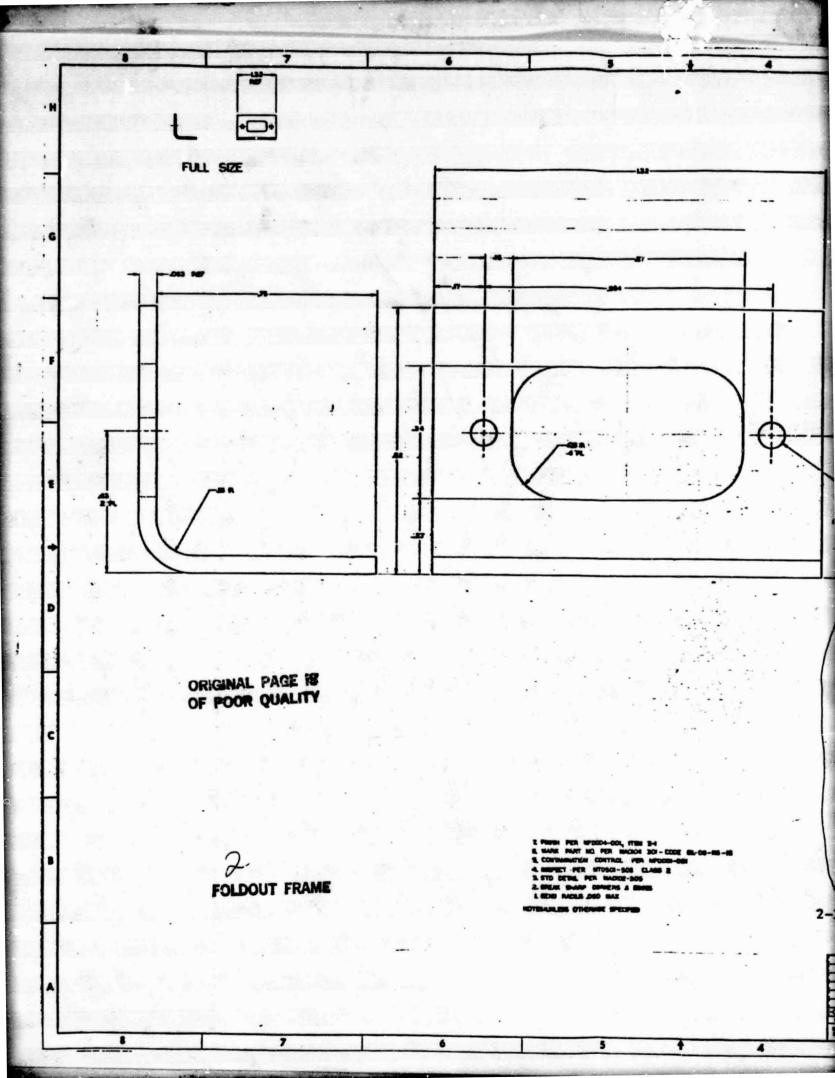
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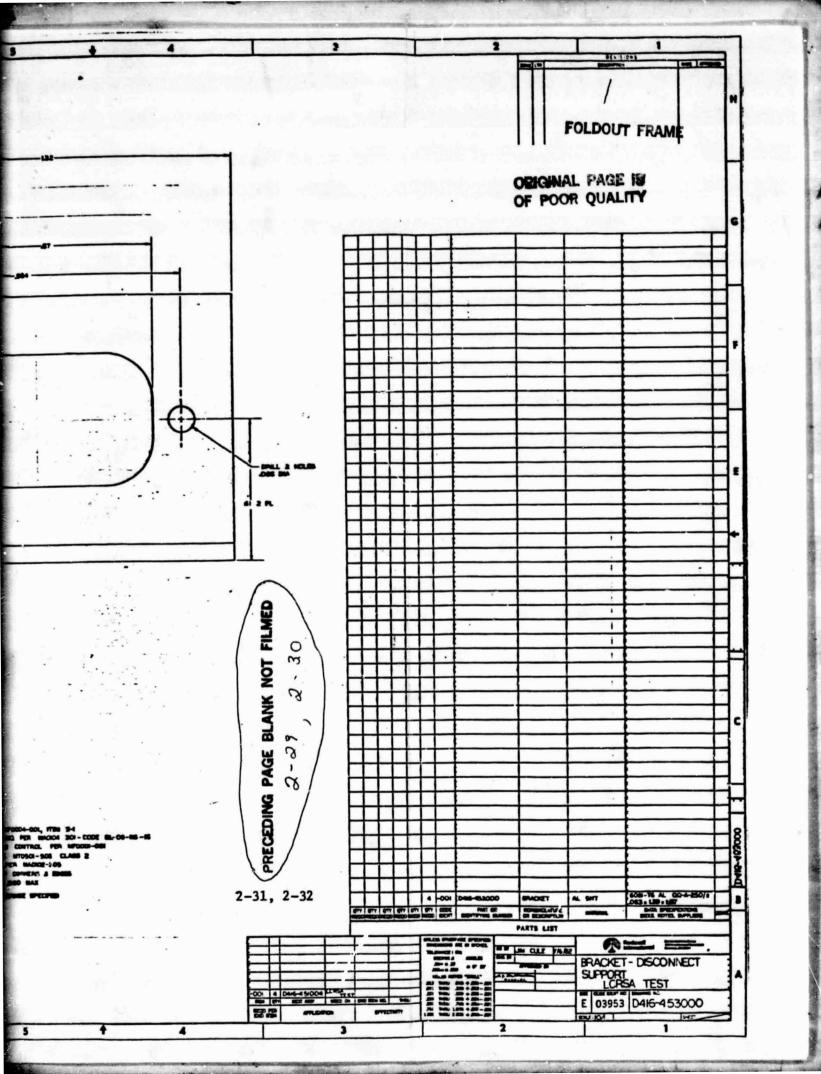
-001 SHOWN

HRU .040: +.001001 HRU .130: +.002001
IRU .229: +.003001
4RU .500: +.004001 4RU .750: +.005001

THERWISE SPECIFIED: ONS ARE IN INCHES. ANGLES ± 0° 30° RU .040: +.001-.001 RU .130: +.002-.001







13 12 11 10 9

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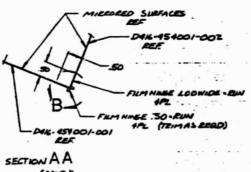
TAPE NIME LOWISE

TAPE NIME .50 WICE

VIEW B

SCALE 4,

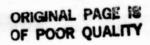
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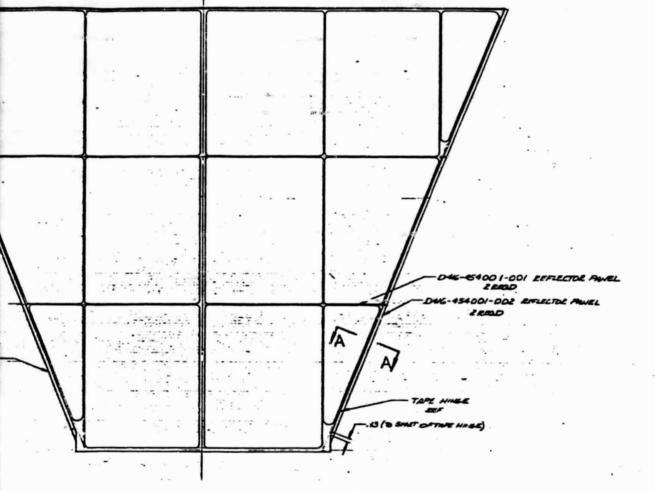


SECTION A A SCALE Y (TYPKAL ALL + COMERS) D4K-454001-002 -

FOLDOUT FRAME

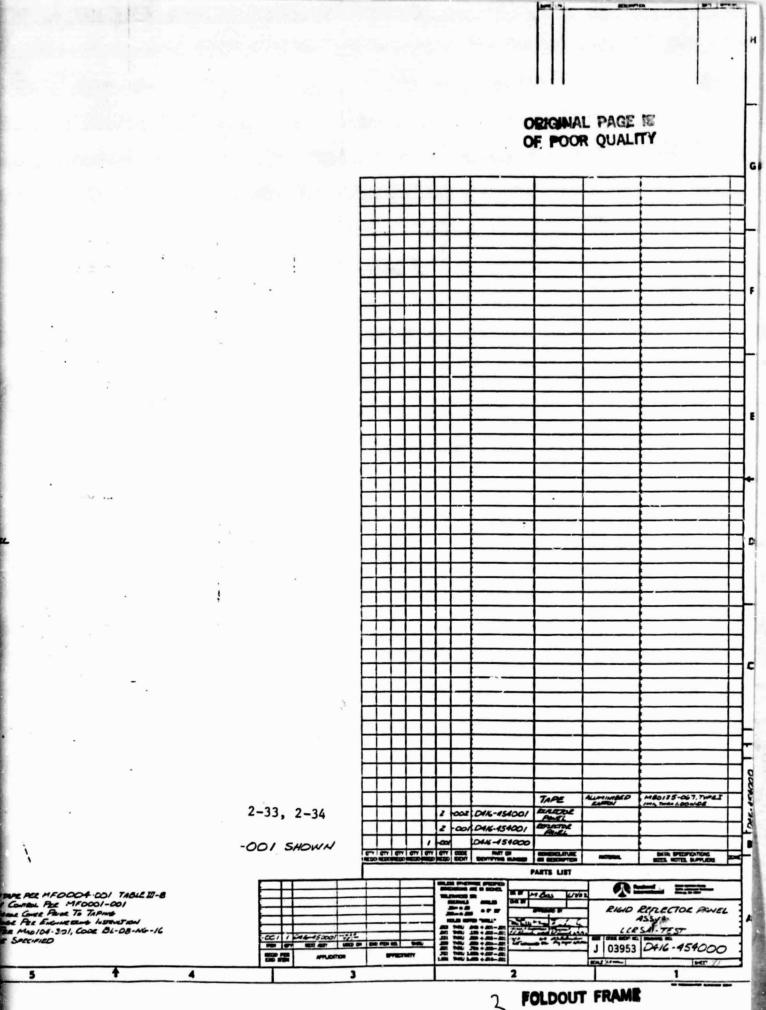
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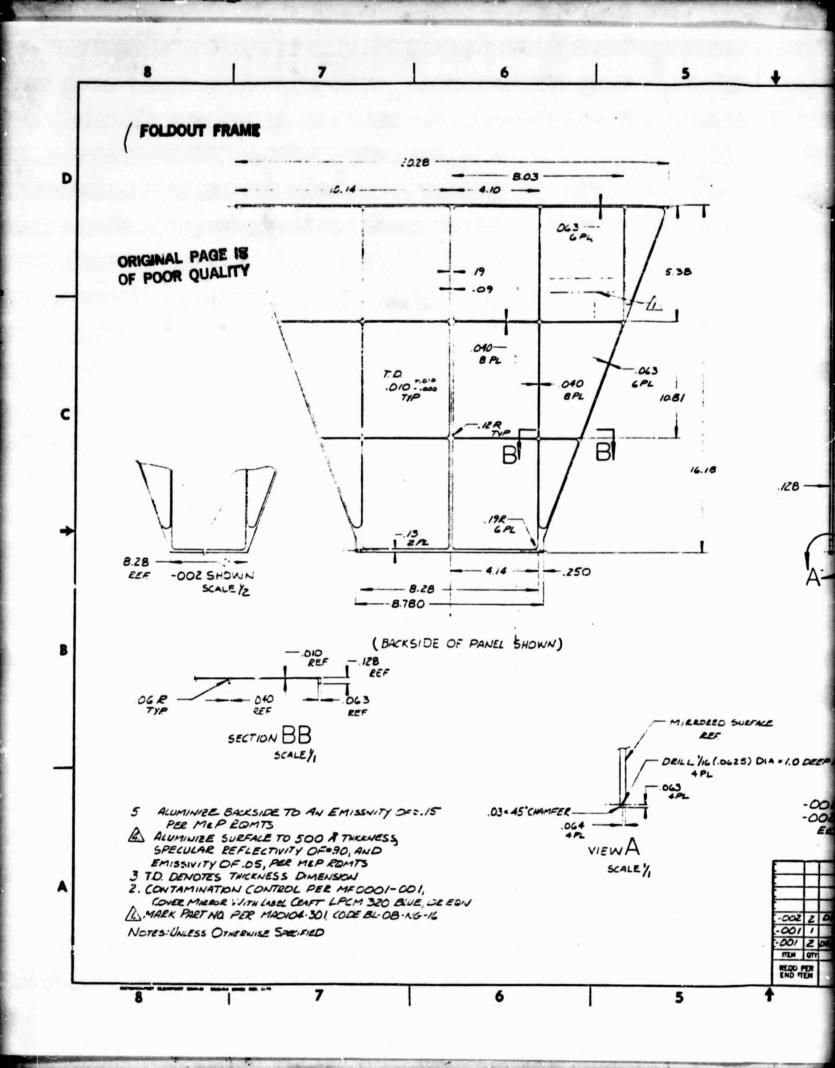


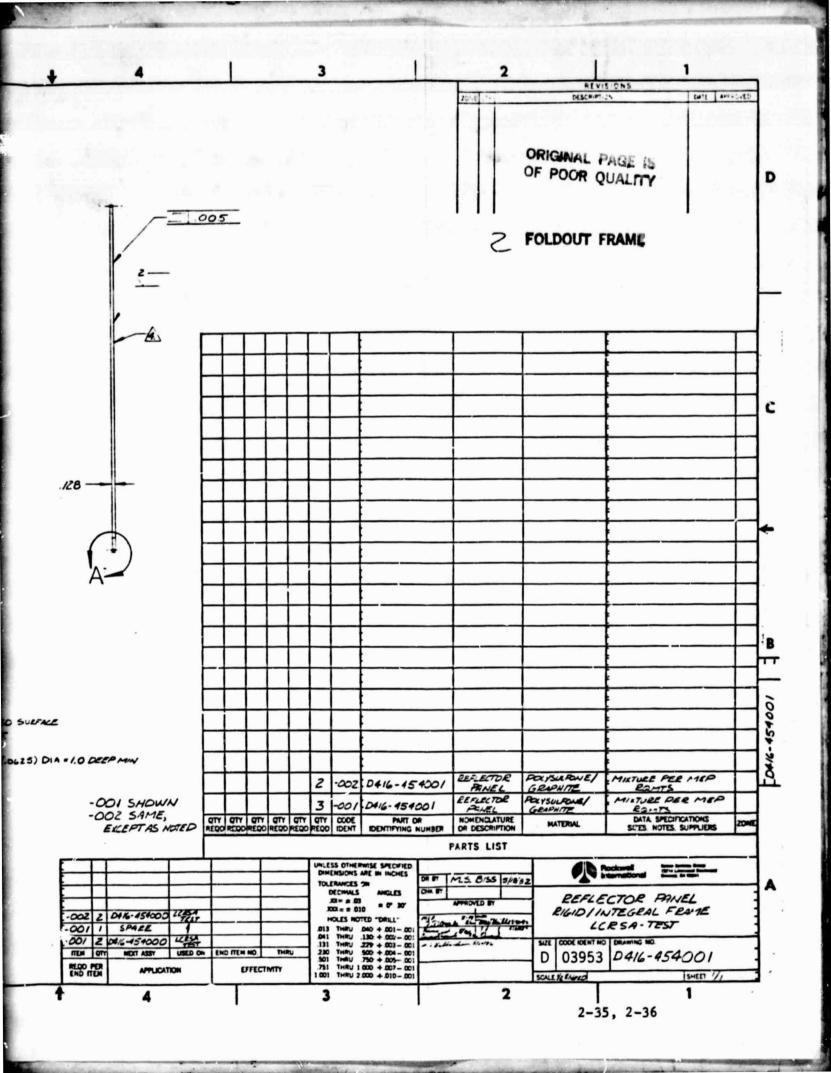


2 FOLDOUT FRAME

5. HANDLE EAPPLY TOPE PLE MFOODA-DOI TABLE II-8
4. CONTAMINATION CONTROL PLE MFDODI-DOI
3. PLEL BALL MARINE GAVE PARK TO TAPANG
2. APPLY TAPE. HADE FRE ENGINEERING INSTRUCTION
1. MARINE RETAIN FRE MAJORI SOI, CODE BL-DB-NG-IG
NOTES: UMLESS KILLE SPECIFIED







11

S OF AMEL SEPTEMBER

TO AMEL SEP

(-DOZ DALY)

14

13

In commence

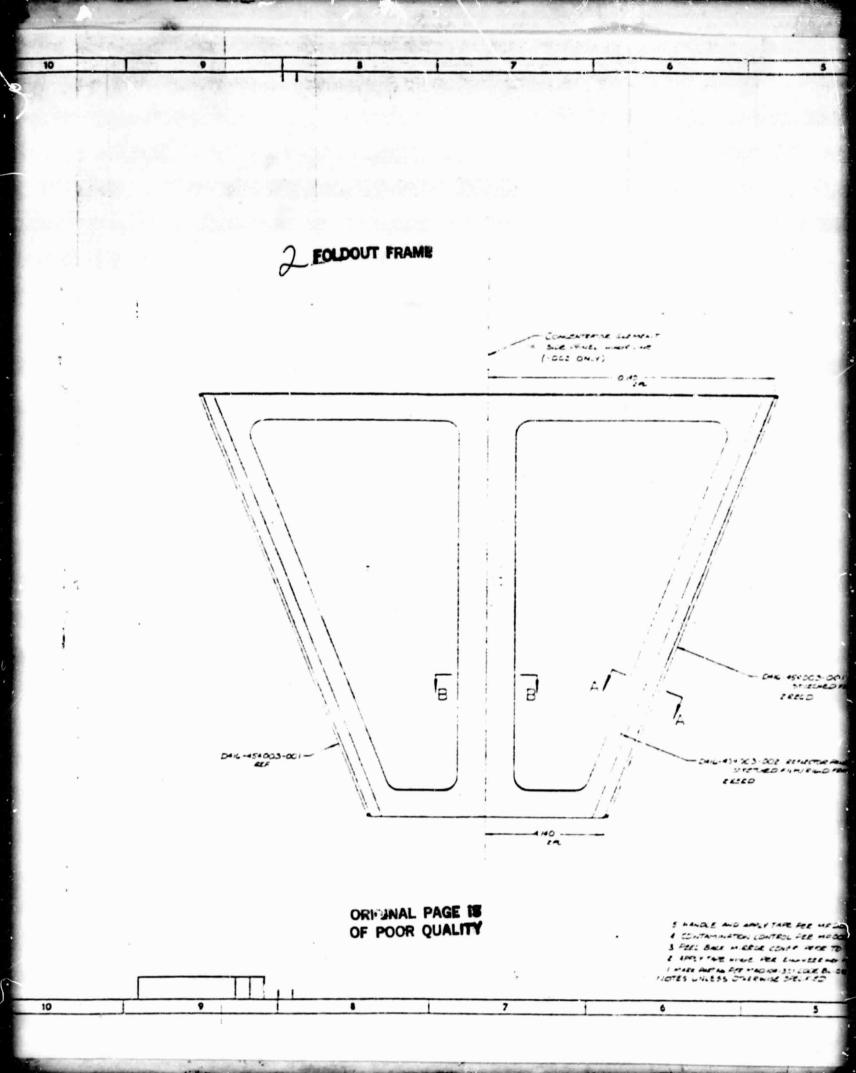
FOR CLASIN

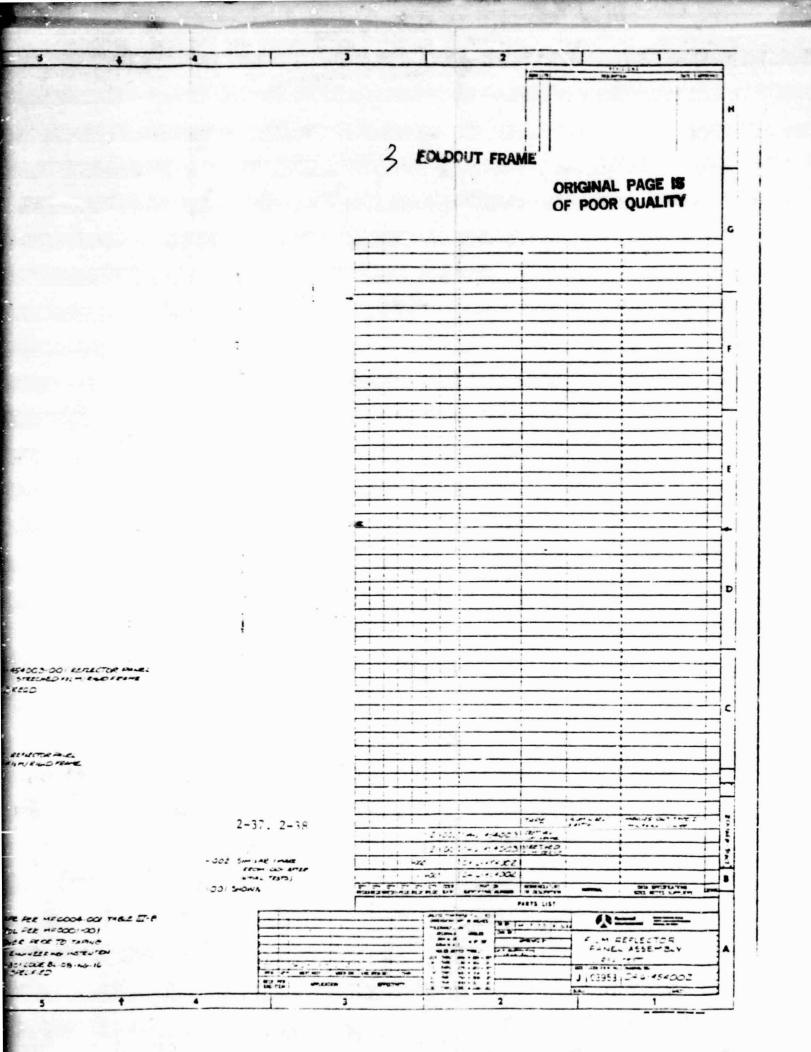
SECTION AA

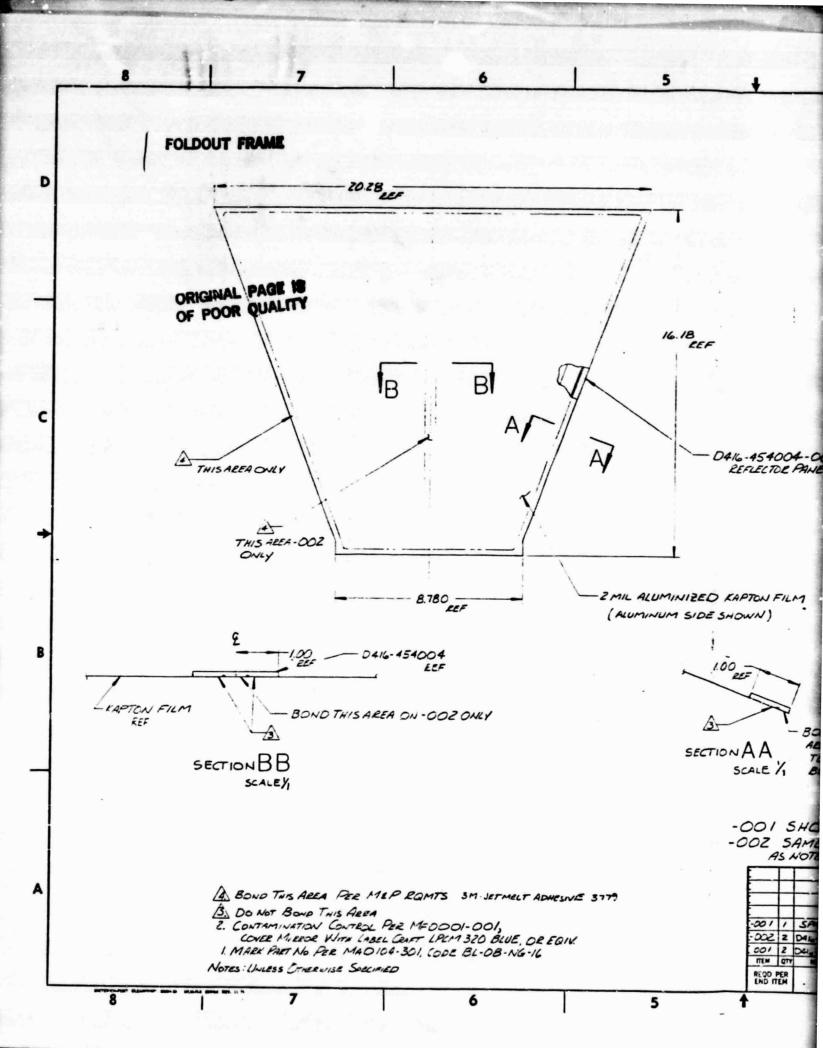
SECTIO

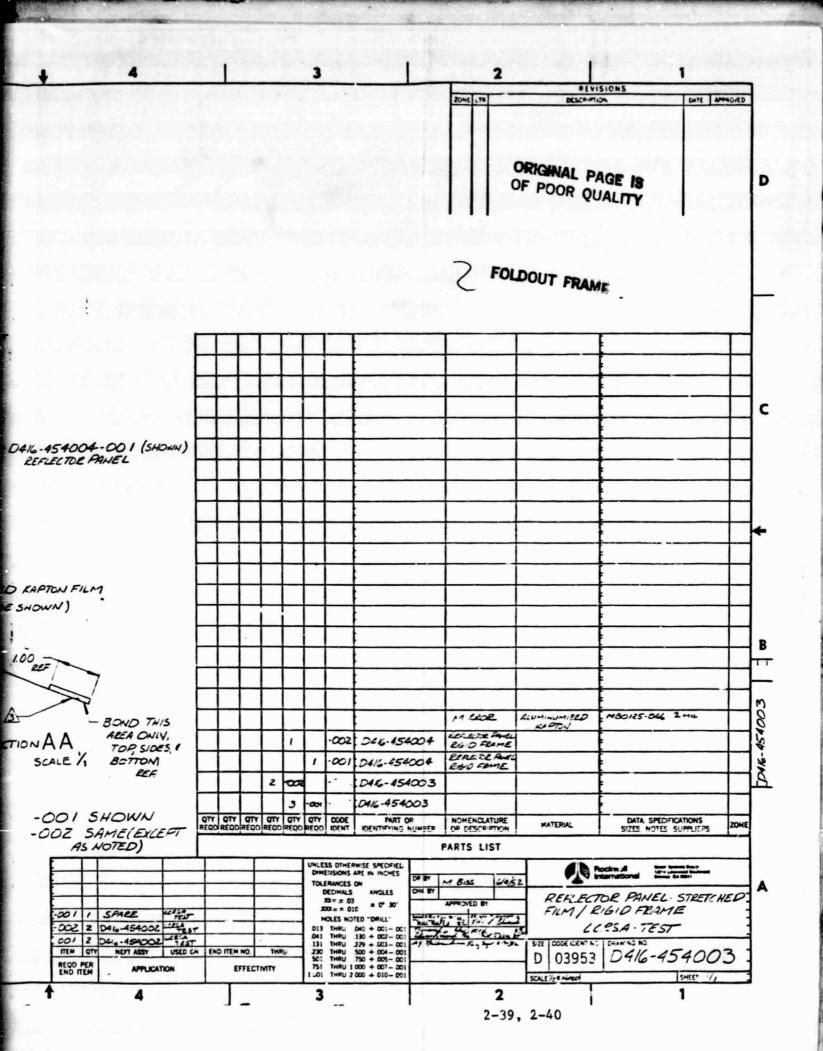
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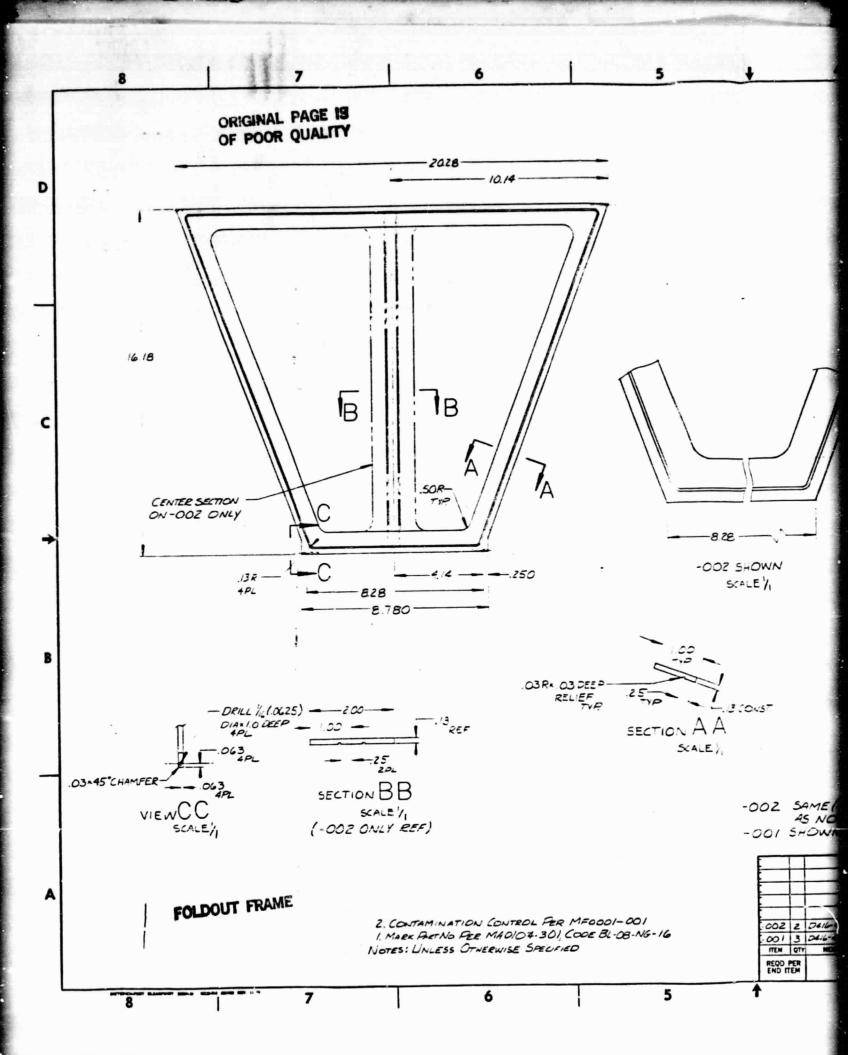
14 13 1 12 11 10

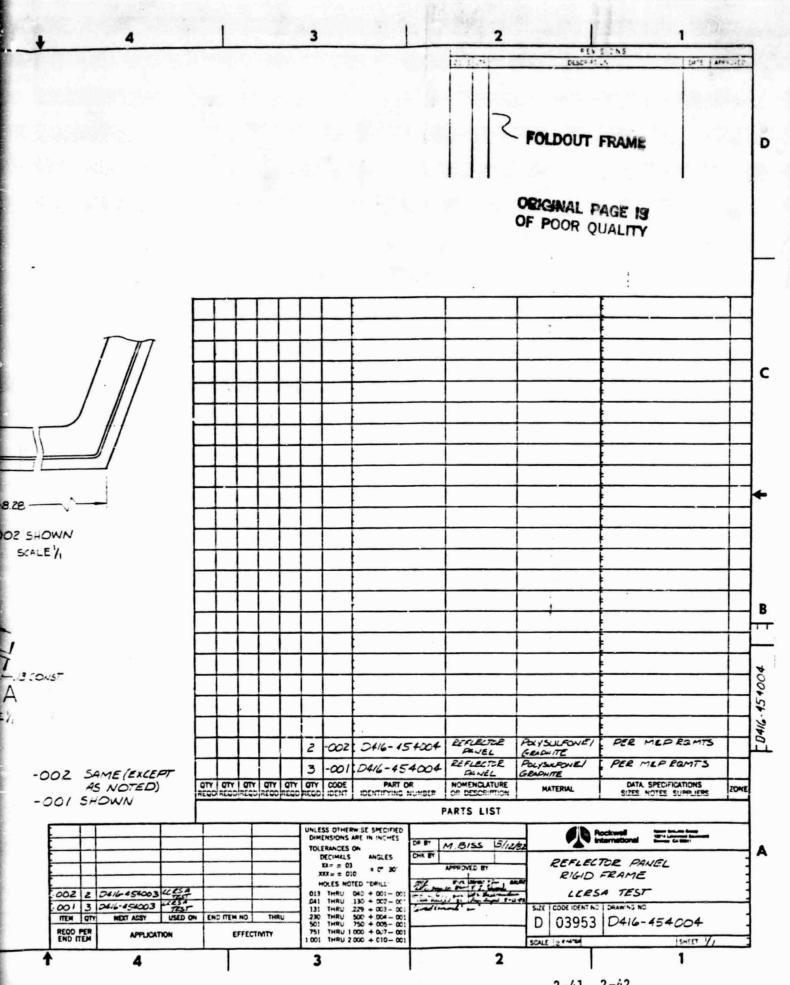




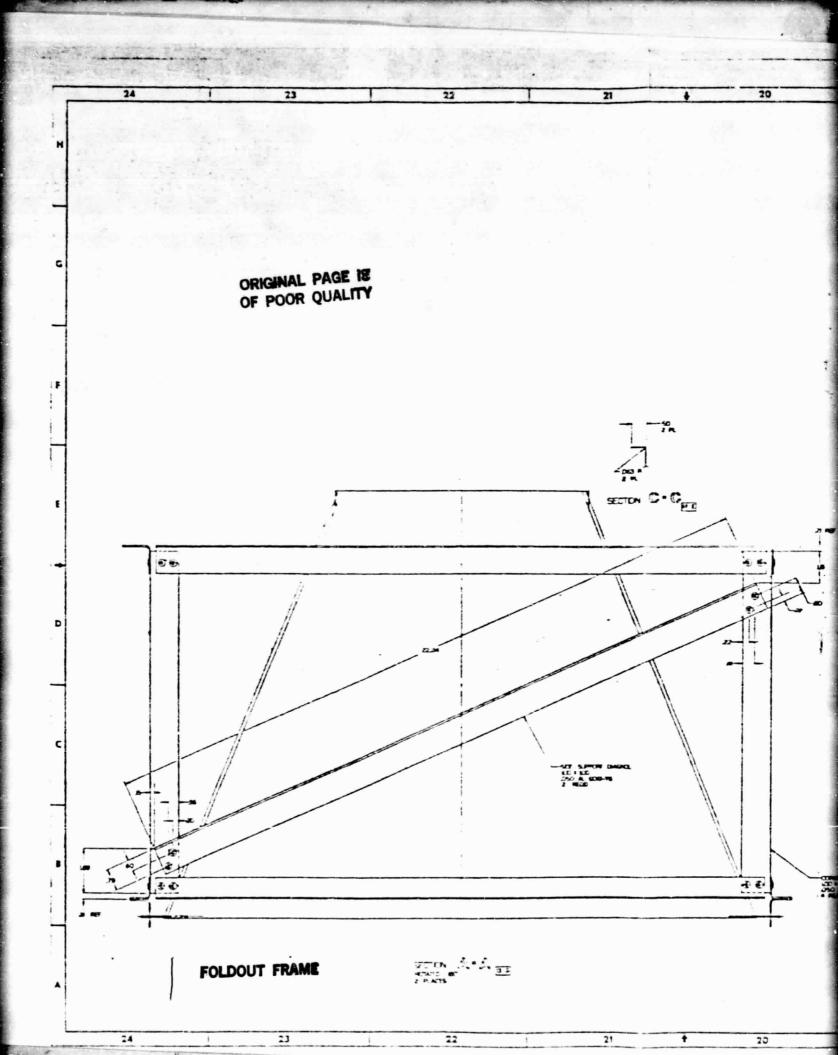


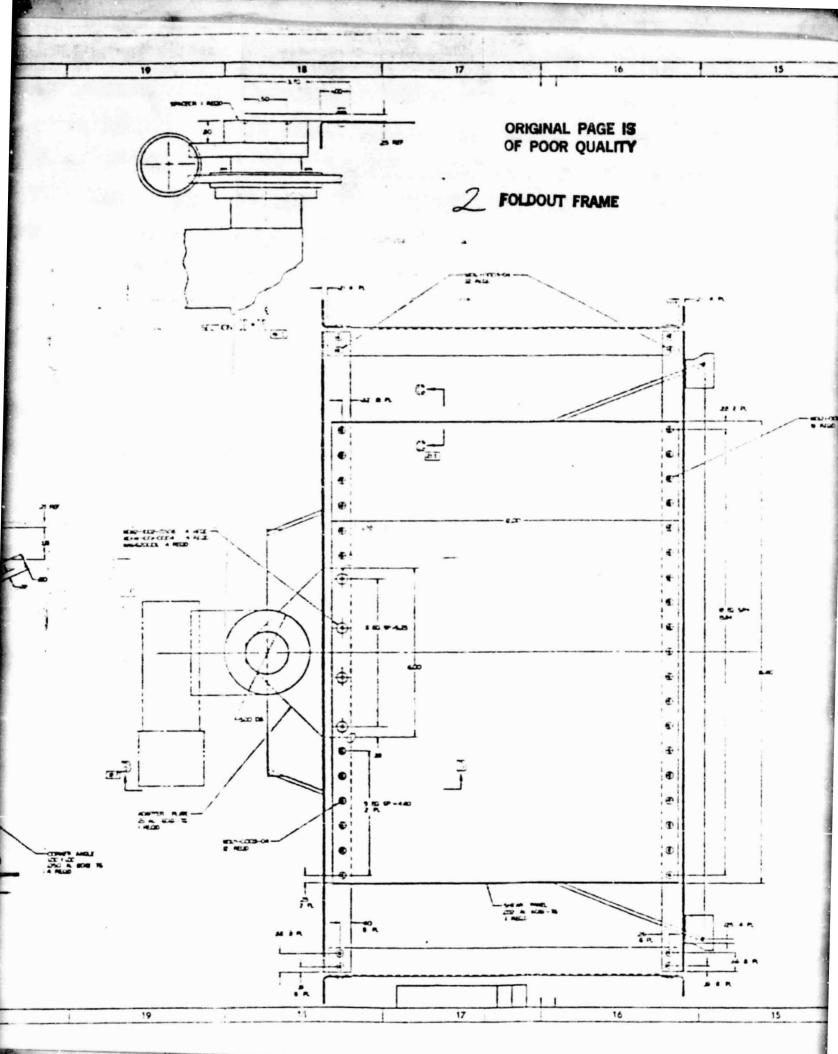


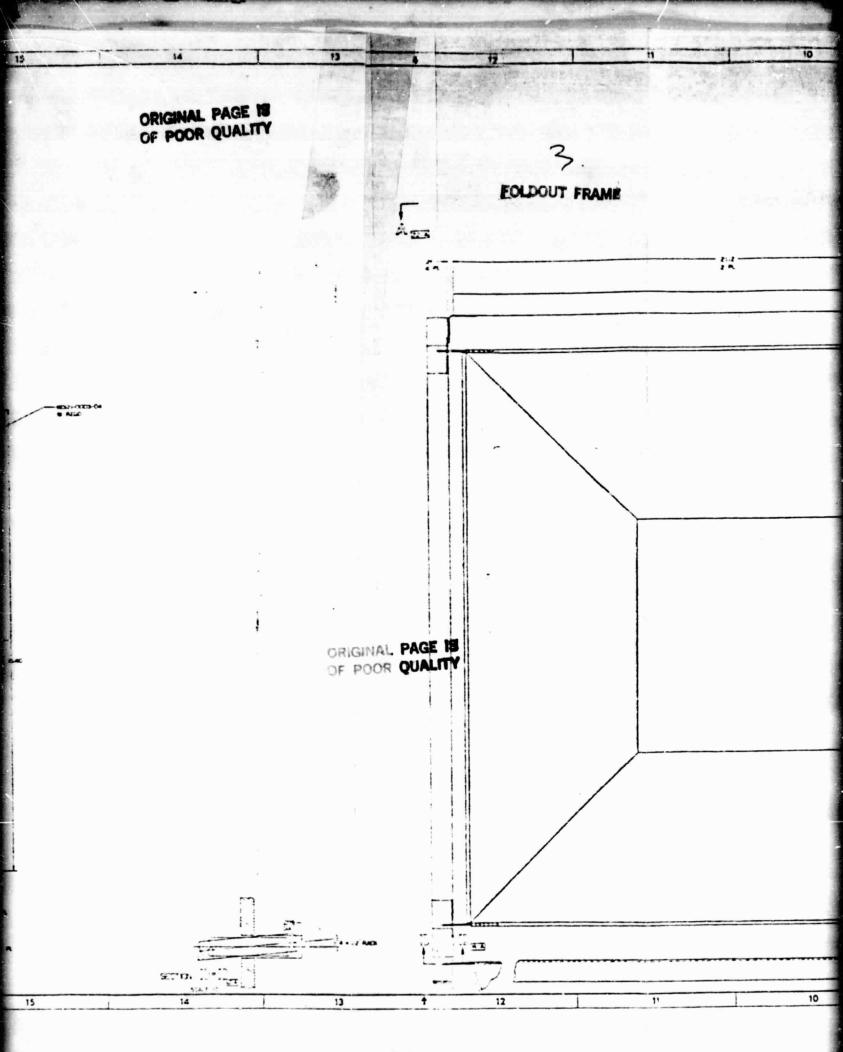


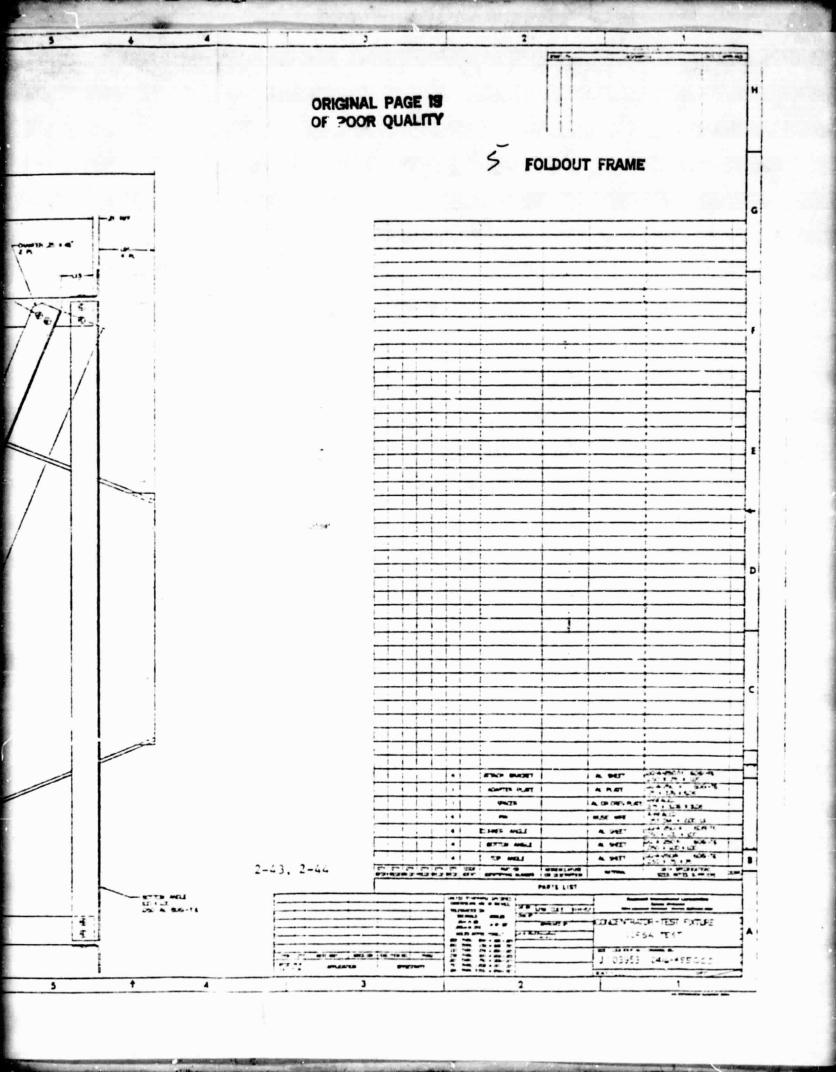


2-41, 2-42









3.0 PRELIMINARY MANUFACTURING FLOW AND BUILD PLAN

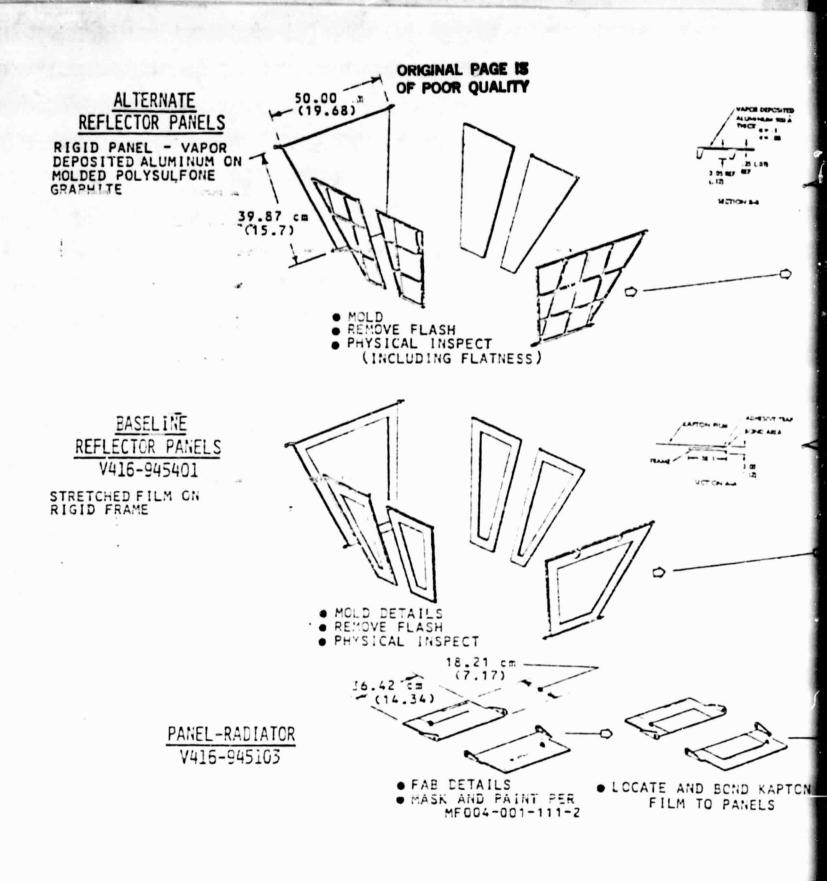
3.1 DISCUSSION

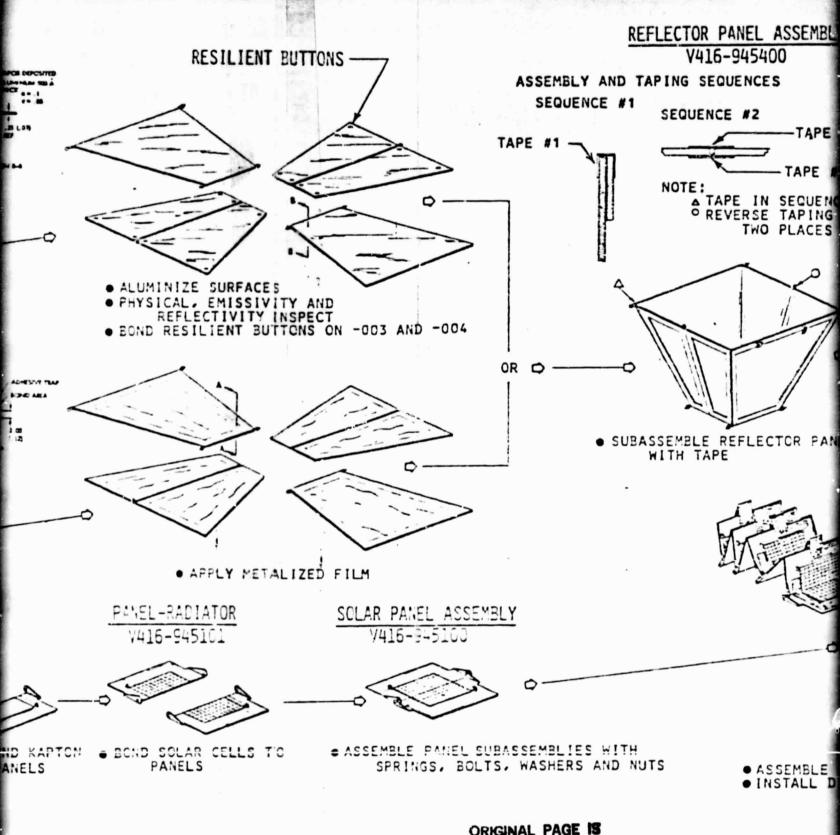
In Section 8.3 of Volume 1 a test plan is described for a ground and flight demonstration array. The following two diagrams show the fabrication sequence for assembling the basic unit of such an array (see Section 4.0 of Volume 1), consisting of one pair of canister-mast assemblies and eight rows of concentrators. The first diagram shows assembly of concentrator rows and solar panels. The second diagram describes construction of the structural elements and their integration with the rows of concentrator elements.

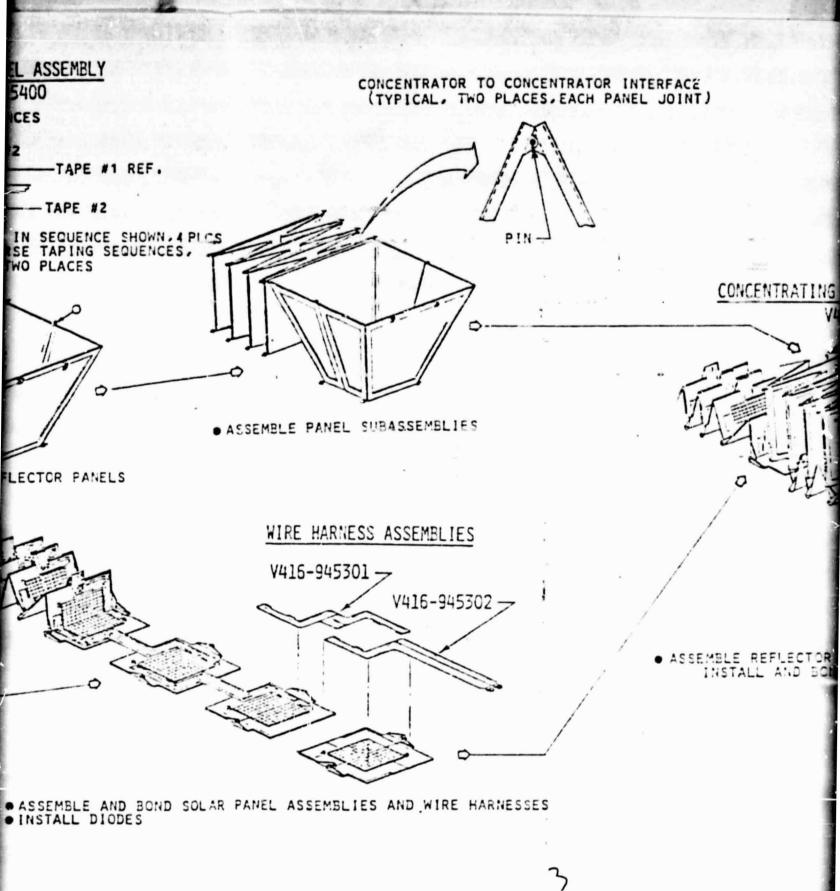
3.2 FLOW AND BUILD DIAGRAMS

J. 5

- 1. Assembly of Concentrator Elements
- 2. Assembly Sequence for Flight Test Configuration



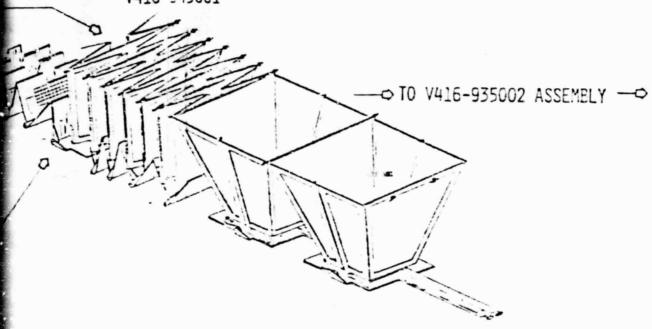




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CONCENTRATING ELEMENT STACK ASSEMBLY V416-945001



MBLE REFLECTOR PASEL ASSEMBLY AND SOLAR PANEL ASSEMBLY, INSTALL AND BOND CLIPS

ASSEMBLY OF CONCENTRATOR ELEMENTS

PRELIMINARY MANUFACTURING FLOW AND BUILD PLAN
LOW CONCENTRATION RATIO SOLAR ARRAY

PREPARED BY: KM HICKS DEPT.: 761 EXT.: 36/8
DATE: JAN 19,1983 REVISED: FEB. 22,1983

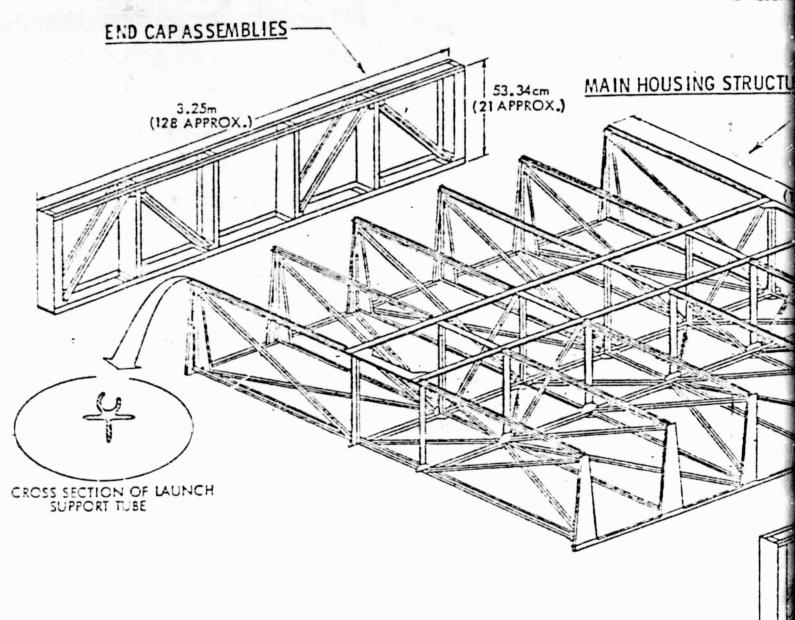
3-3, 3-4 4 FOLDOUT FRAME

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OF POOR QUALITY

HOUSING ASSEMBLY V416-935101

REPRESEN



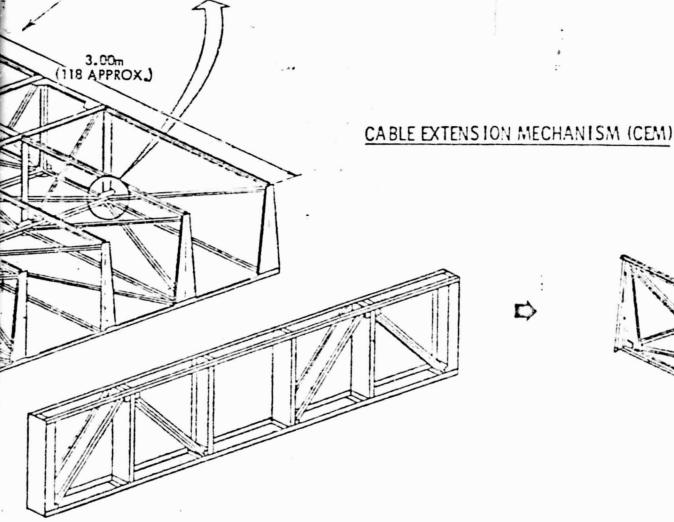
- FABRICATE DETAILS
- ASSEMBLE STRUCTURE AND INSTALL LAUNCH SUPPO

REPRESENTATIVE RIVETED JOINT CONFIGURATION

REFLECTO

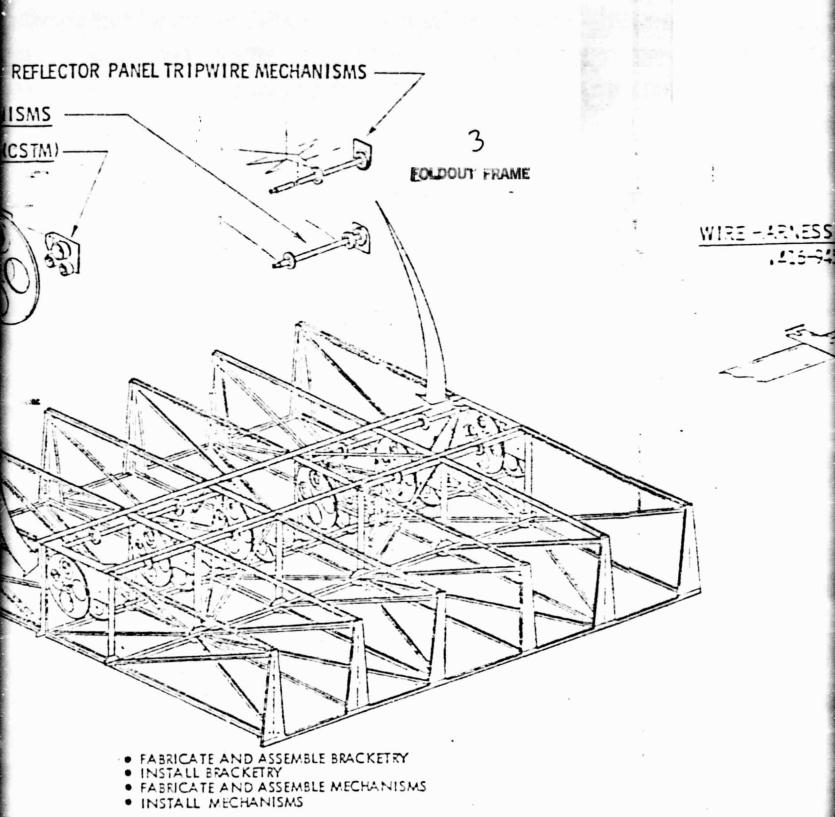
SOLAR PANEL TRIPWIRE MECHANISMS

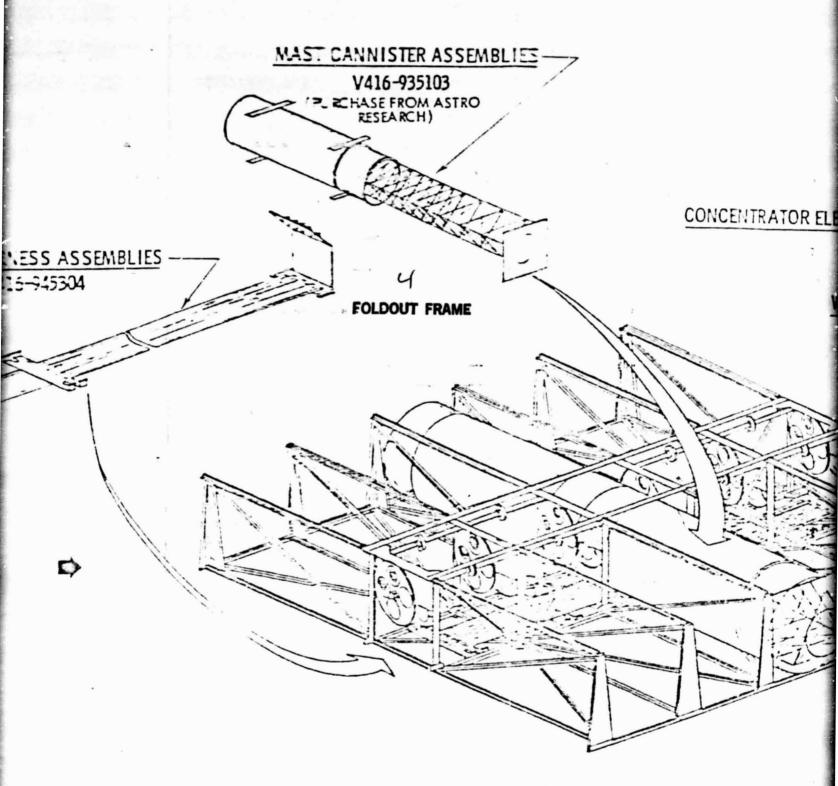
CONCENTRATOR STACK TRANSLATION MECHANISM (CSTM)



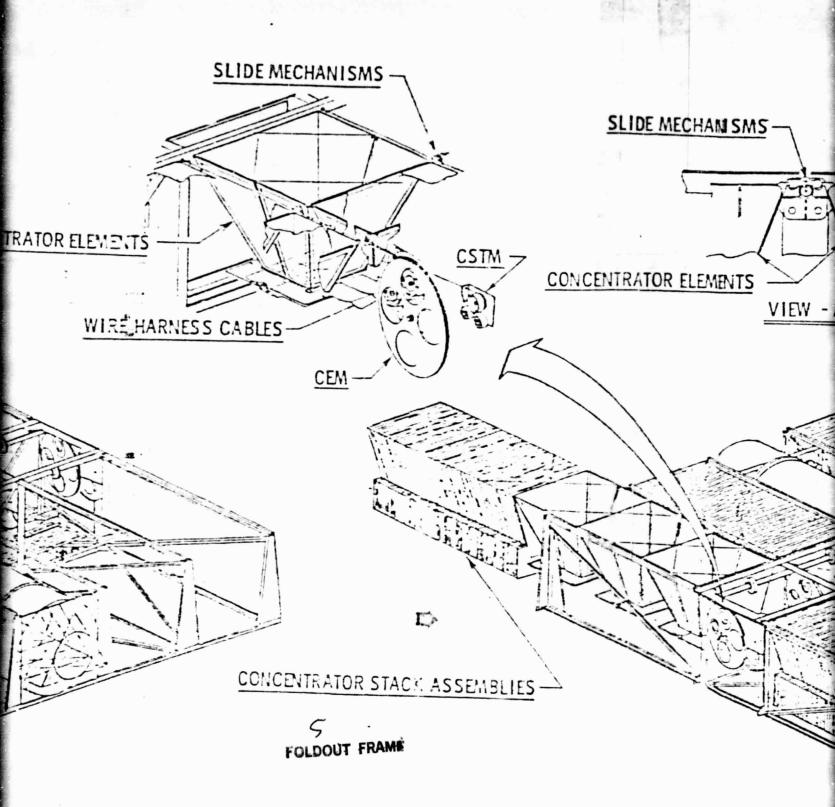
H SUPPORT TUBES

STRUCTURE





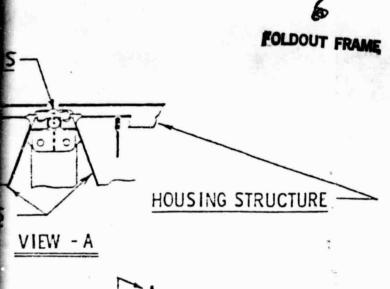
- NSTALL WIRE HARNESS ASSEMBLIES
 WISTALL MAST/CANNISTER ASSEMBLIES



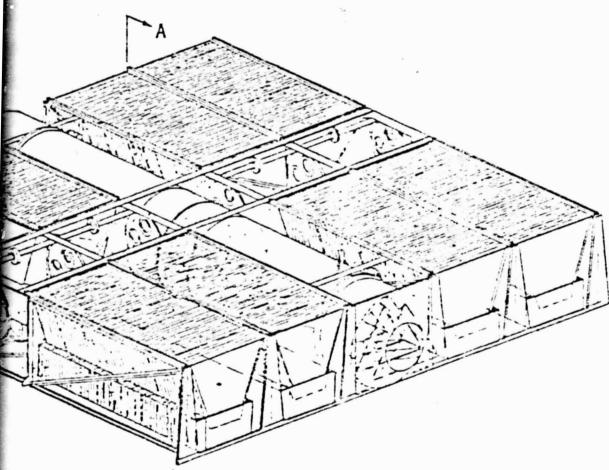
IES MBLIES

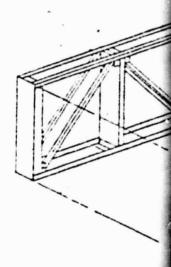
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INSTALL CONCPERFORM



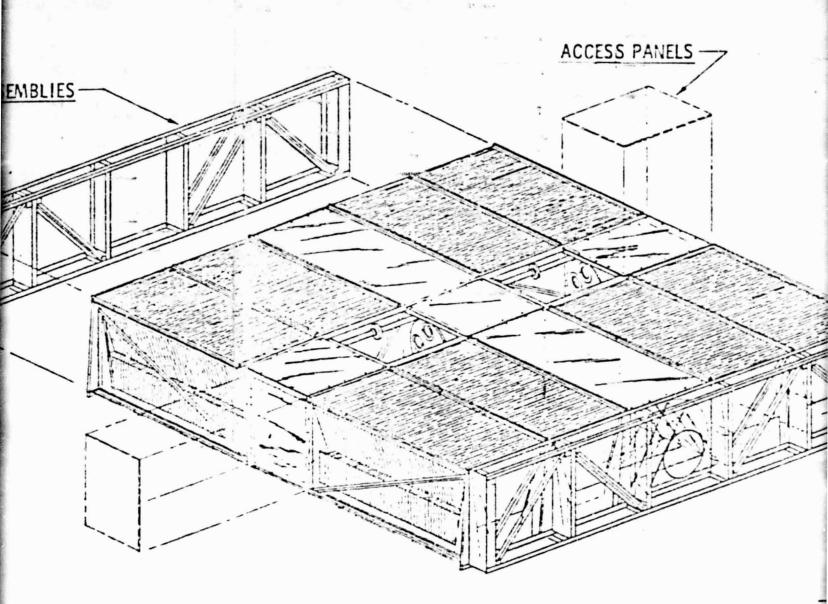
END CAP ASSEMBLIES





- INSTALL CONCENTRATOR STACK ASSEMBLIES AND SLIDE ASSEMBLIES CONCURRENTLY
 PERFORM ELECTRICAL HOOKUP AND CHECKOUT

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- INSTALL END CAP ASSEMBLIES SECURE TO MAST/CANNISTER ASSEMBLIES
 SECURE CABLES TO END CAP ASSEMBLIES
 INSTALL ACCESS PANELS (TOP, BOTTOM AND ENDS, AS SHOWN)

- INSPECT COMPLETE

ACCESS PANELS

ORIGINAL PAGE 19

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TO TEST

ASSEMBLY SEQUENCE FOR FLIGHT TEST CONFIGURATION

TIOM AND ENDS, AS SHOWN)

MANUFACTURING FLOW AND BUILD PLAN LOW CONCENTRATION RATIO SOLAR COLLECTOR

PREPAREL BY: KIM HICKS DEPT: 761

EXT: 3618 DATE: JAN 25, 1983 REVISED: FEB 4,1983

3-5, 3-6